

Thermal Stress Analysis of TPS using Marc

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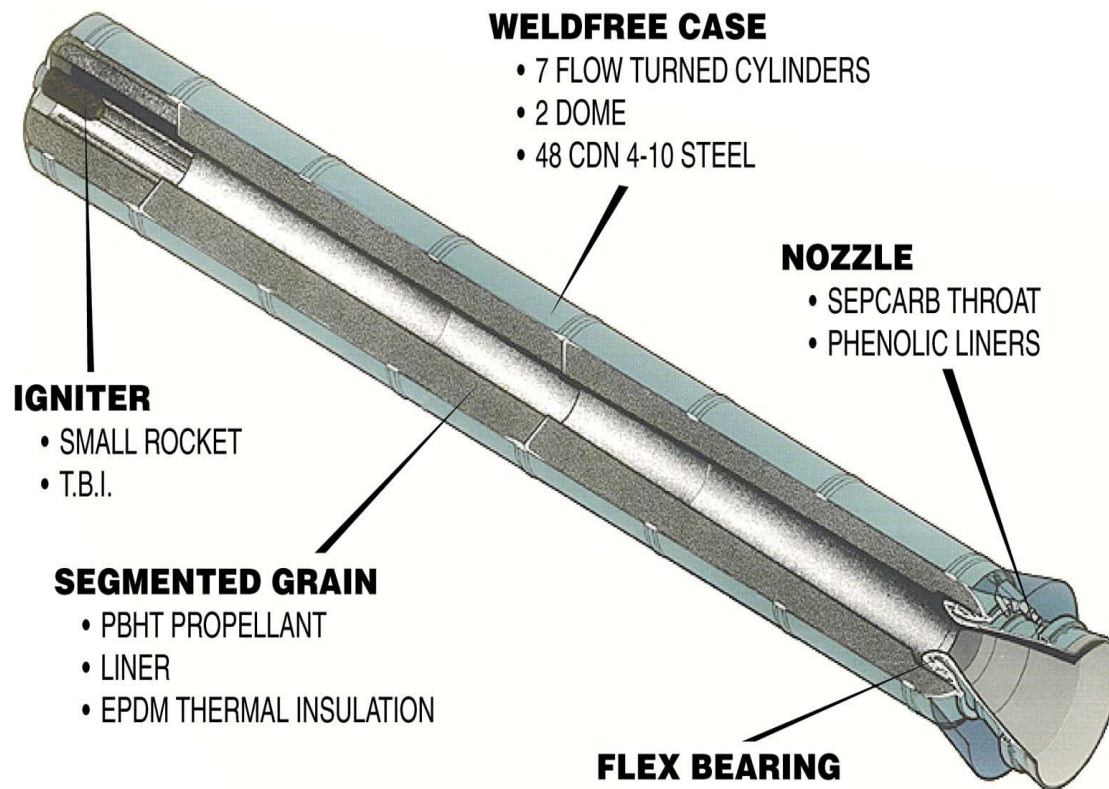


Overview

- Problem Statement
- Introduction of TPS Materials
- Thermal Modeling Methods
- Mechanical Simulation
- Material Models
- Contact
- Examples

Problem Statement

ARIANE 5 SRM - TECHNOLOGIES



Problem Statement

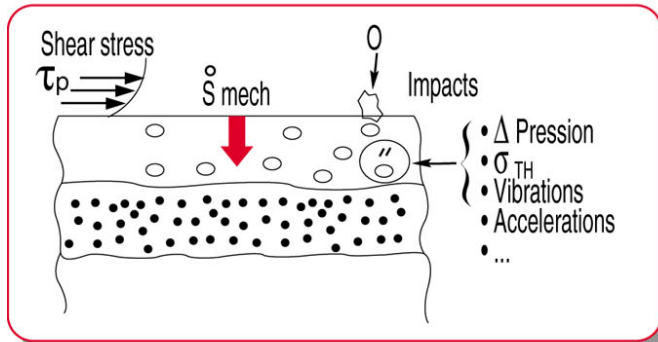
- Extremely High Temperatures Achieved
- Extremely High Thermal Gradients
- Coefficient of Thermal Expansion Mismatch
- Stress Analysis Required
- Large Changes in Geometry
- Requirements for Coupled Analysis

TPS Materials

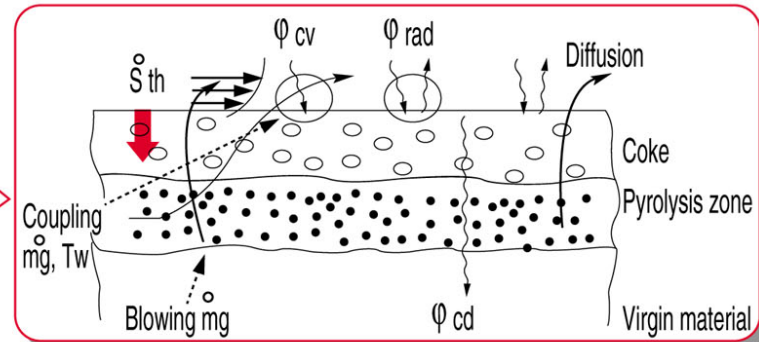
- Carbon/Carbon
- Carbon/Phenolic
- Silica/Phenolic
- Ceramic Matrix composites
- Rubber and Reinforced Rubber
- Low Mass Thermal Insulators

Physics Overview

MECHANICAL ASPECTS



THERMAL ASPECTS

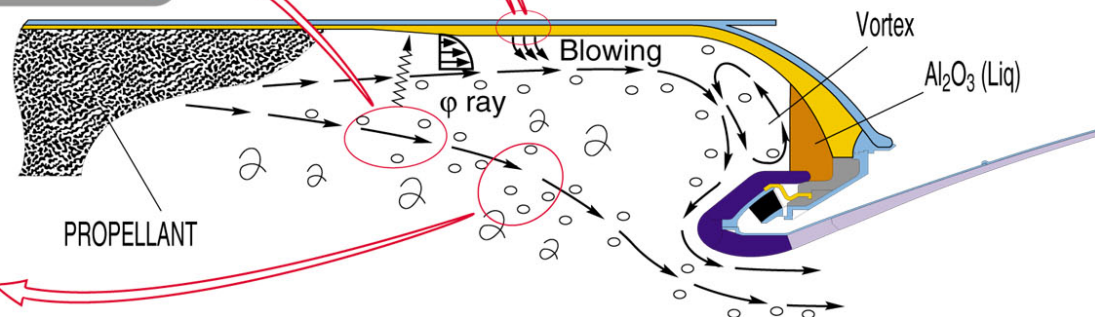


RADIATIVE HEAT TRANSFER

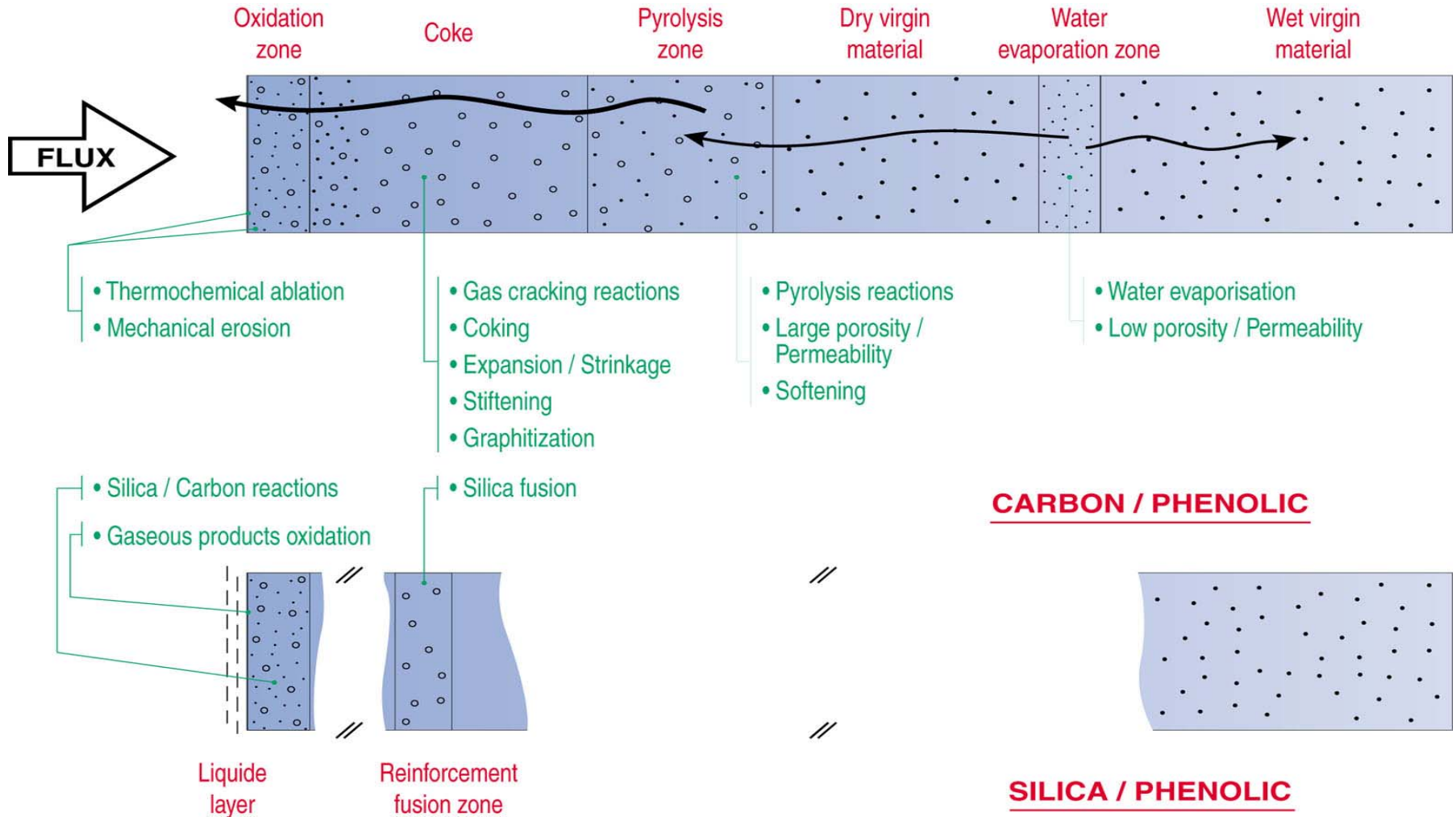
- Emitting / diffusing medium
- H_2O, CO_2, CO, HCl , particles
- Flow / radiation coupling

FLOW

- 3D \Rightarrow 2D
- Two-phase
- Multicomponent
- Turbulent
- Reactive

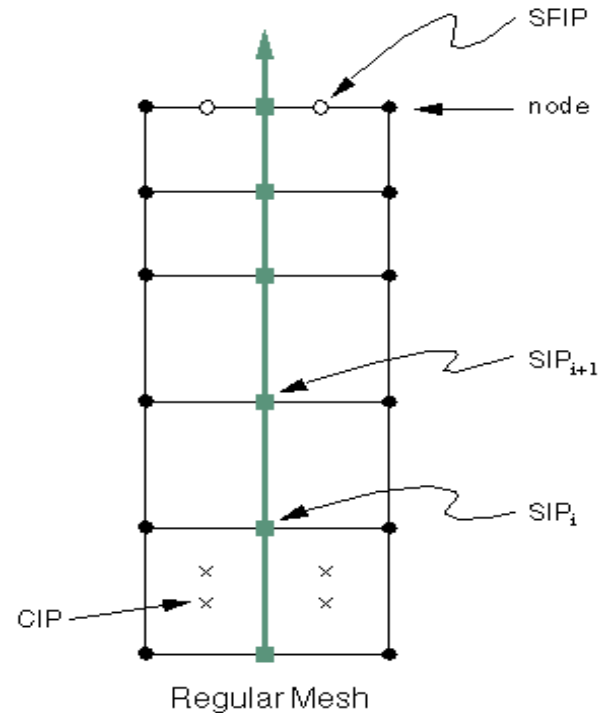


Thermo-Degradation Process



Thermal Modeling Methods

- Level 1 - One Dimensional Fluid Flow
- Level 2 - Three Dimensional Fluid Flow (Darcy Law)



{SIP} Streamline Integration Point¶
 {CIP} Conventional Integration Point¶
 {SFIP} Surface Integration Point§

Advanced Material Model

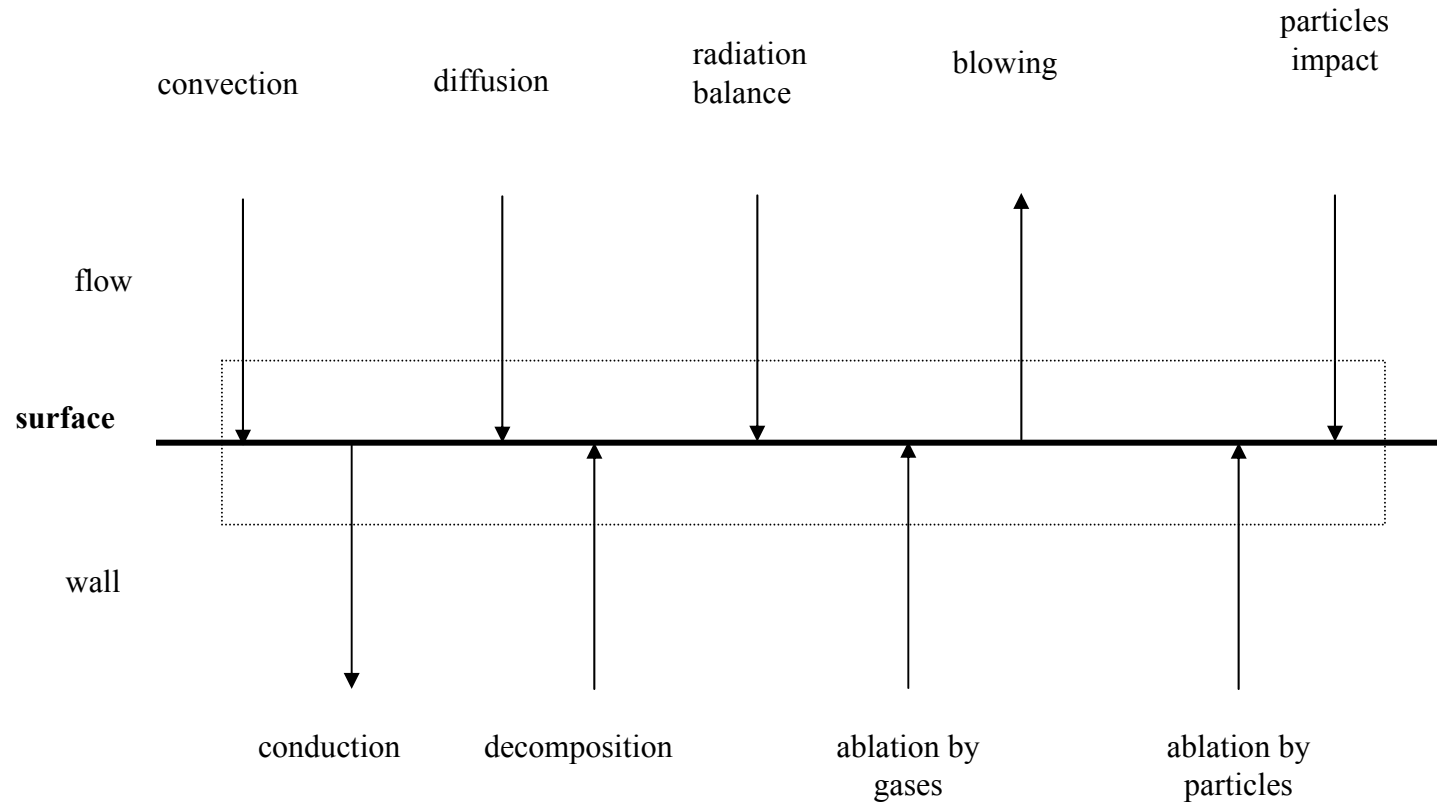
- Pyrolysis of Material
 - Mass Density Controlled by Arrhenius Law
 - Thermal Properties Change based upon a Kachanov Model between Virgin and Charred State
 - Energy absorption and internal convection
- Water Vapor Creation
- Coking
 - Carbon comes out of the Pyrolysis Gases and Deposits onto the Solid

Arrhenius Law for φ_j

- Dimensionless variable φ_j that goes from 1 to 0 during pyrolysis : calculated by Arrhenius law:

$$\frac{\partial \varphi_j}{\partial t} = -B_j \exp\left(\frac{-T_{a,j}}{T_s}\right) \varphi_j$$

Surface Energy Balance



- Thermochemical Ablation (Gases, Particles)

$$\dot{S}_{th} = [\dot{m}_{s,th,g} + \dot{m}_{s,th,p}] / \hat{\rho}_s$$

- Mechanical Erosion
 - Due to impacts of particles
 - Due to other actions such as the shear stress of the flow and vibration of the part

Energy Equation

$$\left(\hat{\rho}_{s,i} c_{pi} + \hat{\rho}_{s,p} \bar{c}_{s,p} \right) \frac{\partial T}{\partial t} + c_{pg} \dot{m}_g \cdot \nabla T =$$

$$\nabla \cdot \left(\lambda^* \nabla T \right) + \frac{\partial \hat{\rho}_{s,p}}{\partial t} \left(H_{g,p} - \bar{H}_{s,p,vc} \right)$$

Mechanical Simulation

- Uncoupled Mechanical Analysis
 - Perform Thermo-Pore Simulation
 - Move Temperatures into Structural Analysis
 - Problems with Geometry
- Coupled Mechanical Analysis
 - Solver Thermo-Pore-Mechanical in Staggard Manner

Thermal-Mechanical Coupled Analysis

- Mechanical Dependence on Thermal Analysis
 - Temperature Dependent Structural Properties
 - Thermal Strains
 - Geometry Changes due to Ablation
- Thermal Dependence on Structural Analysis
 - Geometric Changes Influence Boundary Conditions
 - Contact
 - Heat Generated due to Inelastic Behavior
- Diffusion Dependence on Structural Analysis
 - Porosity dependent upon strain

Structural Material Models

- Linear Elastic
- Elastic-Plastic
 - Von Mises or Hill yield
 - Mohr – Coulomb
 - Cam Clay
 - Isotropic, Kinematic, Combined, Power Law, Kumar, Johnson-Cook Hardening Models
 - Powder Model
- Shape Memory
- Rubber Models
 - Mooney-Rivlin
 - Ogden
 - Arruda-Boyce
 - Gent
 - Foam

Structural Material Models

- Nonlinear Elasticity – Hyperelastic
- Gurson or User Defined Damage Model
- Chaboche
- Composites
 - Layered Shells – Note Ablation is not allowed on a Shell
 - Brick
 - Solid Shell
 - Failure Criteria – Progressive Failure
 - Maximum Stress or Strain
 - Tsai-Wu
 - Hoffman
 - Hill
 - Puck
 - Hashin

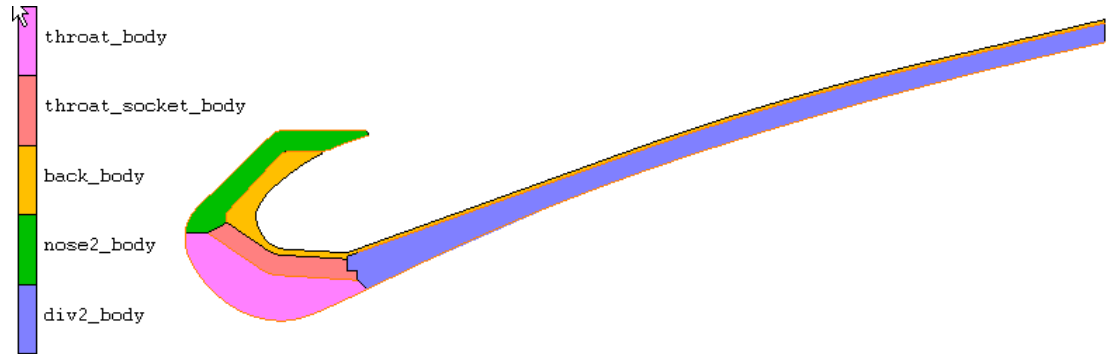
- Mixture Models
- Fixed Volume Fraction
 - Elastic
 - Elastic-Plastic
 - Simplified nonlinear elastic
- Variable Fraction
 - Phase transformations – future
 - Pyrolysis

Structural Material Properties

- Pyrolysis
- Elastic Properties, Coefficient of thermal expansion
- $E^*(T) = (1 - \xi_p) E_v(T) + \xi_p(1 - \xi_c) E_c(T) + \xi_p \xi_c E_{ck}(T)$
- Combine the virgin properties, charred properties and coked properties
- For nonlinear Elasticity will use
- $\mathbf{D}^* = (1 - \xi_p) \mathbf{D}_v(T) + \xi_p(1 - \xi_c) \mathbf{D}_c(T) + \xi_p \xi_c \mathbf{D}_{ck}(T)$
- Where \mathbf{D} is the tangent stress-strain law
 $\Delta\sigma = \mathbf{D}\Delta\varepsilon$

Contact

- Mechanical
 - No Penetration
 - Separation
 - Friction

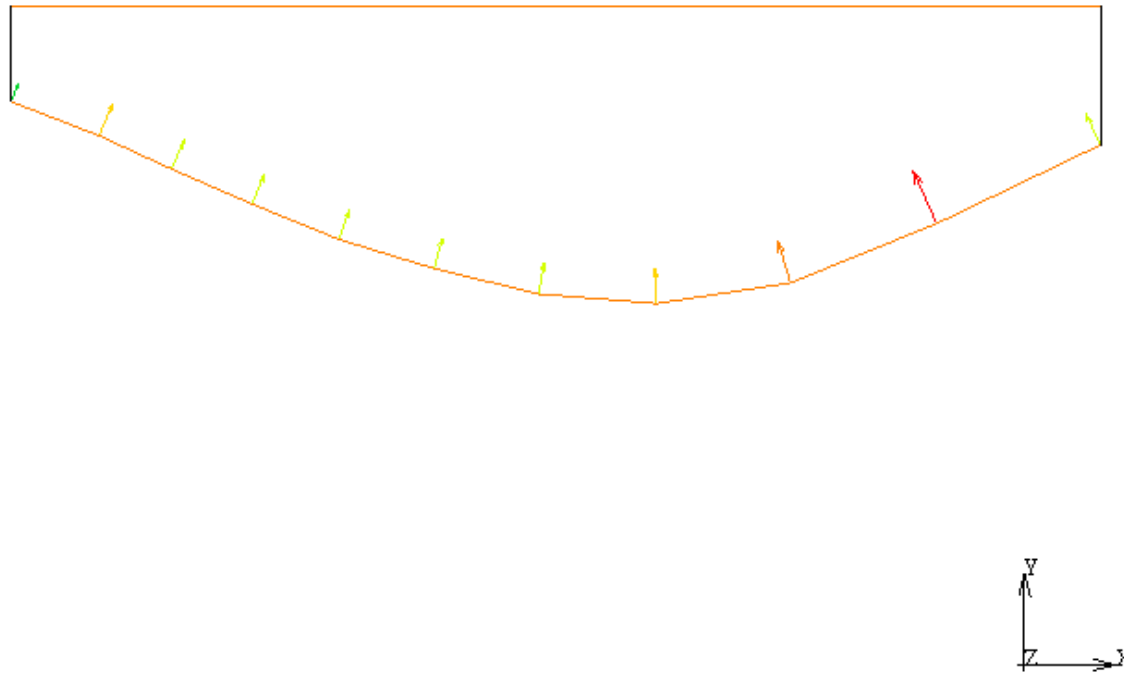


- Thermal
 - True Contact – User defined thermal contact coef.
 - Near Contact
 - To Environment

$$Q = h_{cv} * (T_2 - T_1) + h_{nt} * (T_2 - T_1)^{nt} + \sigma * \epsilon * (T_2^4 - T_1^4) + (h_{ct} - (h_{ct} - h_{bl}) * \text{gap} / d_{qnear}) * (T_2 - T_1)$$

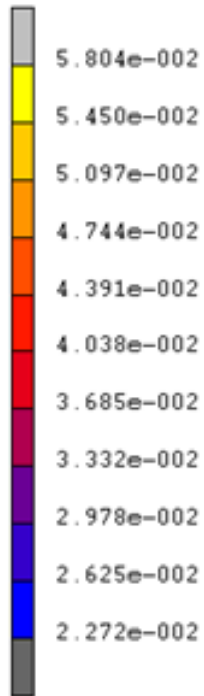
- Diffusion
- Contact Table
- Easy to use

Follower Force Effects

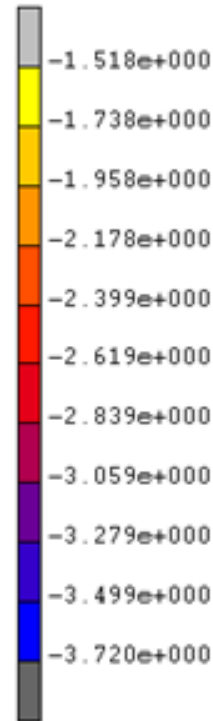


1-D Ablation and Thermal Strains

Inc: 6000
Time: 6.000e+001



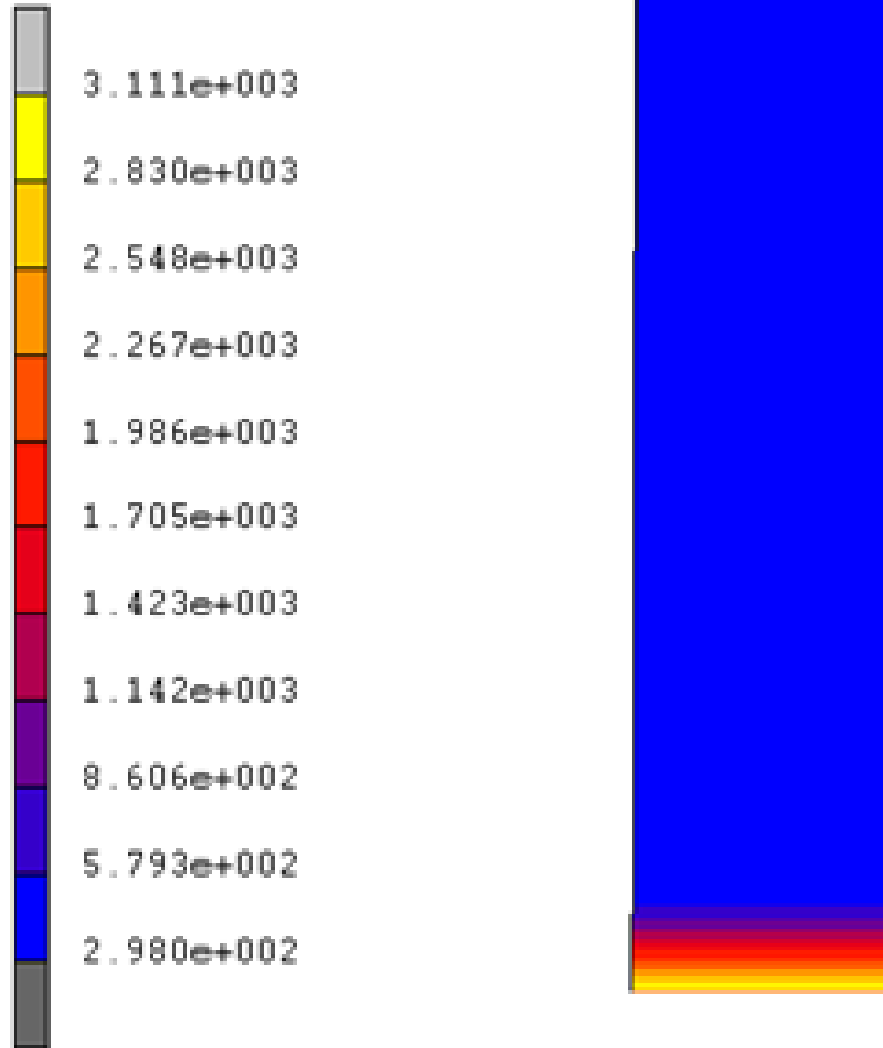
Inc: 6000
Time: 6.000e+001



Max Thermal Strain
<6%

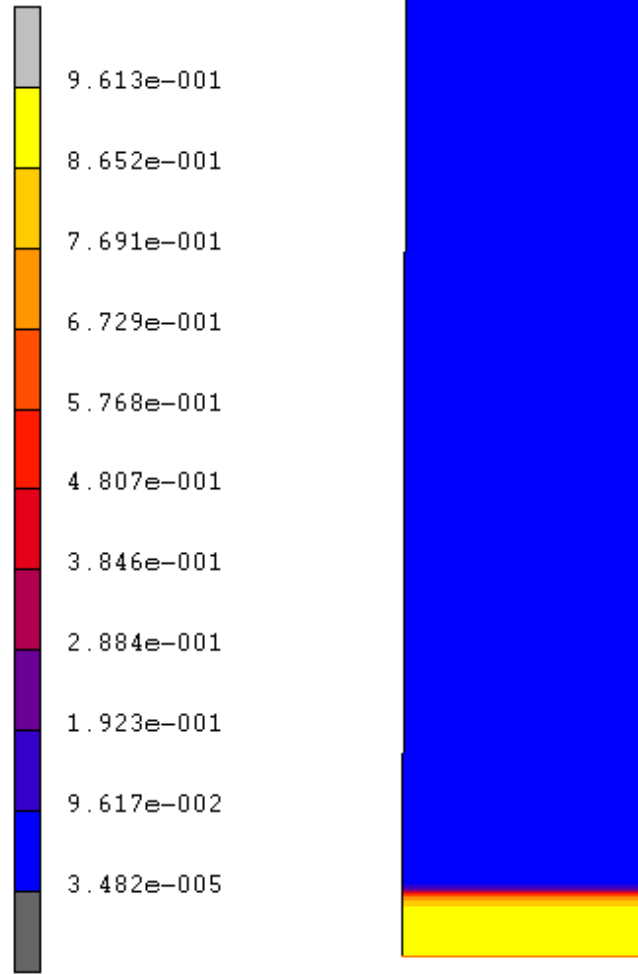
1-D Ablation and Pyrolysis - Temperature

Inc: 4600
Time: 1.100e+001



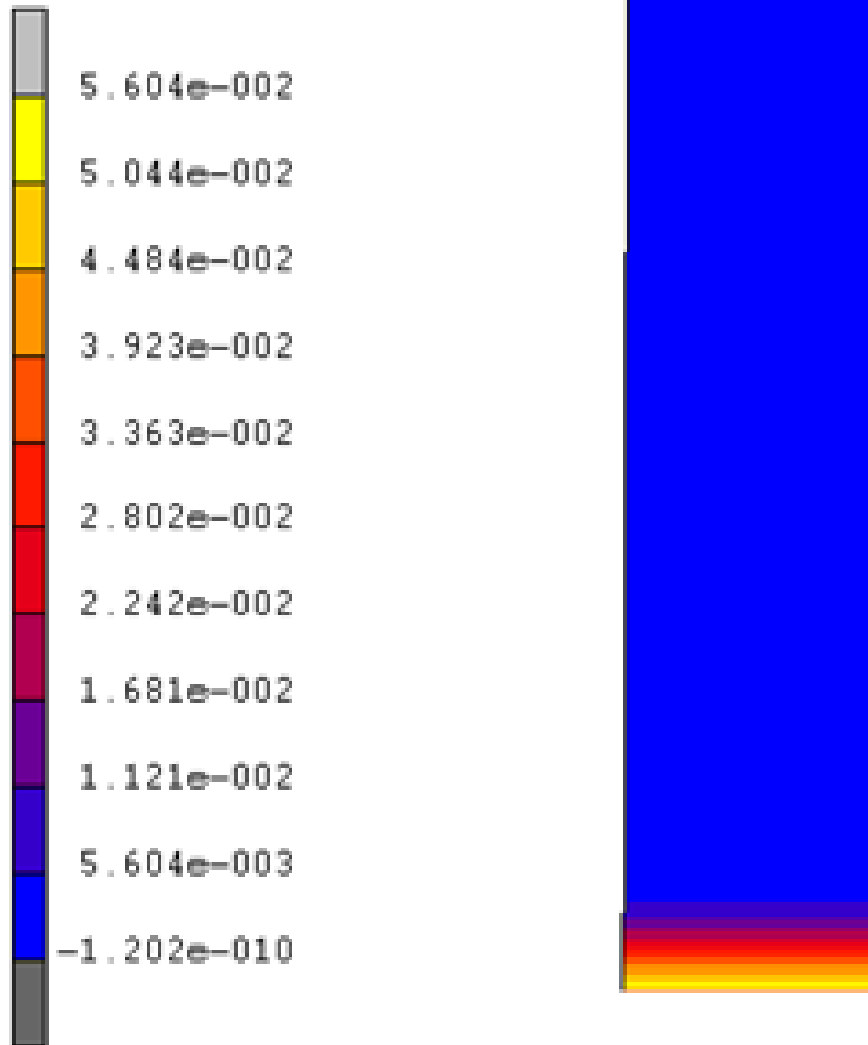
1-D Ablation and Pyrolysis – Char Fraction

Inc: 4600
Time: 1.100e+001



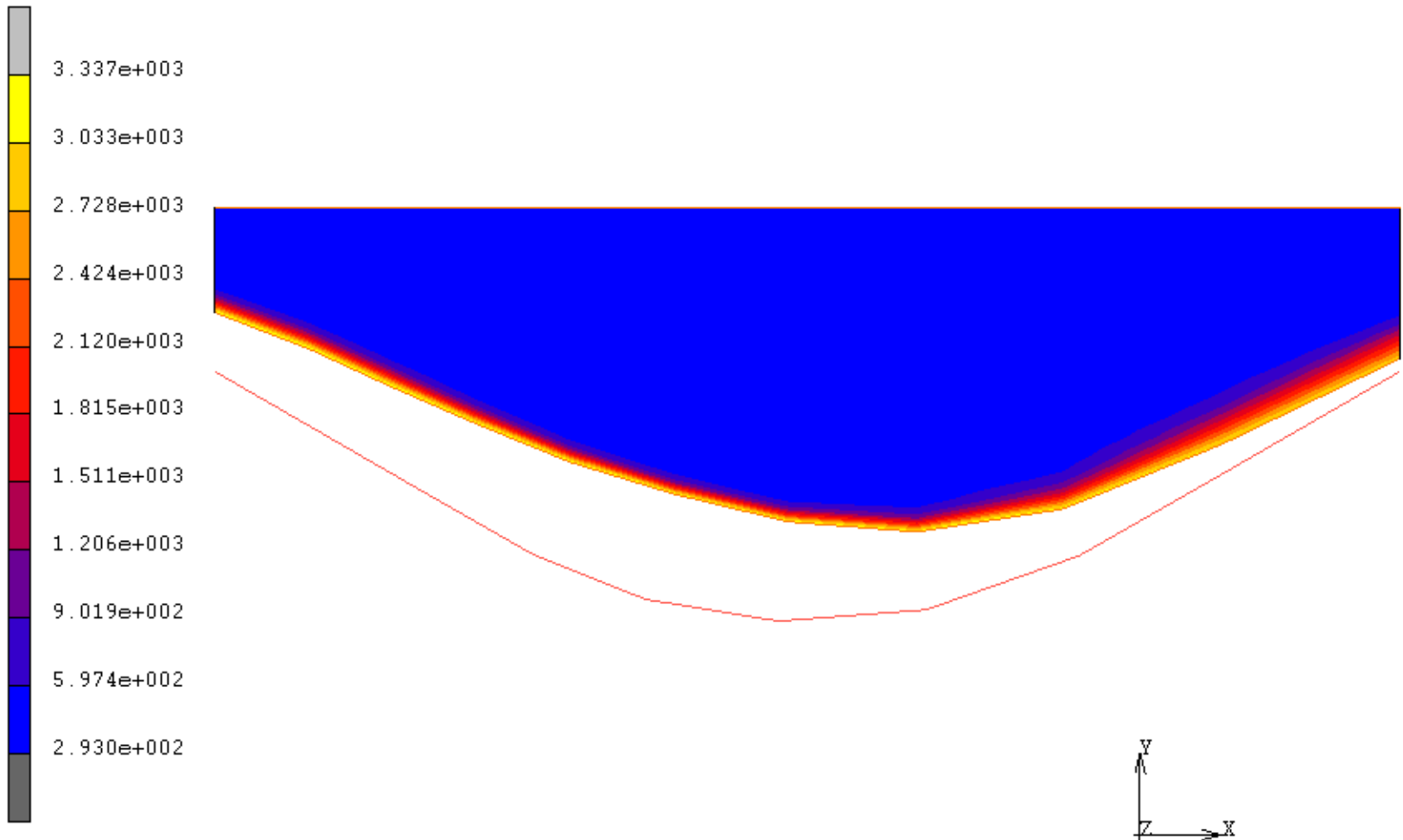
1-D Ablation and Pyrolysis – Thermal Strain

Inc: 4600
Time: 1.100e+001



Simplified Throat Section- Temperatures

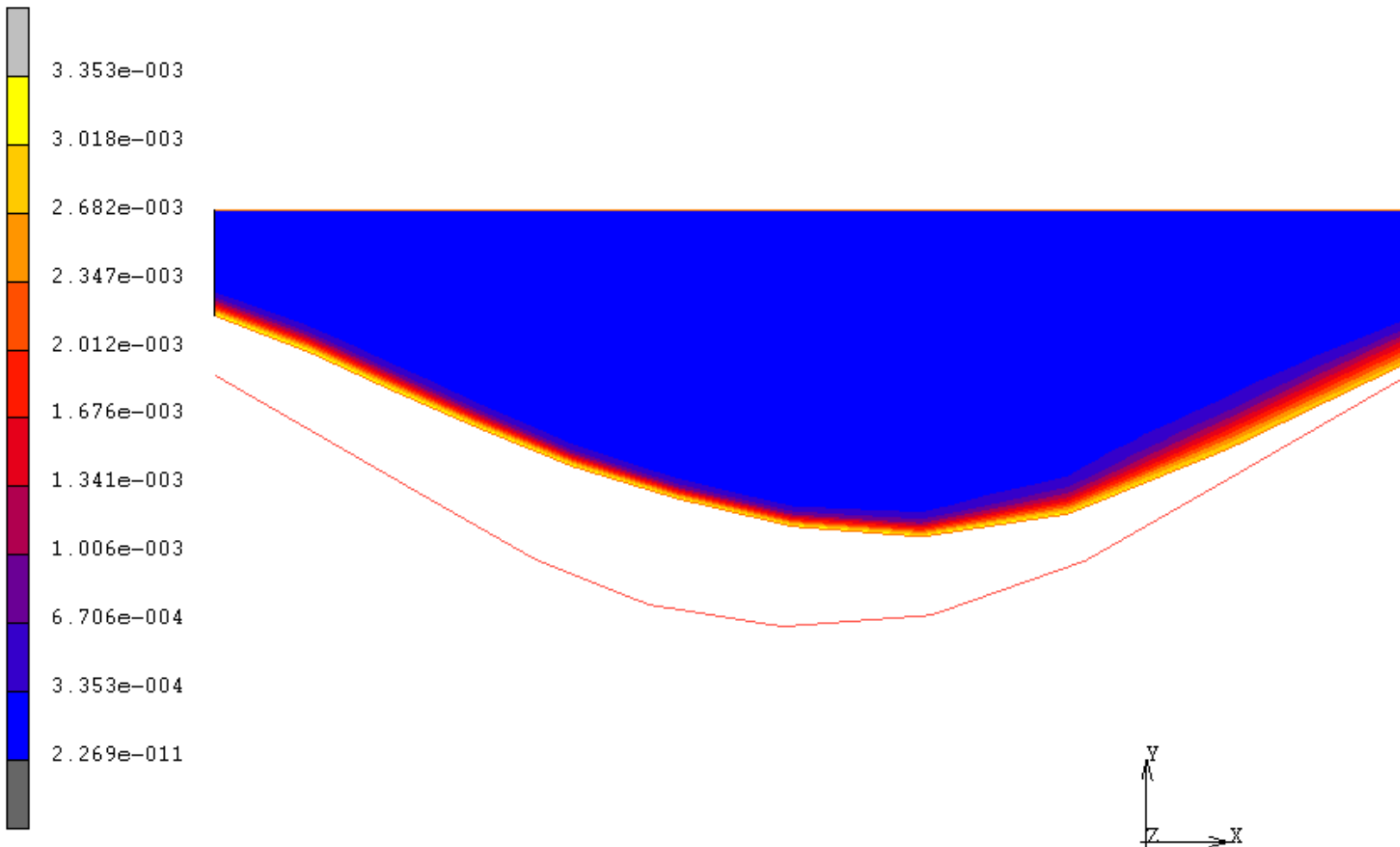
Inc: 1900
Time: 4.000e+001



lcase1
Temperature

Simplified Throat Section – Thermal Strain

Inc: 1900
Time: 4.000e+001



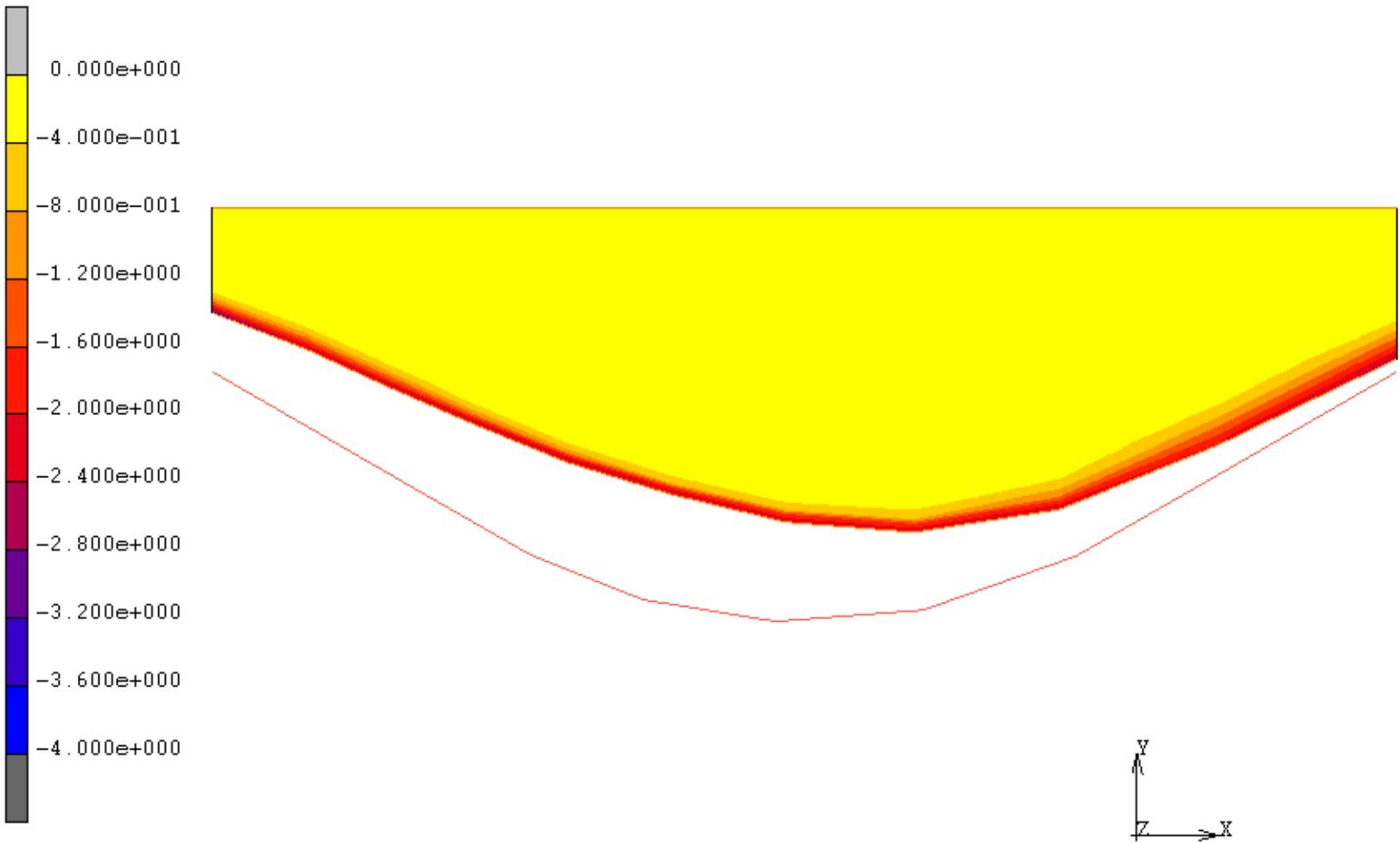
lcase1

Comp 22 of Thermal Strain

Simplified Throat Section – Stress

Inc: 1900
Time: 4.000e+001

MSC Software

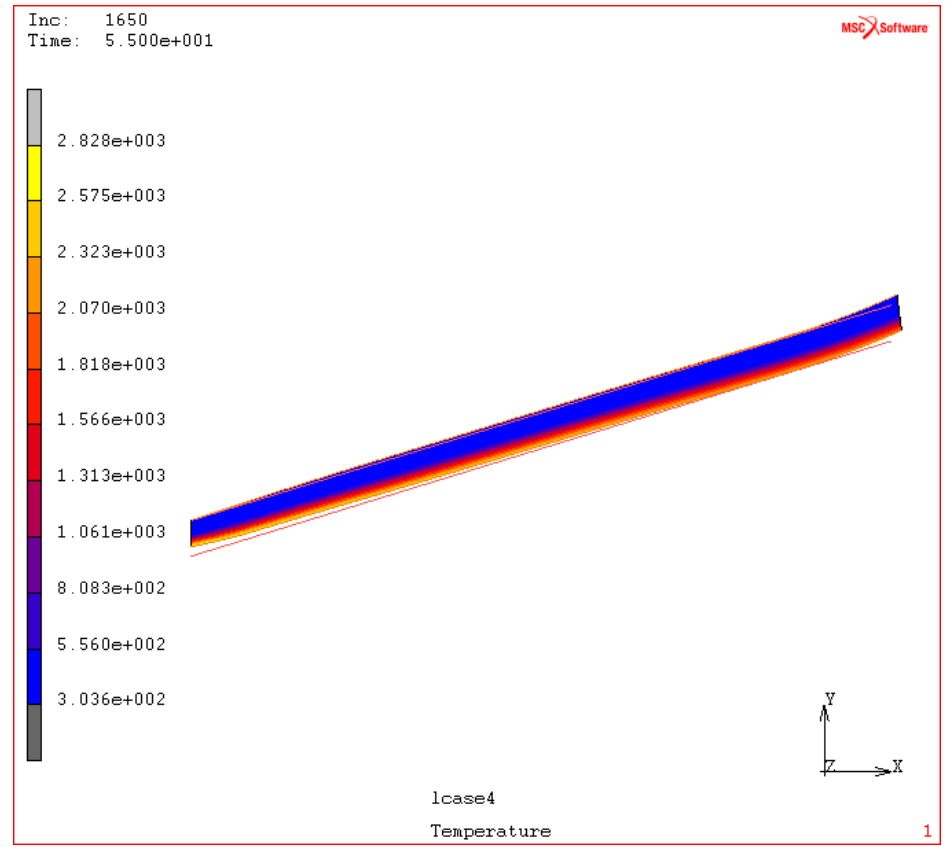
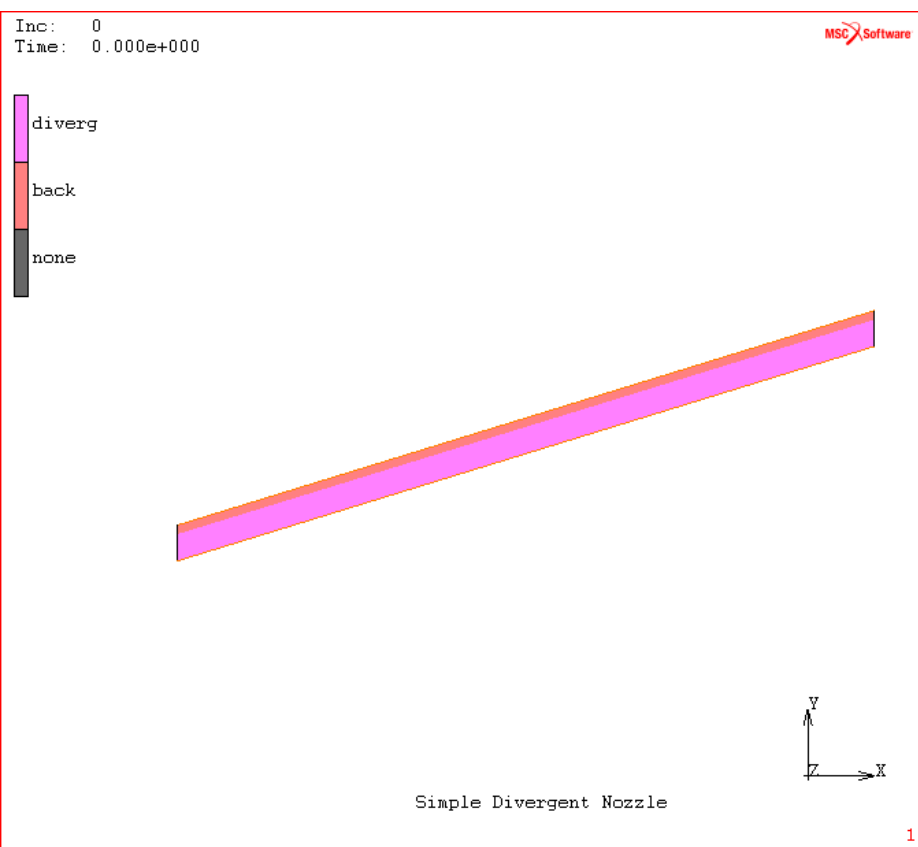


lcase1

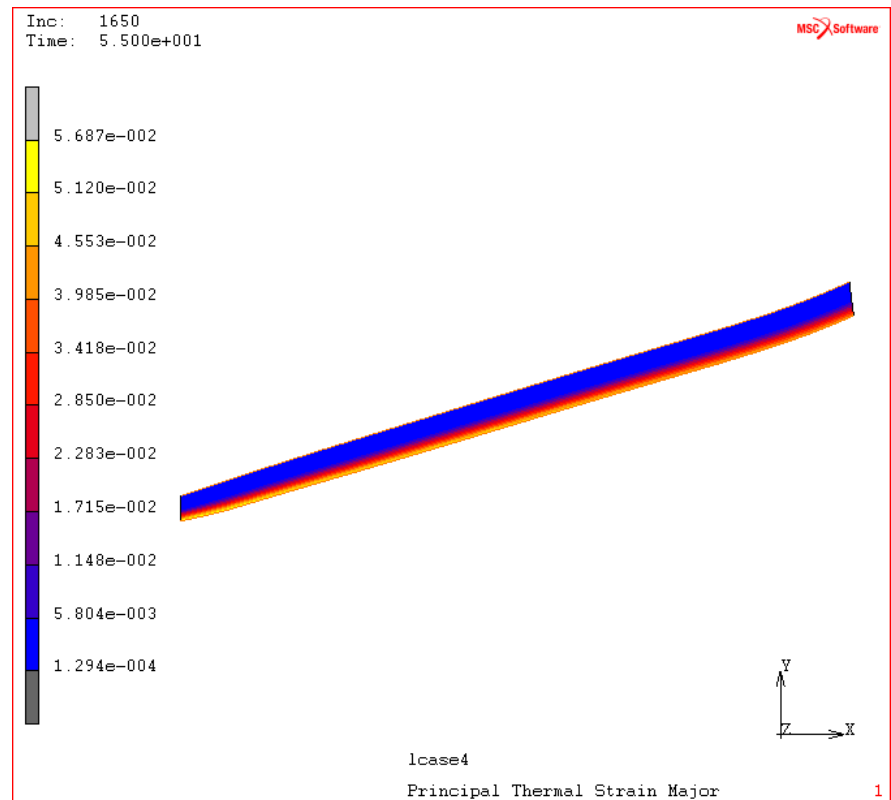
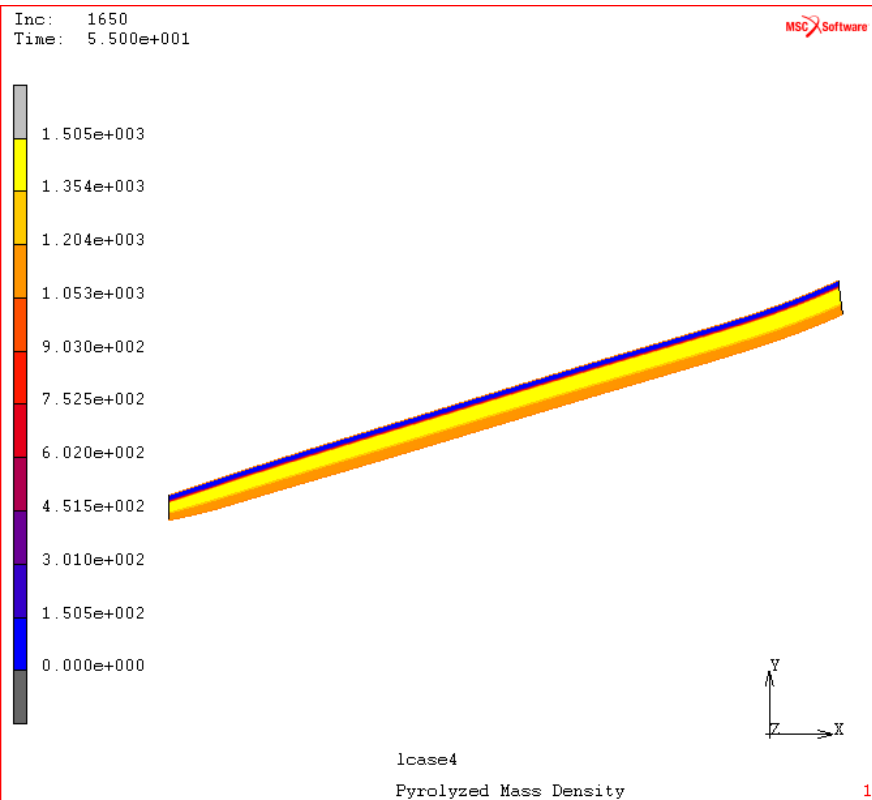
Principal Stress Major

1

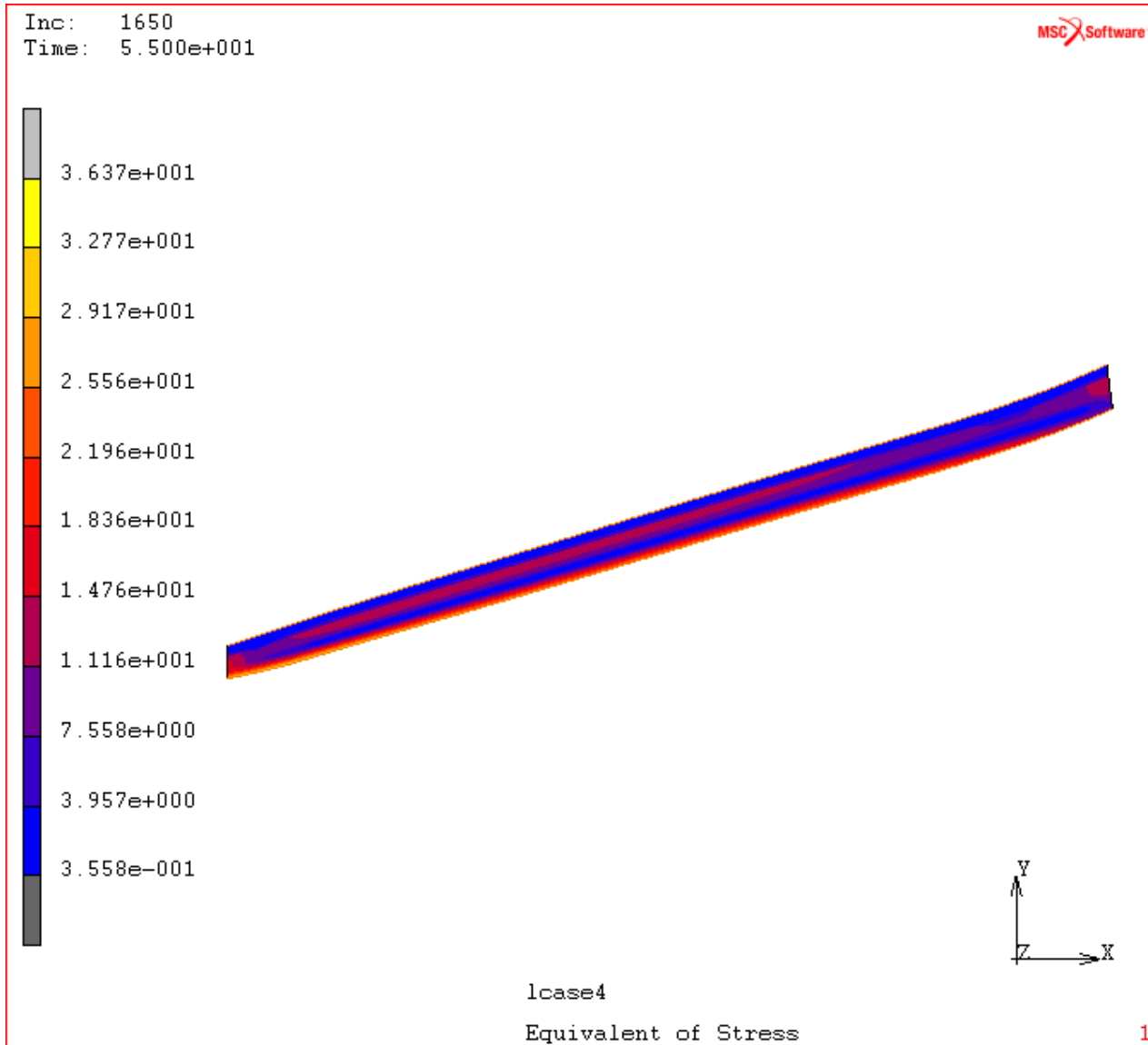
Simplified Exit Nozzle – 1-D



Simplified Exit Nozzle – 1-D

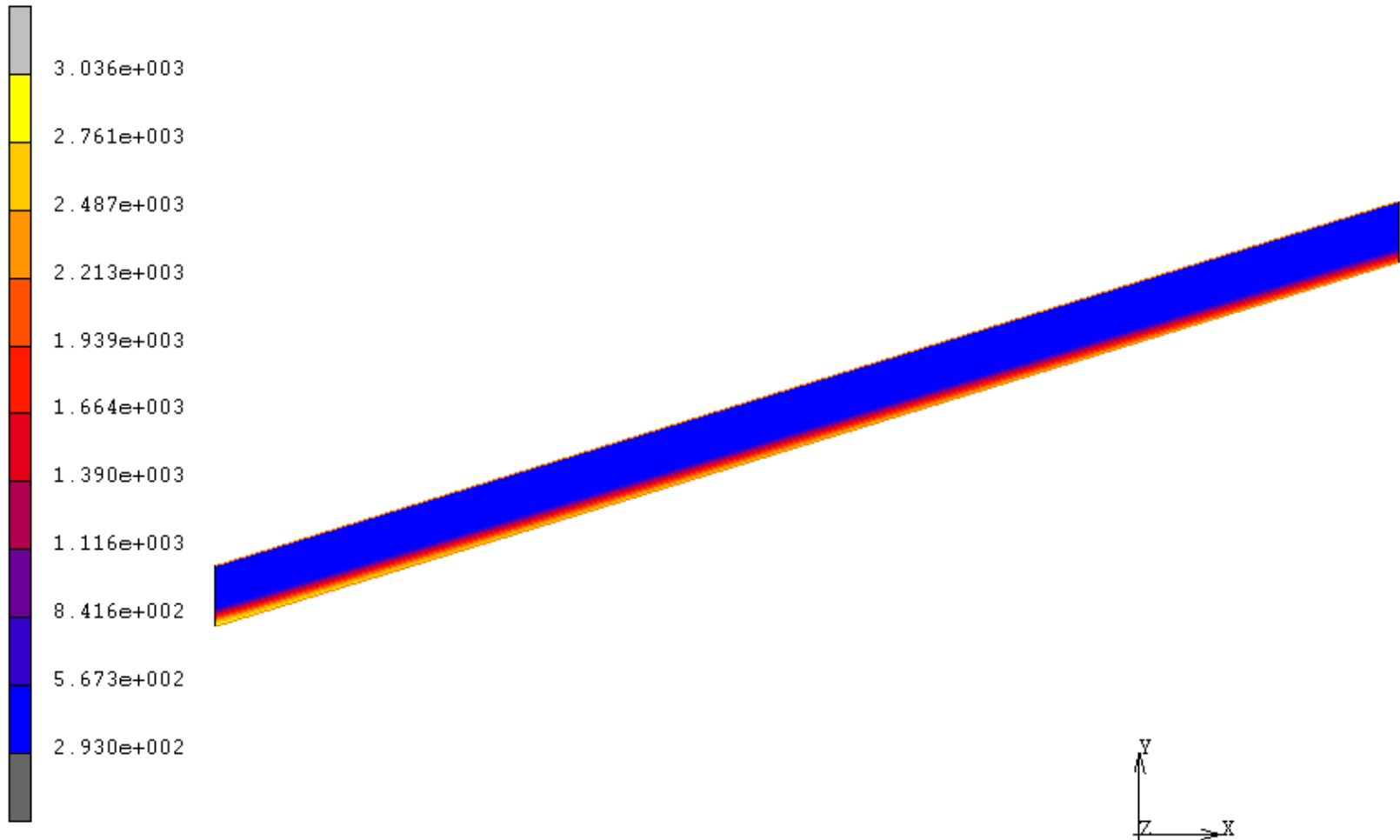


Simplified Exit Nozzle – 1-D



Simplified Exit – Darcy Flow - Temperature

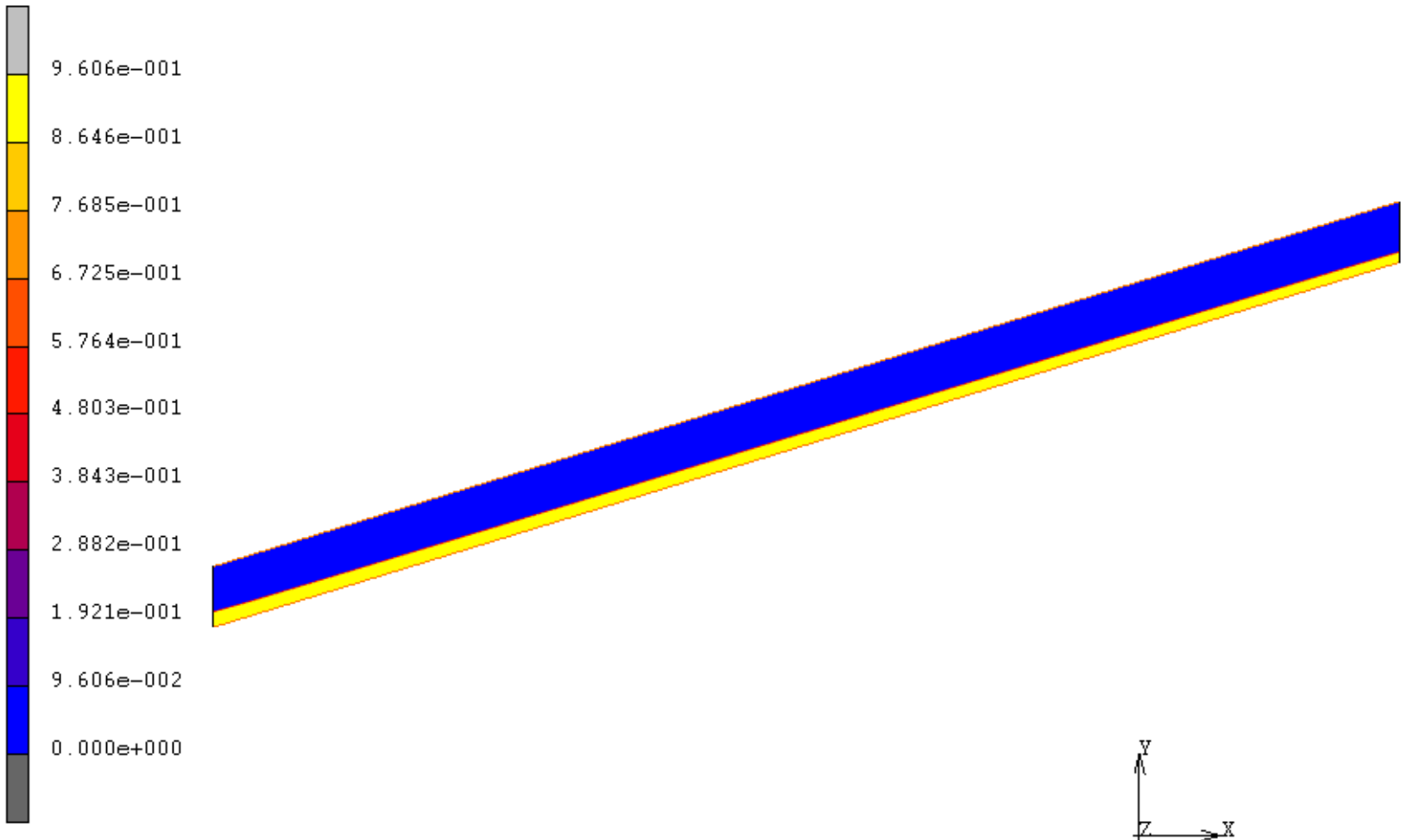
Inc: 1740
Time: 1.650e+001



lcase2
Temperature

Simplified Exit – Darcy Flow - Xsi

Inc: 1740
Time: 1.650e+001



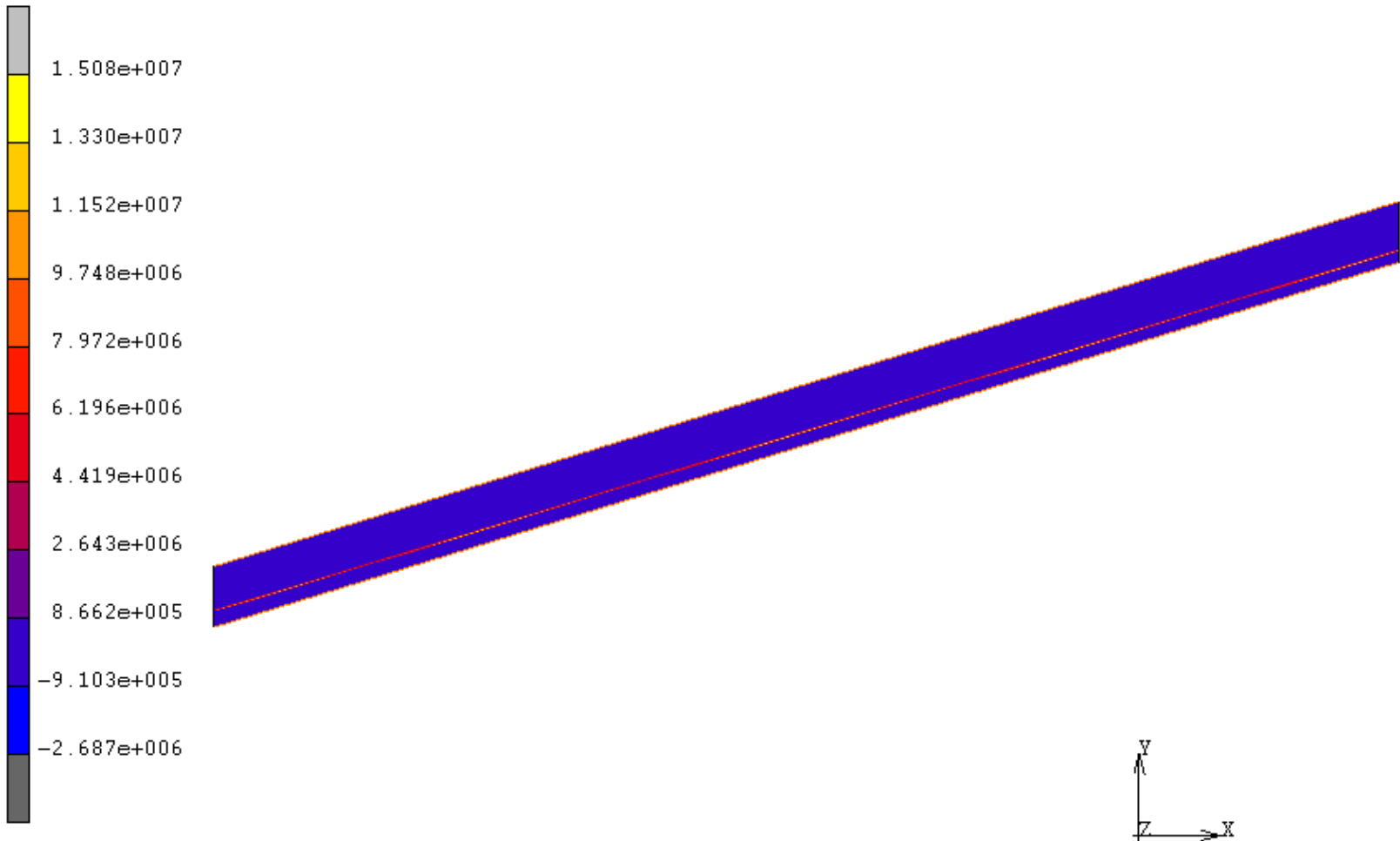
lcase2

xsi.

Simplified Exit – Darcy Flow – Gas Pressure

Inc: 1740
Time: 1.650e+001

MSC Software

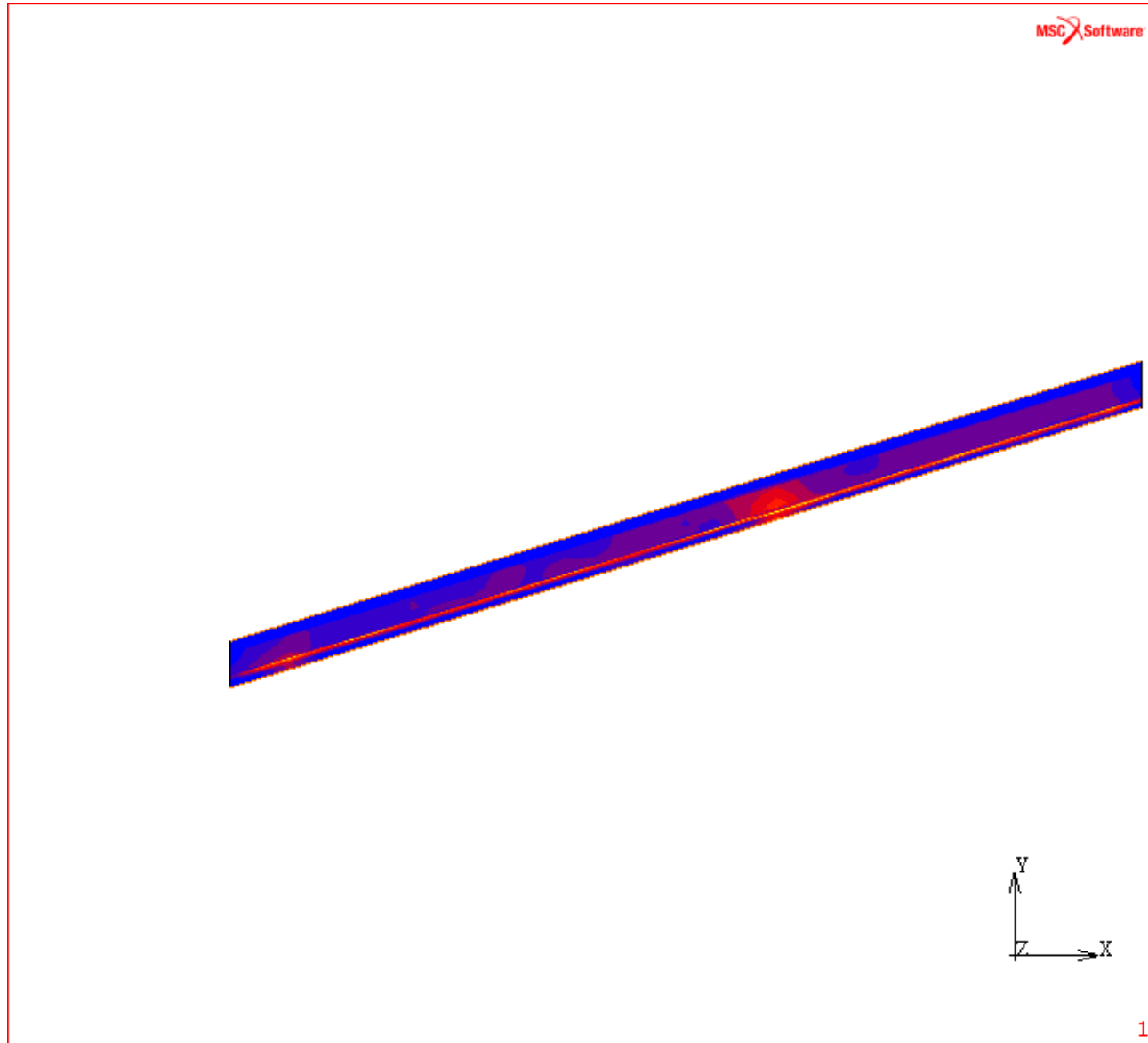


lcase2

Pore Pressure

1

Simplified Exit – Darcy Flow - Stress



Parameters

- TITLE
- ELEMENTS
- COUPLED
- STRUCTURAL
- HEAT
- PYROLYSIS
- DIFFUSION
- ABLATION
- ADAPTIVE
- LARGE STRAIN
- FOLLOW FORCE

Model Definition

- **Geometry**
- POINTS
- CURVES
- CONNECTIVITY
- COORDINATES
- ATTACH NODES
- ATTACH EDGES
- STREAM DEFINITION
- ADAPT GLOBAL
- CONTACT
- CONTACT TABLE
- THROAT
- CAVITY DEFINITION
- **Material Property**
- ISOTROPIC
- ORTHOTROPIC
- ORIENTATION
- THERMO PORE
- TABLES
- LATENT HEAT
- EMISSIVITY
- CREEP

Model Definition

- **Initial Conditions**
- INITIAL PYROLYSIS
- INITIAL TEMP
- INITIAL PRESSURE
- **Boundary Conditions**
- SURFACE ENERGY
- FILMS
- FIXED TEMP
- FIXED PRESS
- FIXED DISP
- DIST FLUXES
- DIST LOAD
- RAD-CAVITY
- RECEDING SURFACE



Model Definition

- Controls
- SOLVER
- DEFINE
- POST
- CONTROL
- LOADCASE

History Definition

- Controls
- TRANSIENT
- AUTO STEP
- CONTROL
- LOADCASE

Conclusions

- Thermal Stress Capability utilizing Mixture Model based upon Pyrolysis Material has been added to Marc
- Coupled Thermo-Pore-Structural has been developed
- Less User Effort in Solving Coupled Problems than transferring temperature data between different codes in simulations with ablation



Thank You