



## Water-Based Phase Change Material Heat Exchanger Development

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Thermal & Fluids Analysis Workshop  
TFAWS 2014  
August 4 - 8, 2014  
NASA Glenn Research Center  
Cleveland, OH



# Overview



- Why use Phase Change Material Heat Exchanger's (PCM HX's)?
- Prior PCM HX Development and Testing
- Copper HX Coupon Design and Testing
- Microgravity Flight Experiment
- Future Water-Based PCM HX Designs



# WHY PCM'S?

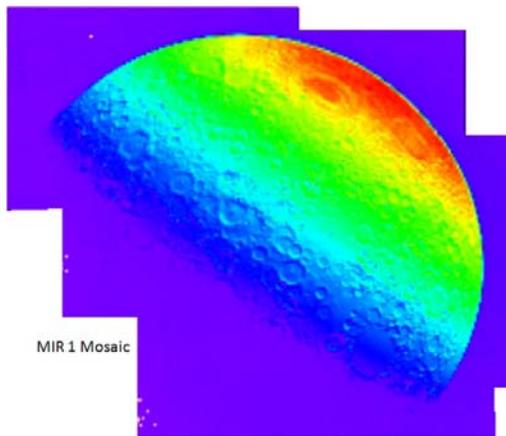


# Why Use PCM HX's

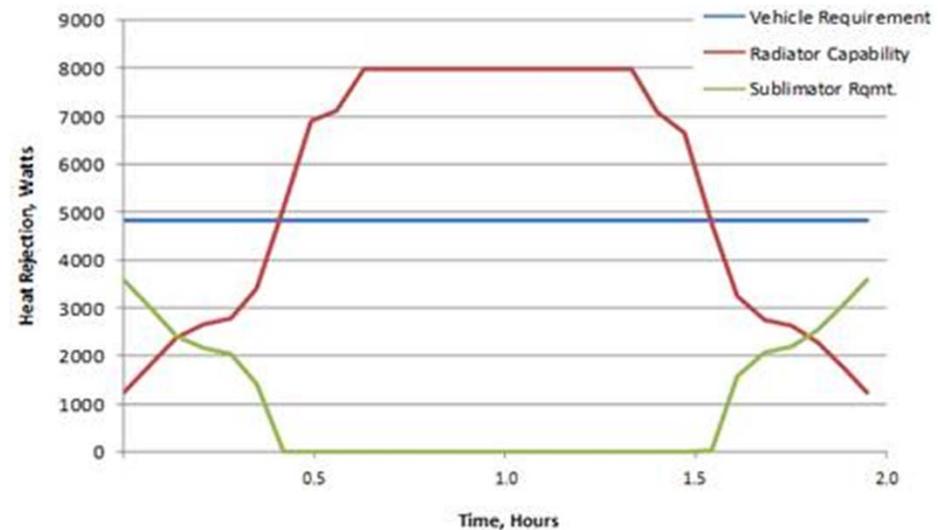


- In cyclical heat load environments, a Supplemental Heat Rejection Device (SHReD) is required
  - Typically, accomplished through evaporators, sublimators, or Phase Change Material Heat Exchangers (PCM HX)
  - PCM's act a thermal battery and do not use a consumable
- Wax PCM is baseline for the Orion Spacecraft, but water is being investigated
  - Water has significantly higher latent heat of fusion than wax (333 kJ/kg vs. 163 kJ/kg)
  - Significant mass and volume savings possible

Problem: Water expands ~10% when frozen



LCROSS IR Image of Lunar Surface





# **PRIOR PCM HX DEVELOPMENT AND TESTING**



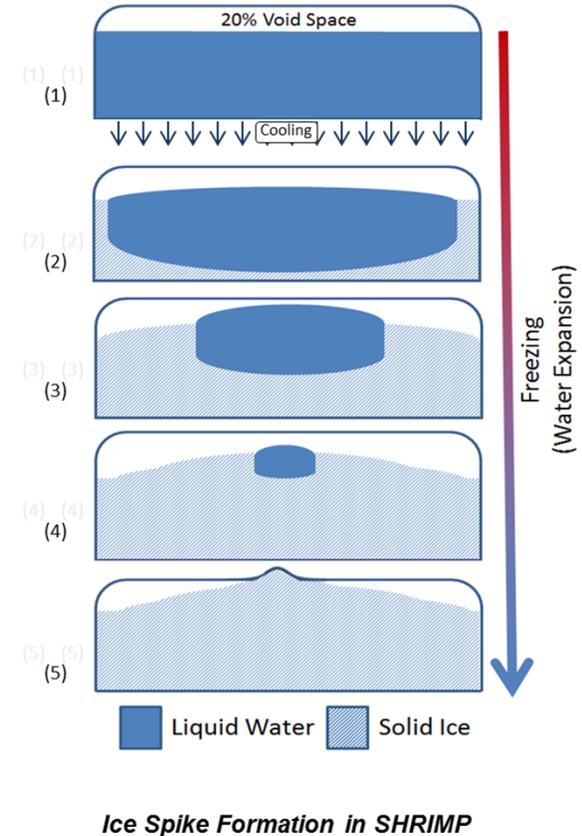
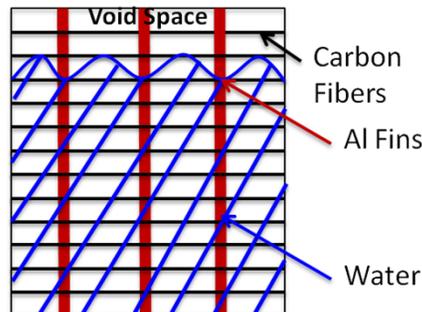
# PCM Testing History



- RIP/SHRIMP's testing focused on utilizing aluminum fins/carbon fibers to control void space location
  - RIP: 450 kJ (1.35 kg)
  - SHRIMP: 45 kJ (0.135 kg)
- Total of 13 RIP/SHRIMP's tested – all failing
  - Tested in various orientations
- Ultimately, knowing void space location is not sufficient
  - Even with 20% void space, test articles still failed
  - Void space will not necessarily be known in microgravity



*Ice Spike Formation Comparison in SHRIMP*



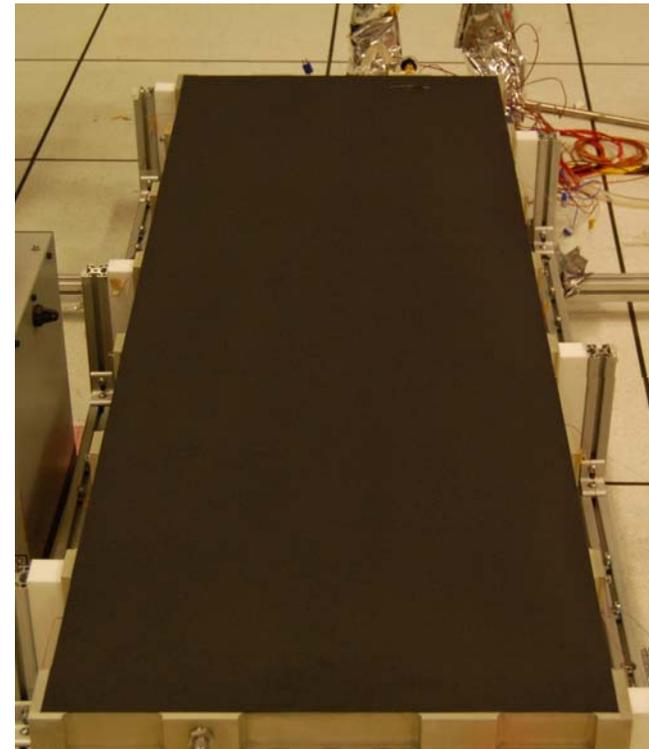
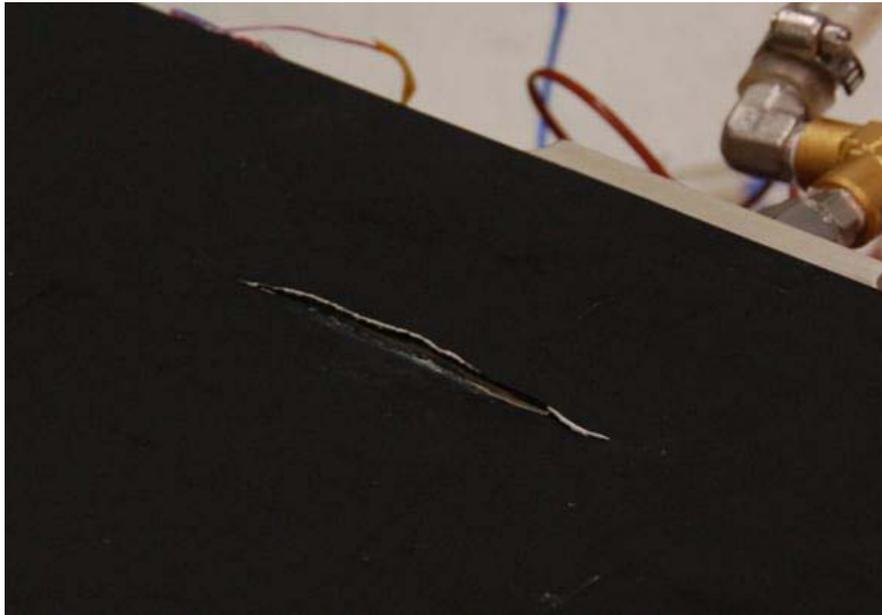
*Ice Spike Formation in SHRIMP*



# PCM Testing History



- Integrated Replicative Ice PCM (IRIP)
  - Full Scale 12.6 kW PCM for use on the Lunar Electric Rover
  - Consisted of aluminum brazed fins and 38 kg of water with 20% void space
  - 5-day vacuum test with thermal cycling
  - Resulted in failure (2" tear) on day 4

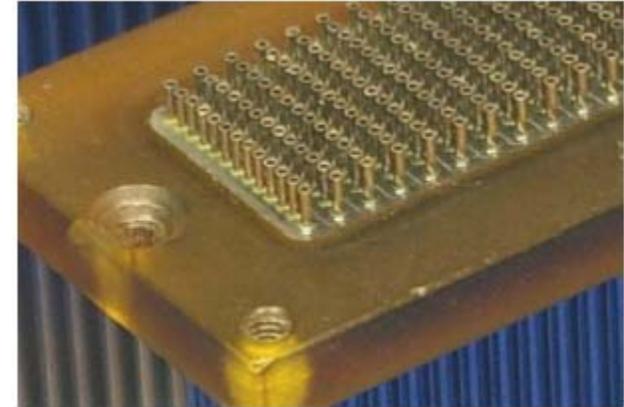




# Mezzo Technologies PCM



- **Microtube HX**
  - HX utilizing ~5,000 tubes positioned in a 4"x4" area
  - Originally used as a wax PCM for Lockheed
  - 19 total cycles in various orientations (favorable, unfavorable, neutral) with no visible signs of failure
- **Noticed volume of ice was greater when frozen in "unfavorable" orientation**



Favorable

Adverse



# **COPPER HX COUPON DESIGN & TESTING**



# Copper Test Articles



- Fabricated to:
  - Understand freeze front propagation and ice spike formation
  - Understand outside-in, inside-out, and uniform freezing
- 2 Outcomes:
  - Ice spike always occur where freezing occurs last
    - Void space (or deformable media) must be present at this location
  - Ice spike distribution should be considered in HX design



*Copper Gen 1*

*Copper Gen 2.0*

*Copper Gen 2.1*

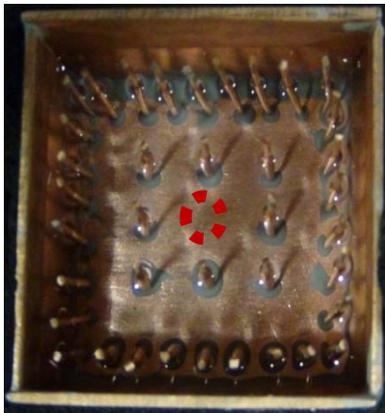
*Copper Gen 2.2*



# Ice Spike Distribution

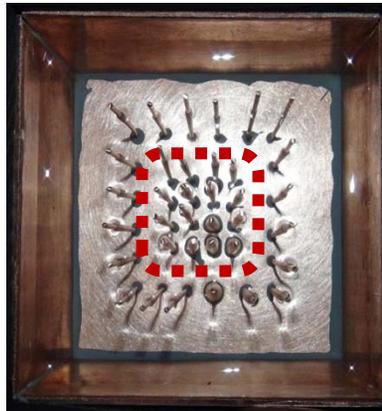


## Outside-In Freezing



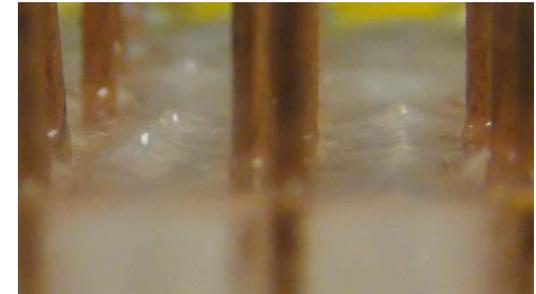
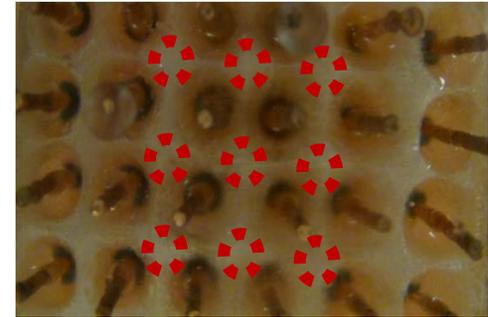
Single ice spike (0.25")  
formed at middle of test  
article

## Inside-Out Freezing



Ice spike ridge (0.18")  
formed at middle of test article

## Uniform Freezing



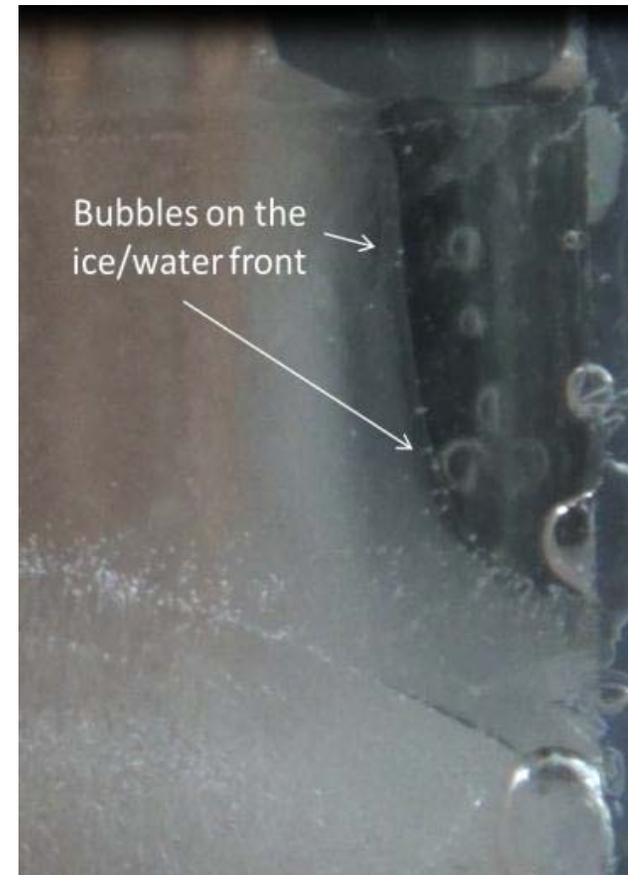
Several small ice spikes  
formed (<0.12")  
at intersections between  
copper rods



# Microgravity Flight Experiment



- Microgravity flight experiment developed in conjunction with University of Houston through NASA's MUREP Program
  - Individual water droplet study
  - Three copper coupon test articles
    - Continually frozen in various gravity loads
- **Results**
  - 1-g: Dissolved gasses escape from freeze front and float to surface
  - 0-g: Dissolved gasses escape from surface and float in place with some becoming entrapped in ice
- **Hypothesis**
  - Because air is trapped in the ice during zero-g a greater volume of ice will be formed when frozen in microgravity
    - Confirms Mezzo unfavorable testing results



*Gas Bubbles forming on the ice/water front*



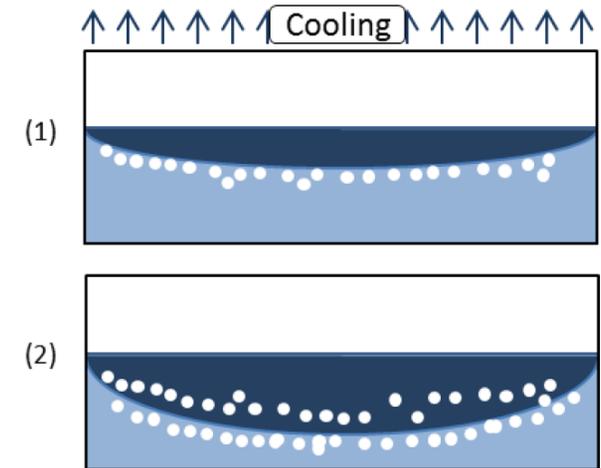
# Freeze Rate and Orientation



- Copper test article was frozen in liquid nitrogen to freeze quickly
  - Ice spike was approximately 0.5” in height compared to 0.25” when normally frozen.
- Hypothesis
  - Typically air bubbles are allowed to escape and float to surface
  - In quick freeze, dissolved gasses do not have sufficient time to escape and to surface, but become trapped in the freeze front



*Ice spike formation in liquid nitrogen*



Unfavorable (1-G)

*Dissolved gas freezing in ice*



# **FUTURE WATER-BASED PCM HX DESIGNS**



# Future Direction



- Currently working with Mezzo Technologies to develop a bladder based HX
- Utilizes a flexible bladder, ice spike distribution, and degassed water
- Two HX designs are currently being constructed
- Testing will be carried out in September 2014
- 1/10 scale will be tested in 2016 in microgravity on ISS

