#### **TFAWS Active Thermal Paper Session**



**Calculation of Radiative Fin Performance Parameters for Radiator Analysis** Louis Jorski The Boeing Company

Presented By Louis Jorski

**ANALYSIS WORKSHOP** 

 $\mathbf{g}$ 

THERRY



**Not subject to US Export Administration Regulations (15 C.F.R. Parts 730-774) or US International Traffic in Arms Regulations (22 C.F.R. Parts 120-130)**

Thermal & Fluids Analysis Workshop TFAWS 2019 August 26-30, 2019 NASA Langley Research Center Hampton, VA





- Active radiators provide the primary source of heat rejection for long duration manned spacecraft
- Radiators consist of a series of flow tubes bonded to a highly conductive face sheet with a specialized surface coating to maximize its radiative heat loss
- The face sheet effectively acts as a series of fins which extends the surface areas of the individual flow tubes







- Temperature of the face sheet decreases as the distance from the flow tube increases due to the material's conductive resistance
- Reduces the amount of additional heat rejection gained by adding additional surface area
- Fin efficiency (η) provides a convenient measure of how effectively the face sheet extends the heated surface area
- Fin efficiency is the ratio of a fin's actual heat rejection to its theoretical maximum heat rejection that would be obtained if the entire fin was at its base temperature (infinite thermal conductivity)

$$
\eta = \frac{q_{Actual}}{q_{Max}}
$$

• Maximum theoretical heat rejection  $(q_{max})$  can be calculated directly from the net radiative heat transfer relationship:

$$
q_{Max} = \sigma \varepsilon L \Delta y (T_{base}^4 - T_{sink}^4)
$$





- Actual heat rejection is equal to the total amount of heat conducted through the fin's base
- Heat conduced through the base can be calculated from the temperature gradient at the base using Fourier's Law
- Requires solving a differential equation for the surface temperature distribution



#### **Fin Model**

# **Differential Equation Development**



$$
q_x = -k\delta \Delta y \frac{dT}{dx} \quad (Fourier's Law)
$$

 $q_{x+dx} = q_x +$  $dq_x$  $\frac{\partial u_x}{\partial x} dx$  (Taylor Expansion)

$$
q_{x+dx} = -k\delta \Delta y \frac{dT}{dx} - k\delta \Delta y \frac{d^2 T}{dx^2} dx
$$

**Fin Differential Element**

$$
q_{loss} = \sigma \varepsilon \Delta y \Delta x (T^4 - T_{sink}^4)
$$

$$
q_x = q_{x+dx} + q_{loss} \quad \text{(Conservation of Energy)} \qquad \text{Note: } \Delta x \approx dx
$$
\n
$$
-k\delta \Delta y \frac{dT}{dx} = -k\delta \Delta y \frac{dT}{dx} - k\delta \Delta y \frac{d^2 T}{dx^2} dx + \sigma \Delta y \Delta x \epsilon (T^4 - T_{sink}^4)
$$
\n
$$
\frac{d^2 T}{dx^2} = \frac{\sigma \epsilon}{k\delta} (T^4 - T_{sink}^4)
$$

NA SA





$$
\frac{d^2T}{dx^2} = \frac{\sigma \varepsilon}{k\delta} (T^4 - T_{sink}^4) \qquad \text{set} \qquad K_1 = \frac{\sigma \varepsilon}{k\delta} \qquad \text{and} \qquad K_2 = \frac{\sigma \varepsilon T_{sink}^4}{k\delta}
$$
\n
$$
\frac{d^2T}{dx^2} = K_1 T^4 - K_2
$$
\n
$$
\text{Boundary Conditions:} \qquad T(x = 0) = T_{Base} \qquad \frac{dT}{dx} |_{x=L} = 0
$$
\n
$$
\frac{dT}{dx} = -\left(\frac{2K_1 T^5}{5} - 2K_2 T + C\right)^{1/2} \qquad \text{(from integrating once)}
$$
\n
$$
C = \frac{-2K_1 T_{tip}^5}{5} + 2K_2 T_{tip} \qquad \text{(from applying second boundary condition)}
$$
\n
$$
\frac{dT}{dx} = -\left(\frac{2K_1}{5} (T^5 - T_{tip}^5) - 2K_2 (T - T_{tip})\right)^{1/2} \qquad \text{where } T_{tip} = T |_{x=L}
$$
\n
$$
-T_{tip} \text{ is unknown so this expression for dT/dx is not usable until } T_{tip} \text{ is found}
$$

– An additional integration must be performed so that the first boundary condition can be applied and used to solve for  $T_{tip}$ 





$$
\int \left(\frac{2K_1}{5}(T^5 - T_{tip}^5) - 2K_2(T - T_{tip})\right)^{-1/2} dT = -\int dx
$$

- Left hand integral cannot be solved analytically
- Numerical integration methods cannot be applied because the integrand is asymptotic at  $T = T_{\text{tip}}$
- Change of variable is required (proposed by Mackay and Leventhal)

$$
set \quad v^2 = \frac{T}{T_{tip}} - 1 \quad dT = 2T_{tip}v dv
$$

$$
2T_{tip}v\frac{dv}{dx} = -v\left(\frac{2K_1T_{tip}^5}{5}\right)^{1/2} \left((v^2+1)^4 + (v^2+1)^3 + (v^2+1)^2 + (v^2+1) + 1 - \frac{5K_2}{K_1T_{tip}^4}\right)^{1/2}
$$

$$
\int_{v_{min}}^{v_{max}} \frac{dv}{\left( (v^2 + 1)^4 + (v^2 + 1)^3 + (v^2 + 1)^2 + (v^2 + 1) + 1 - \frac{5K_2}{K_1 T_{tip}^4} \right)^{1/2}} = \int_0^L \left( \frac{K_1 T_{tip}^3}{10} \right)^{1/2} dx
$$

where  $v_{max} = v(T = T_B) = (T_B/T_{tip} - 1)^{1/2}$  $v_{min} = v(T = T_{tip}) = 0$ 

Not subject to US Export Administration Regulations (15 C.F.R. Parts 730-774) or US International Traffic in Arms Regulations (22 C.F.R. Parts 120-130)



## **Equation Solution**



$$
set \quad F(v) = \int_{v_{min}}^{v_{max}} \frac{dv}{\left( (v^2 + 1)^4 + (v^2 + 1)^3 + (v^2 + 1)^2 + (v^2 + 1) + 1 - \frac{5K_2}{K_1 T_{tip}^4} \right)^{1/2}}
$$
\n
$$
T_{tip} = \left[ \frac{10}{K_1} \left( \frac{F(v)}{L} \right)^2 \right]^{1/3} \quad or \quad 0 = T_{tip} - \left[ \frac{10}{K_1} \left( \frac{F(v)}{L} \right)^2 \right]^{1/3}
$$

– F(v) can be evaluated using standard numerical integration methods if a value for  $T_{\text{tip}}$  is assumed

- Maximum theoretical range for 
$$
T_{tip}
$$
:

 $T_{sink} < T_{tip} < T_{base}$ 

– F(v) complex or undefined for:

 $T_{tip} \leq T_{sink}$  and  $T_{tip} > T_{base}$  ( $F_v$  asymptotic at  $T_{base}$ )

- Solving for  $T_{\text{tip}}$  reduces the problem to solving a system of two equations or finding a root
- Solvable by some root finding algorithms in scientific computing software packages



#### **Tip Temperature Function**



Not subject to US Export Administration Regulations (15 C.F.R. Parts 730-774) or US International Traffic in Arms Regulations (22 C.F.R. Parts 120-130)

#### TFAWS 2019 – August 26-30, 2019 9

NH SI



# **Root Solving - Fixed Point Iteration**

- A fixed point is a value for which a function returns the same value
- By starting with an initial guess and substituting the resultant value back into a fixed point function, an accurate approximation for the fixed point can be obtained
- Procedure:

Pick  $T_{tip0}$ 

$$
T_{tip} = \left[\frac{10}{K_1} \left(\frac{F(v)}{L}\right)^2\right]^{1/3} with \ v(T_{tip0})
$$
  
While  $|T_{tip} - T_{tip0}|$  < *tolerance*  

$$
T_{tip} = T_{tip0}
$$

$$
T_{tip} = \left[\frac{10}{K_1} \left(\frac{F(v)}{L}\right)^2\right]^{1/3} with \ v(T_{tip0})
$$

• Rate of convergence and ability of the iteration to converge is highly dependent on the accuracy of the initial guess

**NAS** 





- Repetitively halve the possible range based on sign change until within the desired tolerance
- Procedure:

Set 
$$
T_{min} = T_{sink}
$$
  $T_{max} = T_{base}$   $T_{tip} = 0.5(T_{max} + T_{min})$   
\n
$$
\mathbb{F}_{max} = T_{max} - \left[\frac{10}{K_1} \left(\frac{F(v)}{L}\right)^2\right]^{1/3}
$$
 with  $v(T_{max})$   
\nWhile  $0.5(T_{max} - T_{min}) <$  tolerance  
\n
$$
\mathbb{F}_{tip} = T_{tip} - \left[\frac{10}{K_1} \left(\frac{F(v)}{L}\right)^2\right]^{1/3}
$$
 with  $v(T_{tip})$   
\nIf  $\mathbb{F}_{tip} \mathbb{F}_{max} > 0 \Rightarrow T_{max} = T_{tip}$   $\mathbb{F}_{max} = \mathbb{F}_{tip}$   
\nElse  $T_{min} = T_{tip}$   
\n $T_{tip} = 0.5(T_{max} + T_{min})$ 

• Iteration will always converge if a root exists in the initial range



– With  $T_{tip}$  known, the actual heat rejection can be calculated using Fourier's Law and the base temperature gradient

$$
q_{Actual} = -k\delta \Delta y \frac{dT}{dx}|_{x=0} \quad (Fourier's Law)
$$

$$
q_{Actual} = k\delta \Delta y \left(\frac{2K_1}{5}(T_{base}^5 - T_{tip}^5) - 2K_2(T_{base} - T_{tip})\right)^{1/2}
$$

- The normalized heat rejection  $q_{Actual}/\Delta y$  is itself a valuable parameter
- With a valid expression for  $q_{\text{Actual}}$ , the fin efficiency can be calculated

$$
\eta = \frac{k\delta \left(\frac{2K_1}{5}(T_{base}^5 - T_{tip}^5) - 2K_2(T_{base} - T_{tip})\right)^{1/2}}{\sigma \varepsilon L (T_{base}^4 - T_{sink}^4)}
$$





NH SI



• Once  $T_{tip}$  is known, the temperature  $(T_i)$  at points  $(x_i)$  along the fin can also be solved for using a similar methodology

$$
T_{tip} = \left[\frac{10}{K_1} \left(\frac{F(v)}{L_i}\right)^2\right]^{1/3} \qquad or \qquad 0 = T_{tip} - \left[\frac{10}{K_1} \left(\frac{F(v)}{L_i}\right)^2\right]^{1/3}
$$

$$
F(v) = \int_{v_{min}}^{v_{max}} \frac{dv}{\left( (v^2 + 1)^4 + (v^2 + 1)^3 + (v^2 + 1)^2 + (v^2 + 1) + 1 - \frac{5K_2}{K_1 T_{tip}^4} \right)^{1/2}}
$$

$$
v_{max} = v(T = T_B) = \left(\frac{T_{base}}{T_{tip}} - 1\right)^{1/2}
$$
  $v_{min} = v(T = T_i) = \left(\frac{T_i}{T_{tip}} - 1\right)^{1/2}$ 

NA S



### **Surface Temperature Distribution**



Not subject to US Export Administration Regulations (15 C.F.R. Parts 730-774) or US International Traffic in Arms Regulations (22 C.F.R. Parts 120-130)

#### TFAWS 2019 – August 26-30, 2019 15

NA SA





- Efficiency curve is a function of ε, k,  $\delta$ , T<sub>base</sub>, and T<sub>sink</sub>
- For a given design, ε, k, and δ will be constant but  $T_{base}$  and  $T_{sink}$  will vary depending on system operation
	- k and δ have identical effects on fin performance
	- $-$  T<sub>sink</sub> has very little impact on efficiency for high efficiency fins ( $\eta >$  ~0.75)
- Decreased efficiency does not necessarily mean decreased heat rejection



### **Efficiency Curve Sensitivity**



Not subject to US Export Administration Regulations (15 C.F.R. Parts 730-774) or US International Traffic in Arms Regulations (22 C.F.R. Parts 120-130)

TFAWS 2019 – August 26-30, 2019 17

NH SI

![](_page_17_Picture_0.jpeg)

## **Summary**

![](_page_17_Picture_2.jpeg)

- Spacecraft radiator surfaces effectively function as a series of fins
- Fin efficiency is a useful measure of radiator surface performance
- Actual heat rejection and fin efficiency can be calculated analytically if the fin tip temperature is known
- Fin tip temperature can be found using iterative numerical methods
- Surface temperatures can be calculated using similar numerical methods once the tip temperature is known
- Fin performance is a function of ε, k,  $\delta$ ,  $T_{base}$ , and  $T_{sink}$

![](_page_18_Picture_0.jpeg)

![](_page_18_Picture_1.jpeg)

### **Questions**

Not subject to US Export Administration Regulations (15 C.F.R. Parts 730-774) or US International Traffic in Arms Regulations (22 C.F.R. Parts 120-130)

![](_page_19_Picture_0.jpeg)

#### **References**

![](_page_19_Picture_2.jpeg)

- Burden, Richard L. Faires, J Douglas. *Numerical Analysis*, 2011
- Incropera, Frank P. Dewitt, David P. Bergman, Theodore L. Lavine, Adrienne S. *Fundamentals of Heat and Mass Transfer*, 2007
- Kraus, Allan D. Aziz, Abdul. Welty, James. *Extend Surface Heat Transfer*, 2001