

TFAWS 2016

Cryogenic Tank Pressurization and Liquefaction System Modeling

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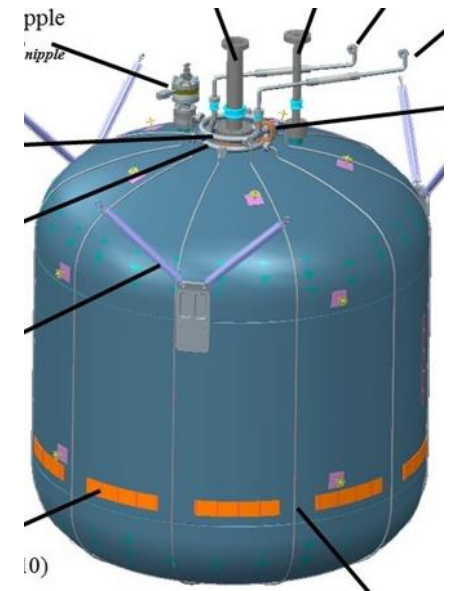
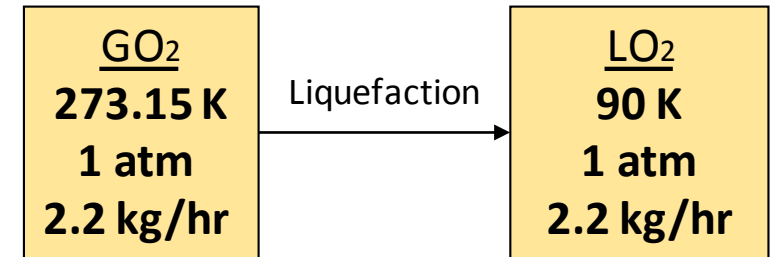
NASA Johnson Space Center

8/2/2016



ISRU Liquefaction Overview

- **PROBLEM:** Commodities from ISRU must be liquefied for efficient storage on surface of Mars
- **MAJOR TRADES:**
 - Liquefaction Methods
 - Tube on tank, tube in tank, conduction, in-line liquefaction, Linde Cycle
 - Where to Liquefy?
 - When to Liquefy?
- Infeasible to test everything
- **APPROACH:** Start off with cryogenic tank pressurization model



Tube on tank



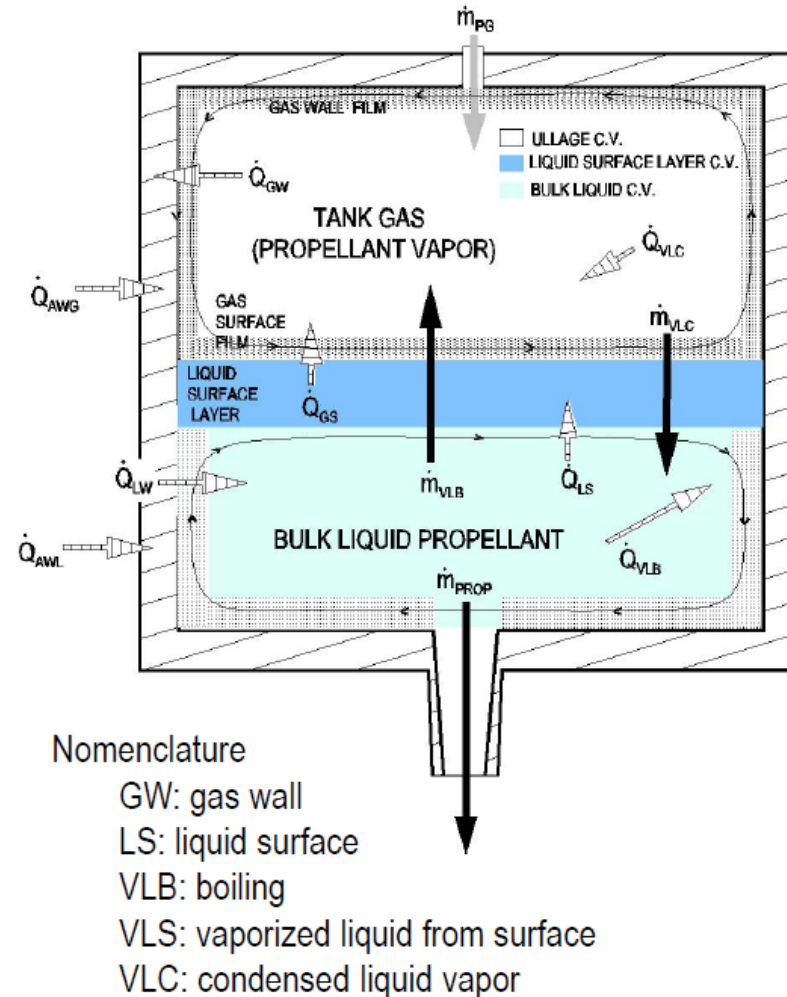
Tank Pressurization

- Heat Transfers

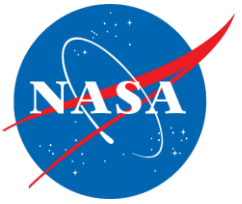
- \dot{Q} between liquid and tank wall touching liquid
- \dot{Q} between liquid and surface layer
- \dot{Q} between gas and surface layer
- \dot{Q} between gas and tank wall touching gas
- \dot{Q} between environment and wall touching liquid
- \dot{Q} between environment and wall touching gas

- Mass Transfers

- Mass transfer between liquid and gas
- Propellant leaving tank
- Gas entering tank



$$P_{tg} = \frac{m_{tg} T_{tg} R_u}{V_{tg}}$$

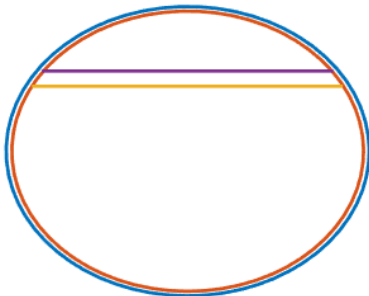


Modeling Approach

Tank Pressurization Model Comparison

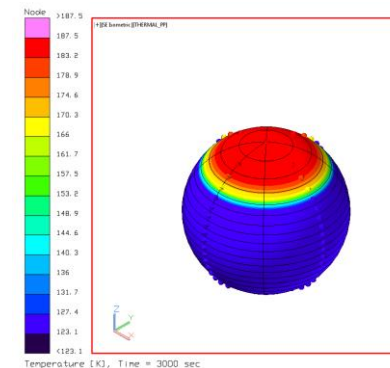
MATLAB

- First principles analysis
- Based on algorithm used for Shuttle External Tank
- Gas temperature uniform
- Temperatures of wall exposed to gas and liquid are uniform



Thermal Desktop

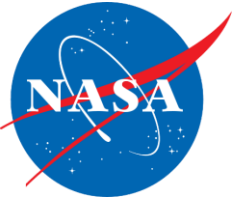
- Complex transient model
- Industry software, finite element based
- Gradient in wall temperature (nodal)
- Higher Fidelity than MATLAB Model
- Takes less time to build



Morpheus/Cold Helium Data

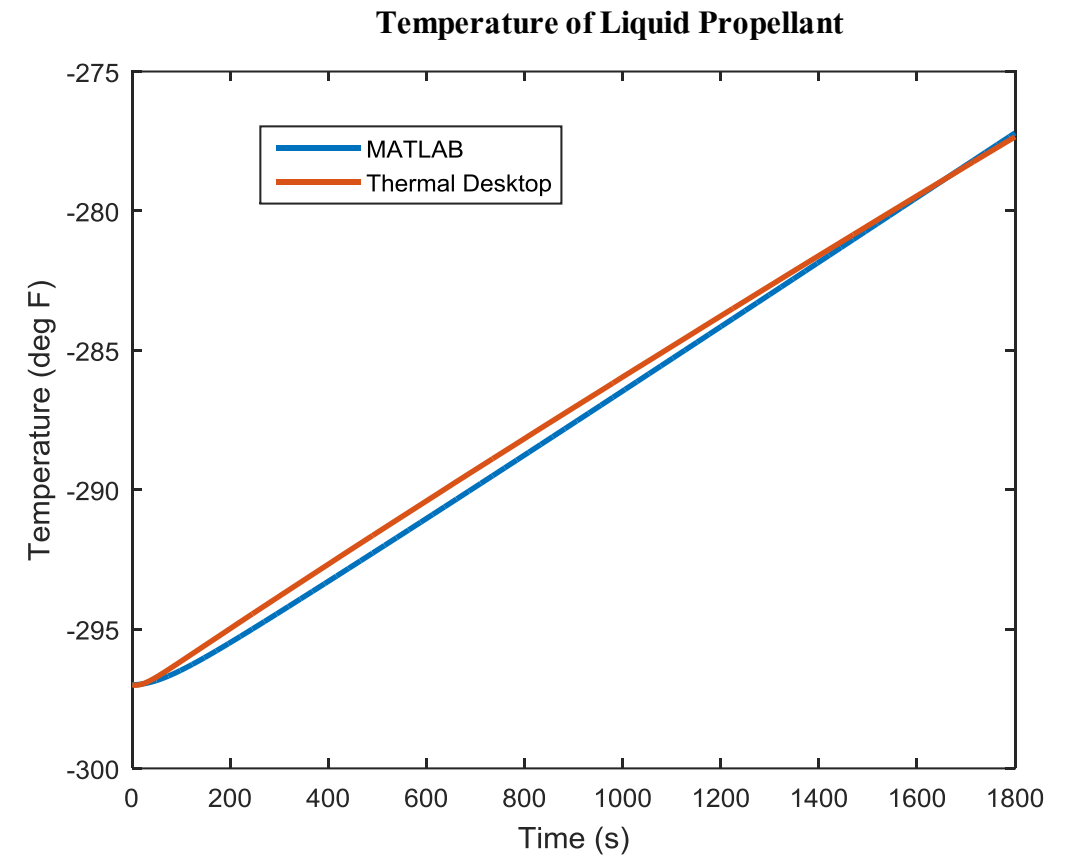
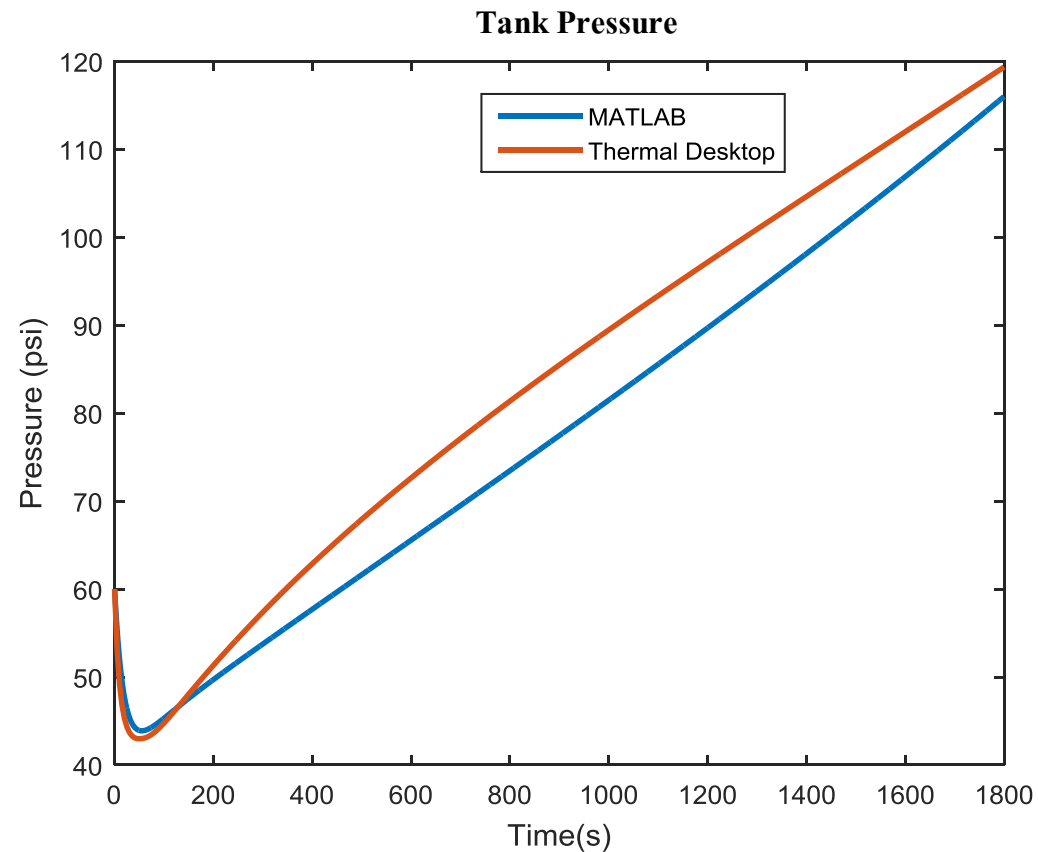
- Pressurization data from Cold Helium hot fires
- Anchor model

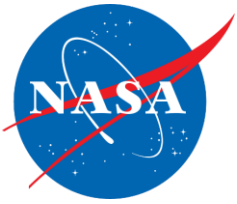




Comparison Results

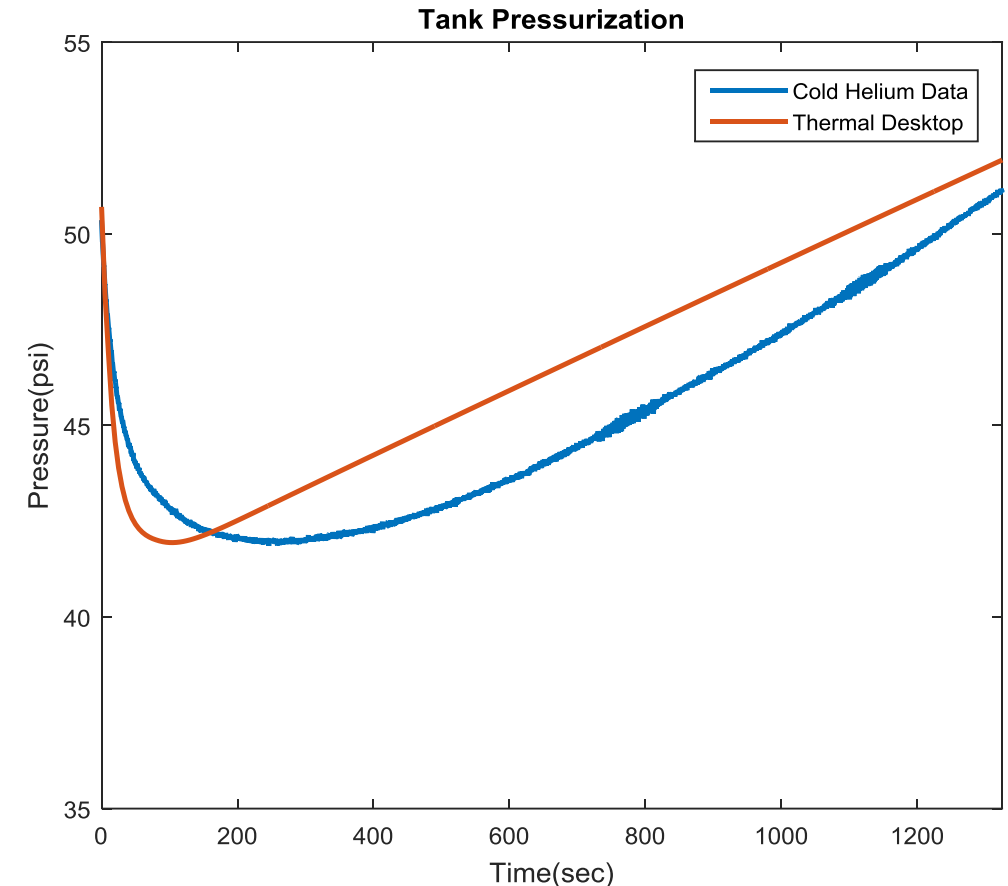
- MATLAB and Thermal Desktop Models compared well





Cold Helium Pressurization Data

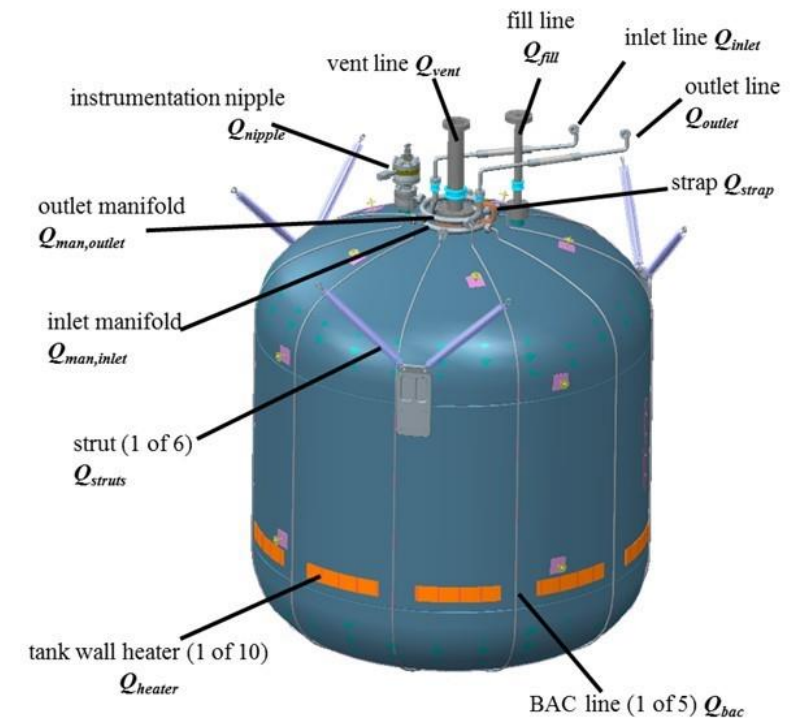
- Tank pressurization data available from Cold Helium tests
 - Tanks pressurized to ~50 psig, then closed off for ~25 min
- Initial Conditions
 - Ullage temperature: -125°F
 - Tank wall average temperature: -200°F
 - Ullage fraction: 26%
 - Propellant temperature: -297°F
- Next steps for model:
 - Model liquid, ullage thermal stratification





Tube on Tank Concept

- Broad area cooling distributes cooling with tubing network over the outer surface of tank
 - Cold gas (neon) is circulated in tubing loops in around cryo tank to eliminate boil-off
 - Tubing is welded and epoxied to tank wall
- Uses Reverse turbo-Brayton cycle



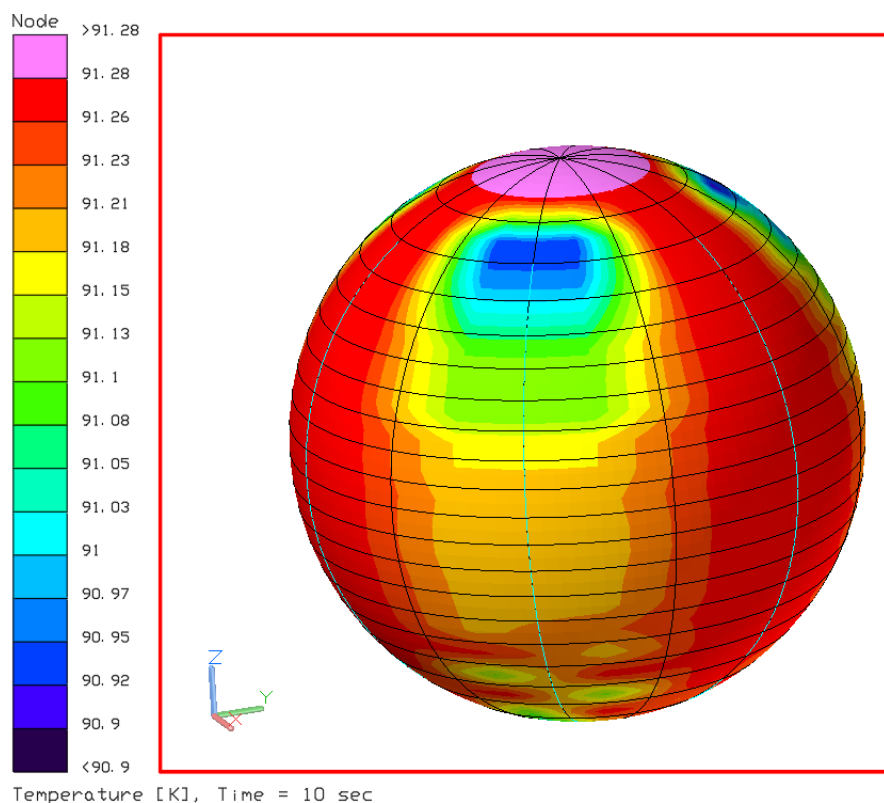
Zero Boil-off Tank



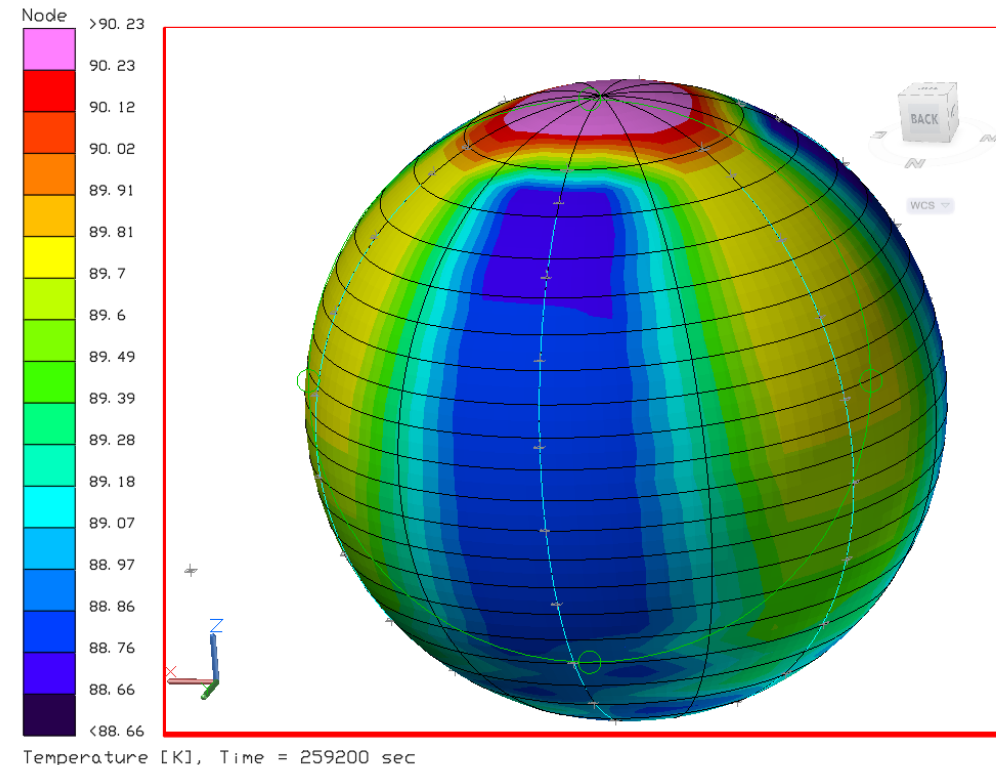
Current Work - Tube on Tank Transient Model



- Neon Gas flow in tubes
- Tubes and tank wall started at 90 K at time 0. initial ullage volume fraction: 99%



After 10 seconds



After 72 hours



Acknowledgements



- **AES LanderTech Liquefaction Team**

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