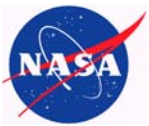




Thermal Energy Storage Devices

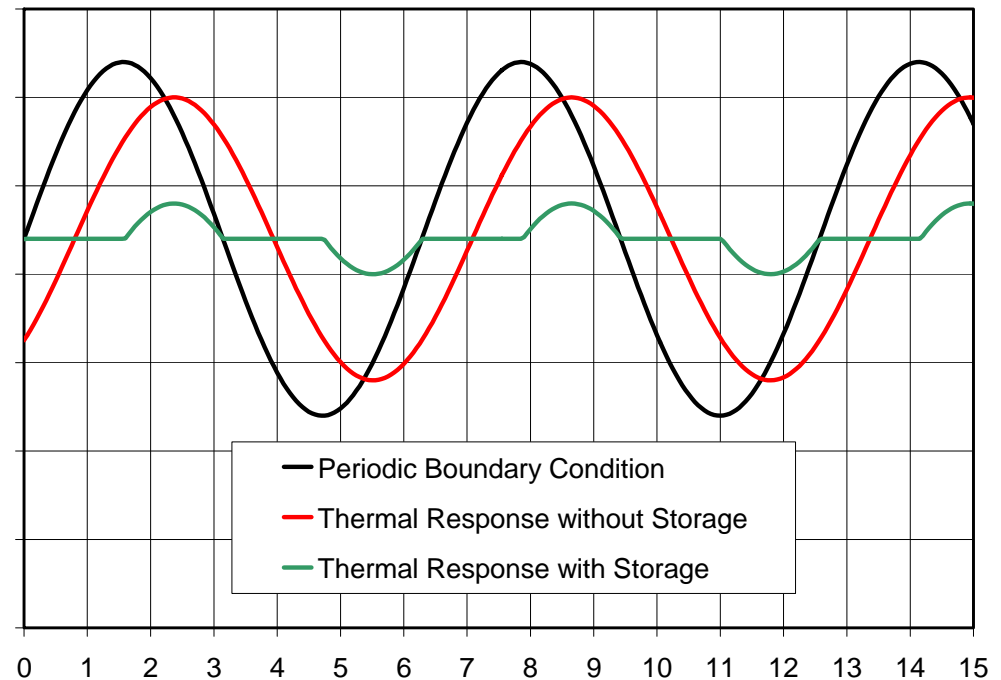
Mike Pauken, Nick Emis
Jet Propulsion Laboratory

August 8, 2006
TFAWS 2006



Why Thermal Energy Storage?

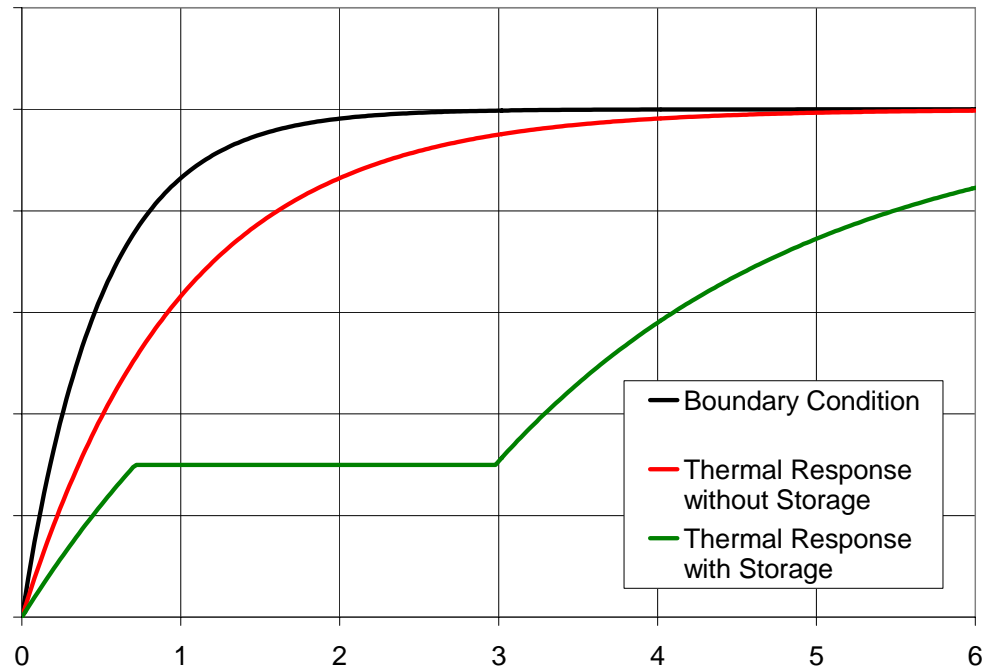
- Dampen effects of periodic boundary conditions on low thermal mass objects.
 - Spacecraft on orbit
 - Spacecraft on planetary surfaces such as Mars
 - Terrestrial Applications





Why Thermal Energy Storage?

- Reduce heating rates of light weight, high powered devices on short duration missions to extreme environments
 - Landing on Venus Surface
 - Atmospheric probes to the Gas Giant Planets





Back-of-the-Envelope Analysis



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- Five basic equations describe Thermal Energy Storage Performance:
 - Conservation of Mass
 - Mass of Phase Change Material, Filler Material and Casing
 - Conservation of Energy
 - Maximum energy storage capacity
 - Temperature Range Constraint
 - Temperature gradient between component and melting temperature
 - Computation of Equivalent Conductance
 - Net conductance of filler plus PCM.
 - Additive Area Relationship
 - Heat flow cross-sectional area of PCM and filler materials

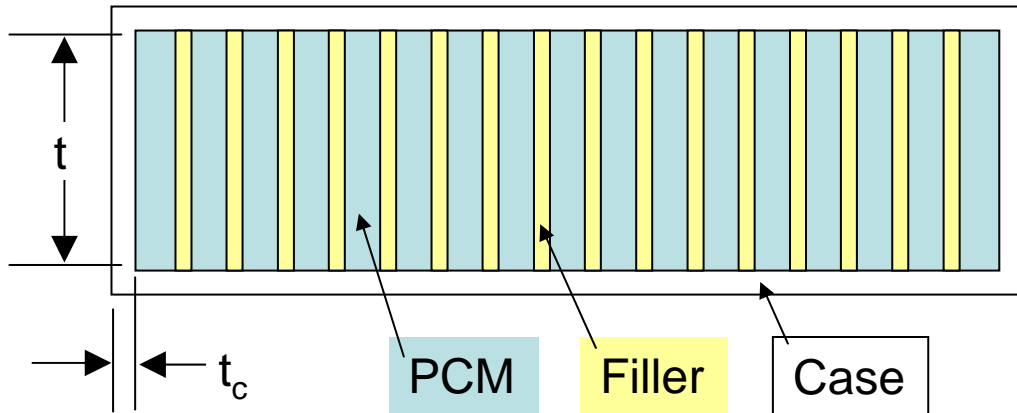
$$\{1\} \text{ Mass} = (\rho_{\text{PCM}}A_{\text{PCM}} + \rho_{\text{F}}A_{\text{F}})t + \rho_{\text{C}}(2A_{\text{T}} + 4tA_{\text{T}}^{0.5})t_{\text{C}}$$

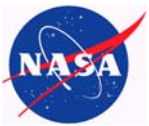
$$\{2\} E_{\text{MAX}} = \rho_{\text{PCM}}A_{\text{PCM}} \cdot t \cdot h_{\text{f}} + [\rho_{\text{F}}A_{\text{F}}C_{\text{P-F}} + \rho_{\text{PCM}}A_{\text{PCM}}C_{\text{P-PCM}}]t/2(T_{\text{comp}} - T_{\text{melt}})$$

$$\{3\} Q = k_{\text{T}}A_{\text{T}}(T_{\text{comp}} - T_{\text{melt}})/t$$

$$\{4\} k_{\text{T}}A_{\text{T}} = k_{\text{PCM}}A_{\text{PCM}} + k_{\text{F}}A_{\text{F}}$$

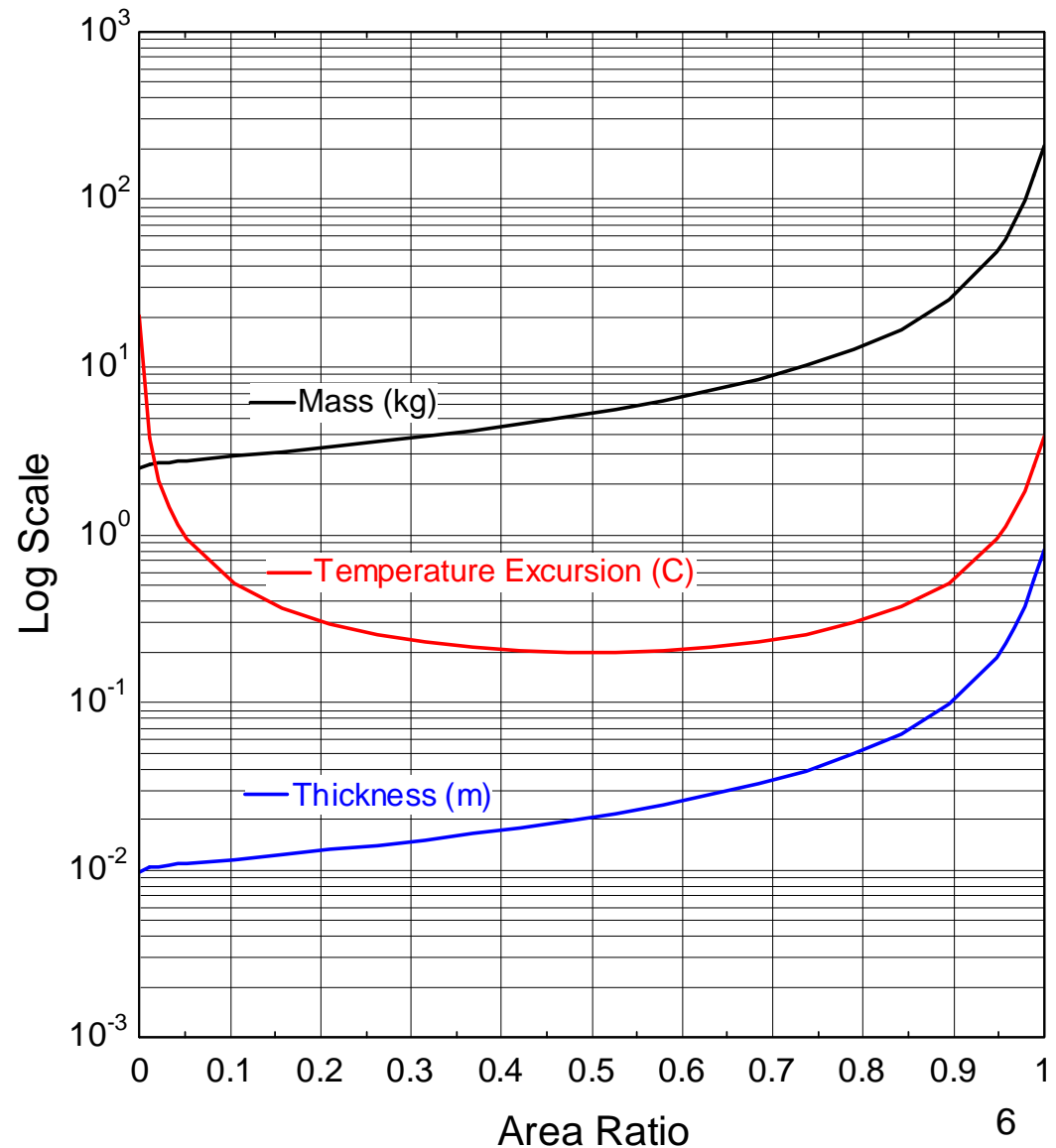
$$\{5\} A_{\text{T}} = A_{\text{PCM}} + A_{\text{F}}$$



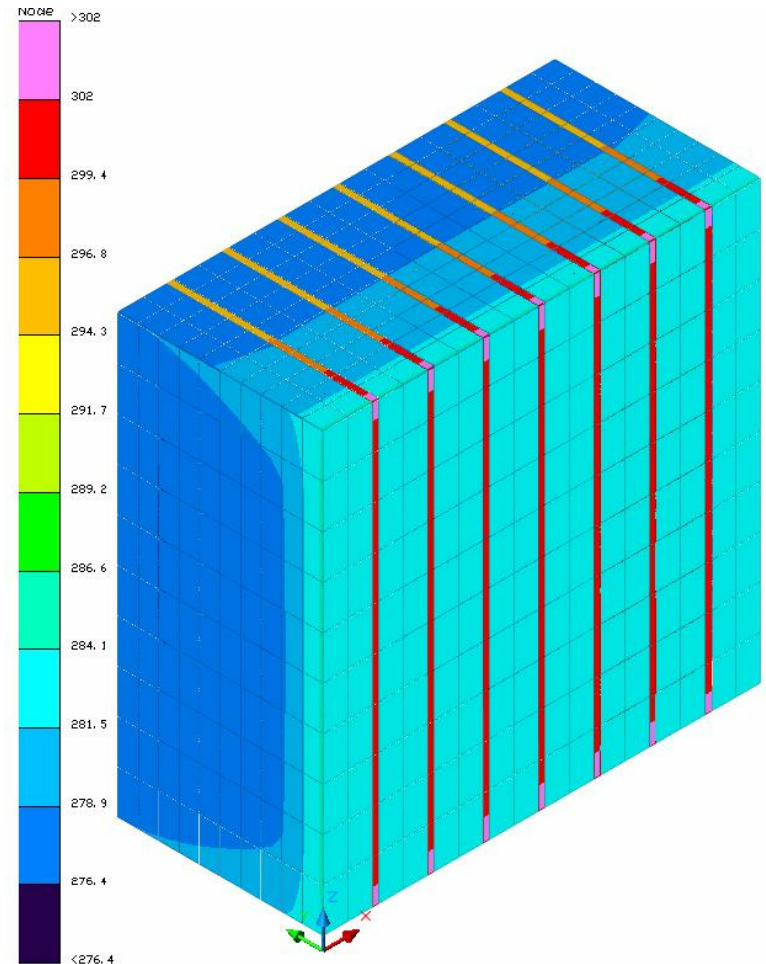


Back-of-the-Envelope Analysis

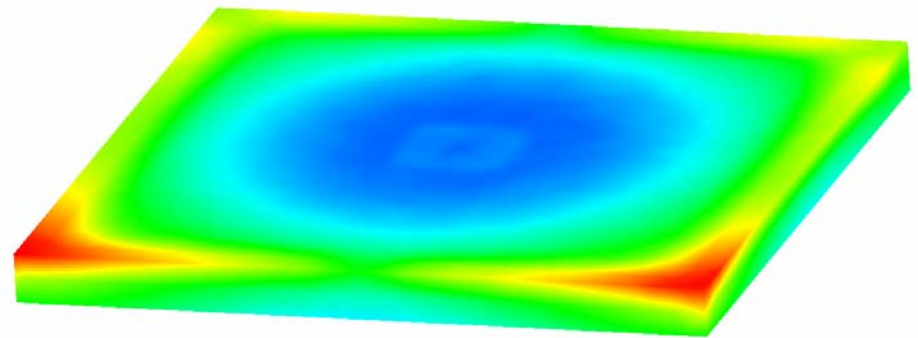
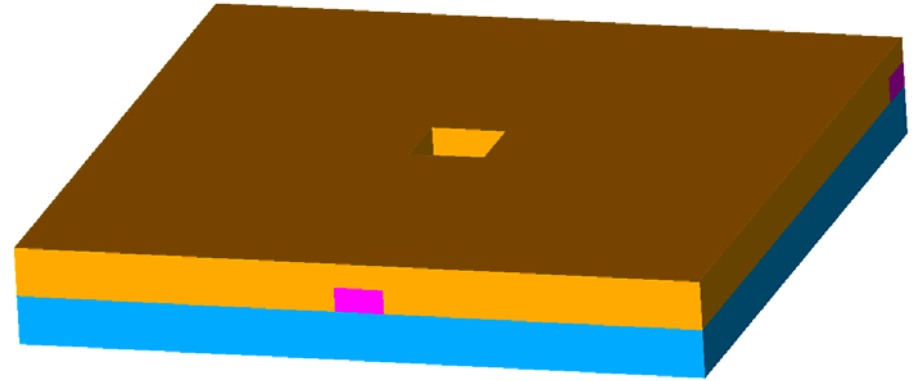
- Sample Results from Simple Analysis
- Area Ratio is:
Filler area/Total Area



- Examine more complex shapes/geometry than rectangular boxes
- Evaluate 2-D and 3-D effects
- Utilize more accurate specific heat data (as a function of temperature and phase)
- Shown is a model of a simple unit with aluminum fins



- Thermal models make it easy to model complex geometries or configurations.
- Module w/ graphite shell, Rubitherm-35 PCM and copper foam filler.
- Copper heater plate mounted on top



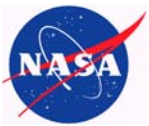


Phase Change Materials



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- Conducted search of phase change materials with high heats of fusion.
- Paraffins are most common melt material.
- Lithium Nitrate has one of the highest heats of fusion and has a high density especially compared to paraffins.
 - Measured h_f for LiNO_3 was ~ 287 kJ/kg
 - Most paraffins range from 170 to 230 kJ/kg
 - Density of LiNO_3 is ~ 1.5 g/cc
 - Most paraffins are around 0.8 g/cc
- Implication: For a given container size, LiNO_3 has more than double the heat capacity of a paraffin!



Lithium Nitrate Developments



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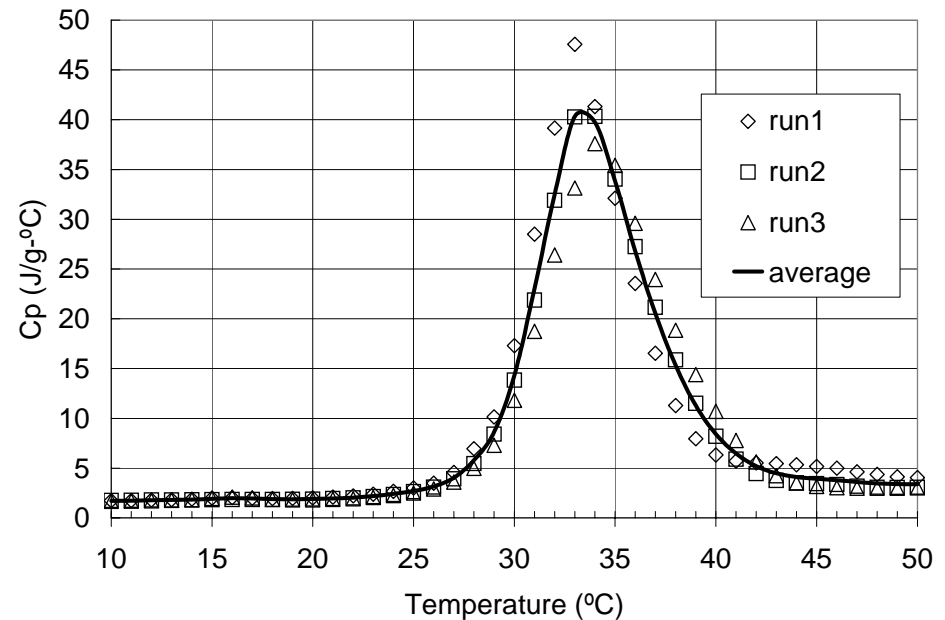
- LiNO_3 is more difficult to handle than paraffins:
 - Must add water to it to form $\text{LiNO}_3 \cdot 3\text{H}_2\text{O}$
 - It readily absorbs water from ambient air and reduces heat capacity.
 - Must keep it sealed until loaded into Thermal Storage Module
 - Melting point is about 30°C , freezing point is around -5°C .
 - This phenomenon is called subcooling.
 - Can be reduced with a catalyst: add 1% by weight of ZnNO_3
 - With catalyst, freezing point is around 28°C .
- LiNO_3 appears to be compatible with aluminum, no corrosion products have been observed.

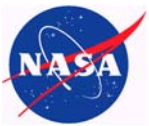


Heat Capacity Measurements

- Heat capacity of LiNO_3 was measured because literature data in both solid and liquid phases were incomplete
- Measurements were made using Differential Scanning Calorimetry
 - This tends to produce broader melting temperatures than observed in bulk samples.

Specific heat measurements of LiNO_3





Filler Material Developments



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- Aluminum fins have traditionally been used as the filler material for conducting heat into the melt material.
- Carbon foam, trademarked name “Pocofoam”, has been used successfully as a filler material with paraffin loaded storage modules by others.
- Carbon foam is hydrophobic and will not absorb water based materials such as $\text{LiNO}_3 \cdot 3\text{H}_2\text{O}$.
- Adding a small amount of surfactant to $\text{LiNO}_3 \cdot 3\text{H}_2\text{O}$ solves the absorption problem.

- Pocofoam floats in untreated water as shown in top photograph demonstrating hydrophobia.
 - This means LiNO_3 would not be absorbed into Pocofoam
- Adding surfactant to the water causes immediate absorption by the Pocofoam as shown in bottom photograph and sinks.
 - Same trick works for water spiders!



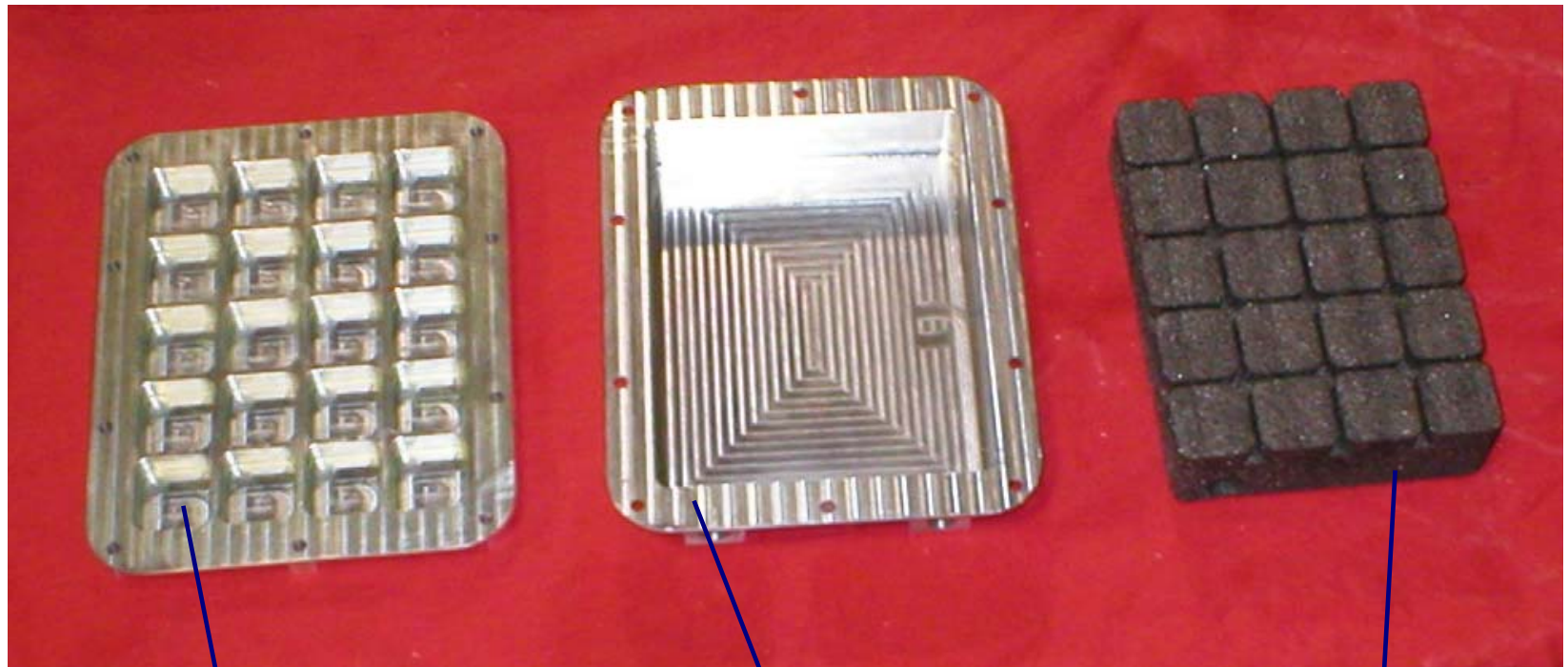


Prototype Design Features



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- A phase change thermal energy storage module was designed with the following features:
 - A stiff lid to make a stable mounting surface for accommodating electronic components (we just used a heater for testing).
 - A thin backplate to act as a diaphragm to reduce pressure variations as LiNO_3 changed phase.
 - Carbon foam filler core
 - $\text{LiNO}_3 \cdot 3\text{H}_2\text{O}$ plus 1% zinc nitrate plus surfactant

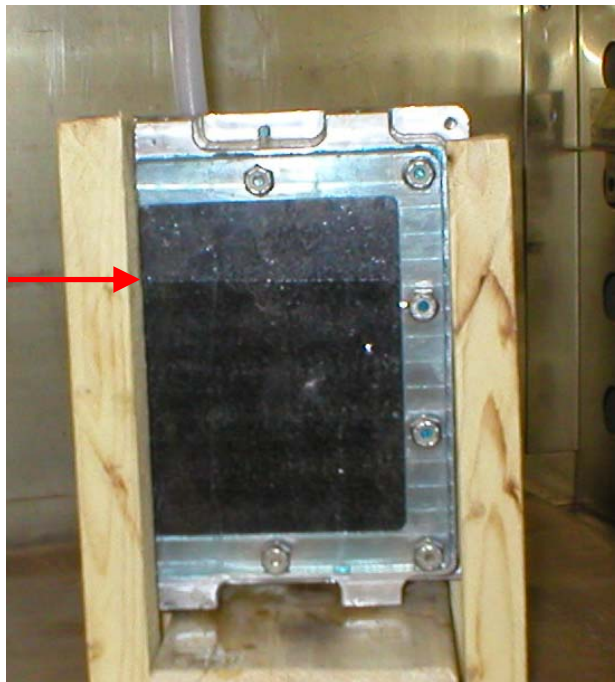


LID

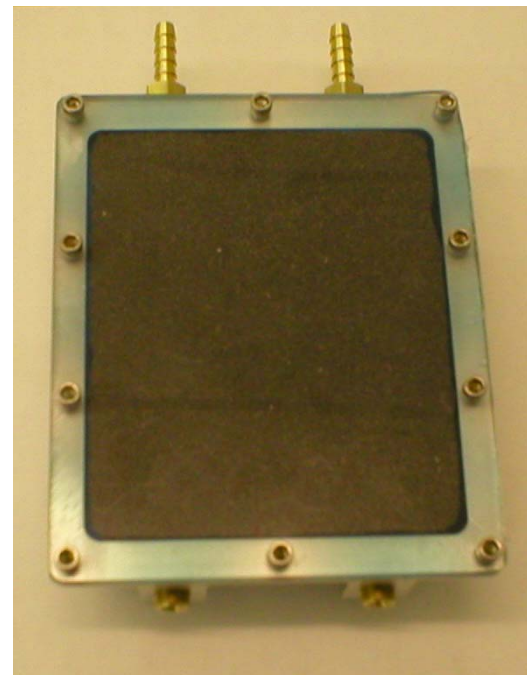
ENCLOSURE

POCO Foam

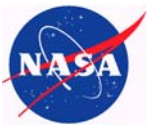
- Before making the aluminum casing module, a Plexiglas unit was fabricated to verify filling because of the hydrophobic nature of Pocofoam.



Liquid line evident during filling



Completely filled module

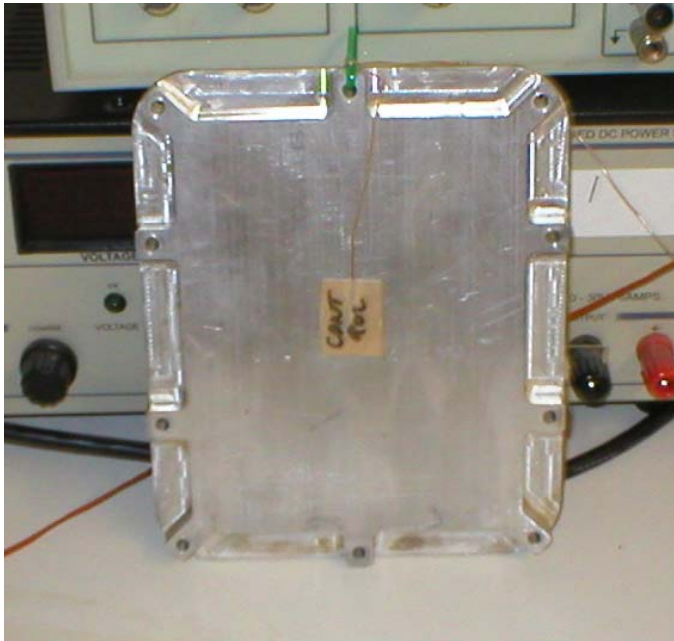


Aluminum Casing Prototype



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- Photographs of the completed phase change module



- Power levels applied to unit: 15, 30, 60 and 90 watts
- Tests ran from 0 to 90°C
- Tests repeated 5 times each to check consistency
- Module was placed inside insulated box to reduce heat leaks.
 - Observed approximately 8 watts of heat leak through insulation

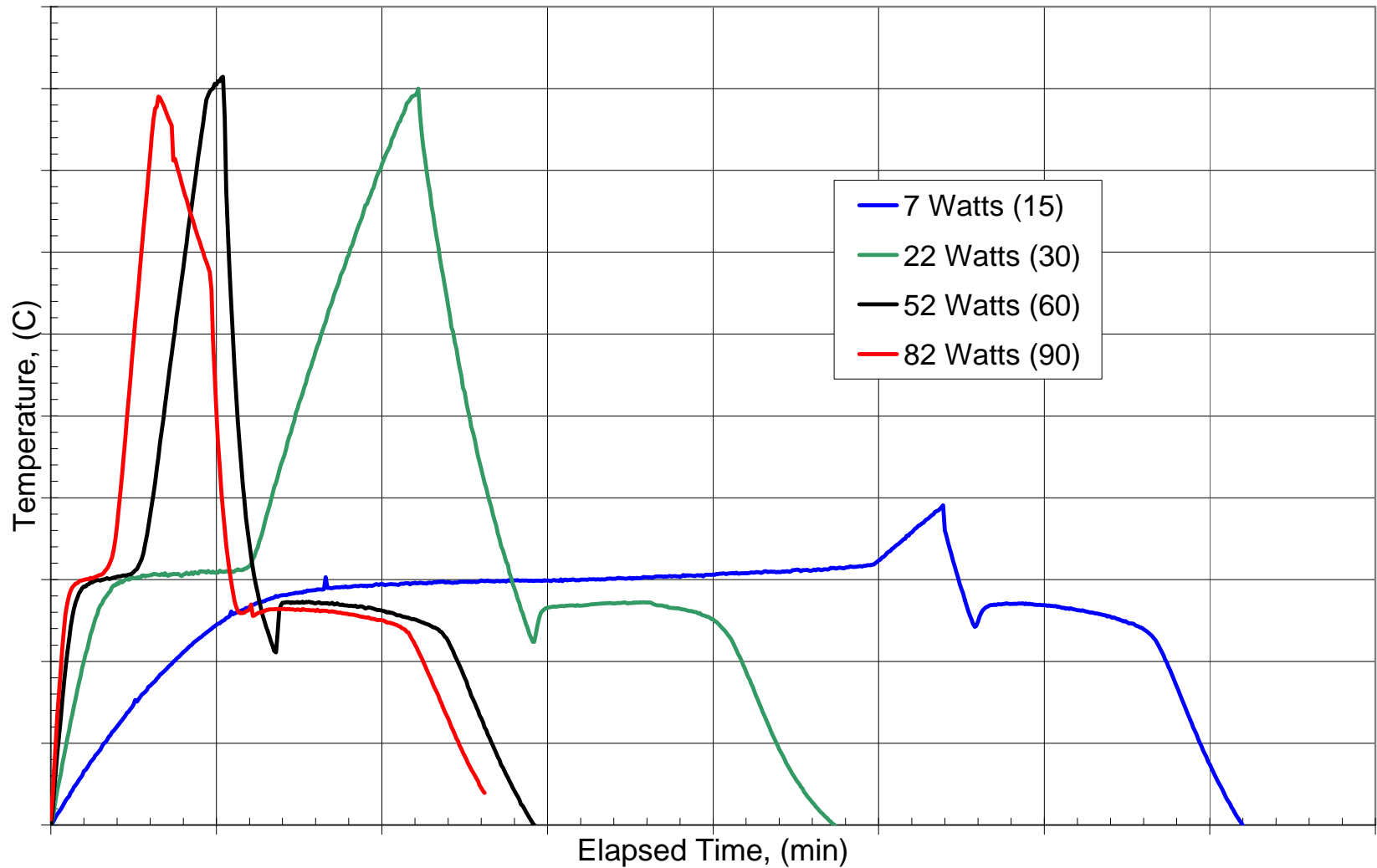
Test Set up at
Vendor's facility

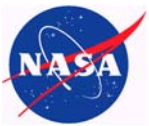




Test Results

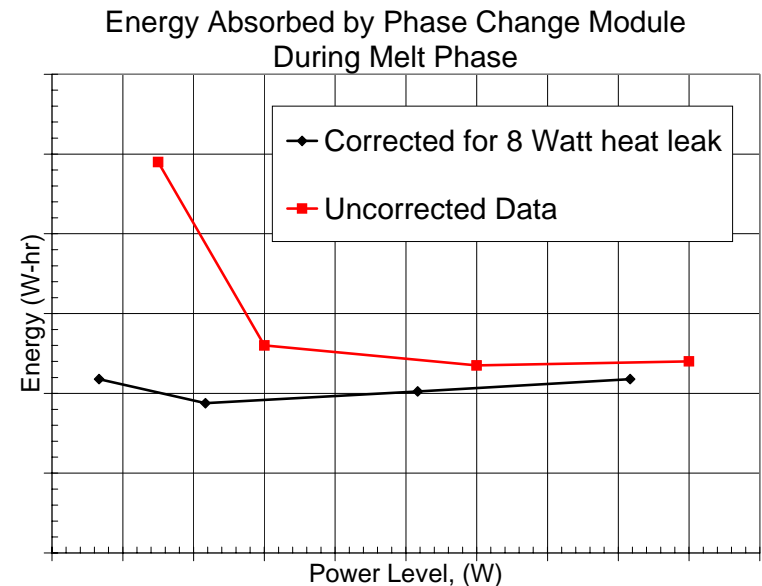
PCM Tests Comparing Different Heat Loads

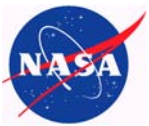




Correction for Heat Leak

- Energy absorbed during melt should not change with power.
- Observed 15 Watt case took much longer to melt than expected.
- Estimated 8 Watt heat leak and corrected the melt energy to be around 40 W-hr/kg.
 - If paraffin were the melt material, melt energy would be around 20-30 W-hr/kg





Conclusions



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- An improved Thermal Energy Storage Device suitable for spacecraft applications has been developed.
- The melt material was lithium nitrate and had a melt temperature around 30C.
- The freezing point subcooling was reduced to only 2C with the addition of 1% zinc nitrate.
- POCOfoam was used as the filler material, hydrophobic effects were eliminated using a surfactant.
- Energy storage capacity of unit is 30% to 100% greater than a paraffin filled unit.



Acknowledgments



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- This work was performed by Jet Propulsion Laboratory, California Institute of Technology under contract with the National Aeronautics and Space Administration.
- Fabrication and testing of the thermal storage device was performed by XC Associates, Stephentown, NY.