### **TFAWS Passive Thermal Paper Session**



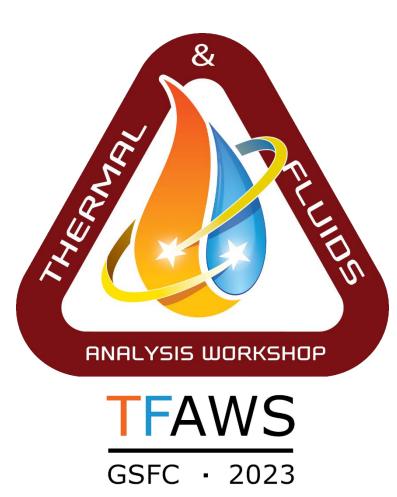
# Autonomous Melting Probe for Icy World Exploration

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### Advanced Cooling Technologies Inc.

Presented By: Calin Tarau

Thermal & Fluids Analysis Workshop TFAWS 2023 August 21-25, 2023 NASA Goddard Space Flight Center Greenbelt, MD





### Introduction

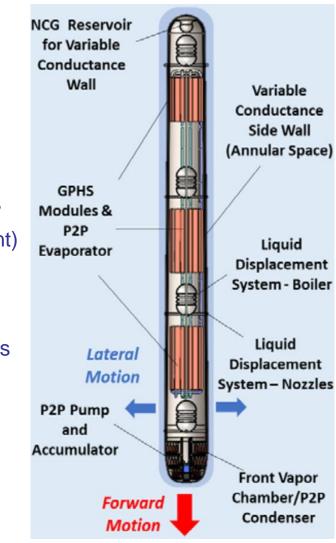


- Extraterrestrial ocean worlds are the most promising places for any signs of life beyond our planet Earth
- NASA, ESA and other space agencies have shown considerable interest in studying them
- Of specific interest is Jupiter's moon Europa
  - NASA JPL's Galileo spacecraft indicated liquid water may exist ~30 km underneath icy surface
- One technique to penetrate icy crust is to use thermal probe with heated front
- Environmental conditions on Europa present unique challenges for thermal probe
  - Low pressure on icy surface leading to sublimation after melting
  - High thermal conductivity of low-temperature ice makes melting process difficult
  - Heat "runs away" from source, not allowing surrounding ice to reach melting point
- Key requirements for radioisotope-powered ice melting probe
  - Minimum penetration time and vertical footprint
  - Maximize forward melting heat
- Advanced Cooling Technologies, Inc, (ACT) developed a thermal management concept for the ice melting probe
  - NASA Phase-II SBIR Program





- ACT's proposed thermal features inside the probe for efficient and reliable ice penetration :
  - Pumped Two-Phase (P2P)
  - Vapor Chamber
  - Variable Conductance Walls
  - Liquid Displacement Steering System (LDSS)
- Heat Source modules (GPHS or CPHS) provide heat to thermoelectric converters (TECs)
- Waste heat (~9 kW) is collected by P2P evaporators in contact with the cold end of the TECs
- Heat collected by P2P loop is transported to P2P condenser located at front end (melting front)
- P2P condenser rejects heat into the front Vapor Chamber
- Vapor Chamber focuses heat to front end of the probe for forward ice melting
- Tubular extensions from vapor chamber to top end of probe form Variable Conductance Walls
- Variable Conductance Walls provide side wall heating if:
  - Probe is stuck due to lateral freezing
  - Probe is stuck due to nonmeltable obstacle (and steering is needed)
- Liquid Displacement System provide lateral steering capability for maneuvering/steering



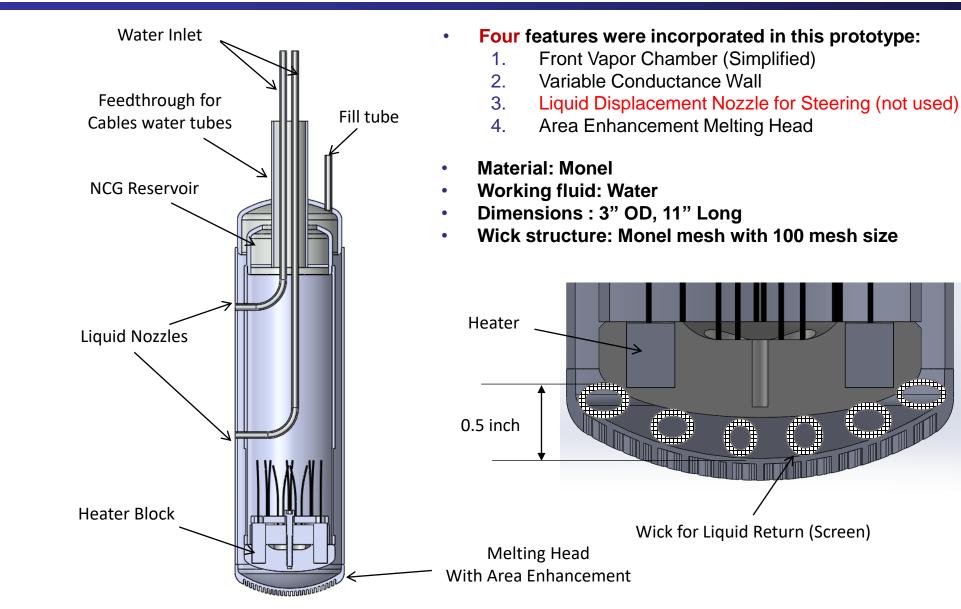




# Phase I work

# **Reduced-Scale Ice Melting Probe for Concept Demonstration**





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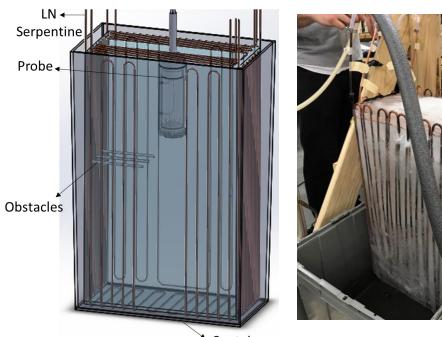
Ice melting probe prototype features



Area Enhancement Fins



- Ice block dimensions: 16" x 8" x 36"
- Enclosed by 6 LN serpentines (i.e. heat guard), which can provide subcooling and refreezing capability
- Can reach -160° C subcooled ice condition



Container

# **Thermocouple Map and Preliminary Test Results**



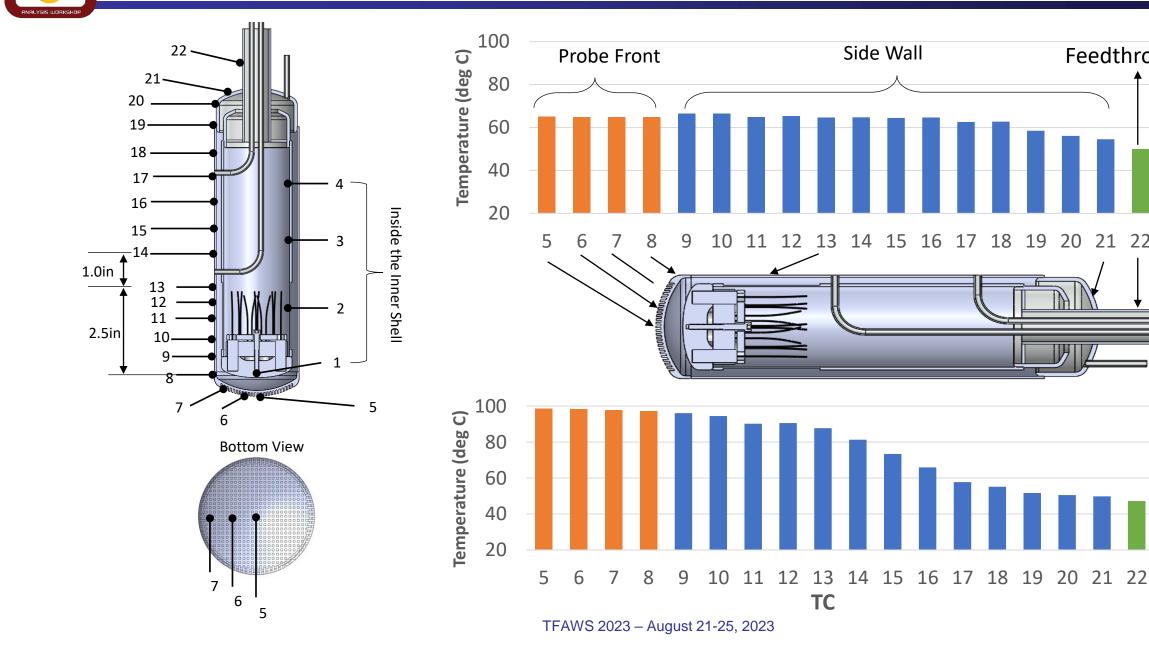
No NCG

Q=33W

Feedthrough

21 22

20



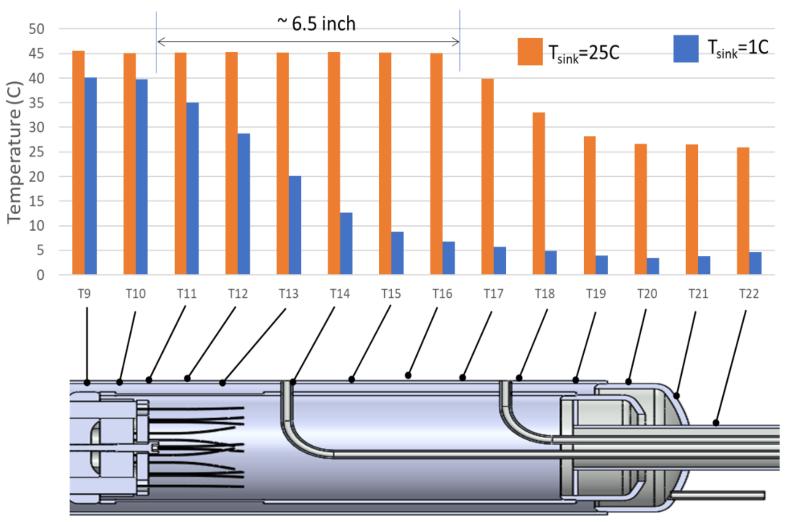
With NCG

(He)

Q=33W



### Instantaneous temperature profiles of the probe under two sink (water) conditions

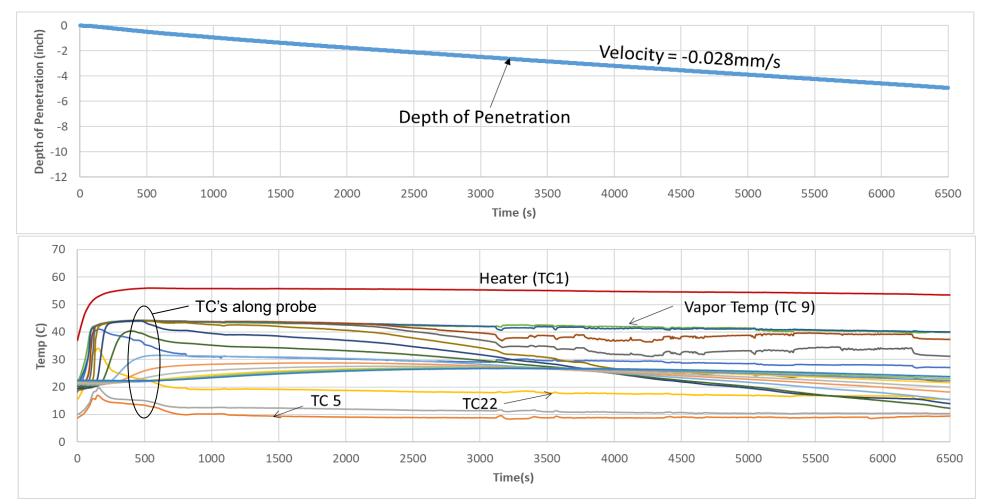


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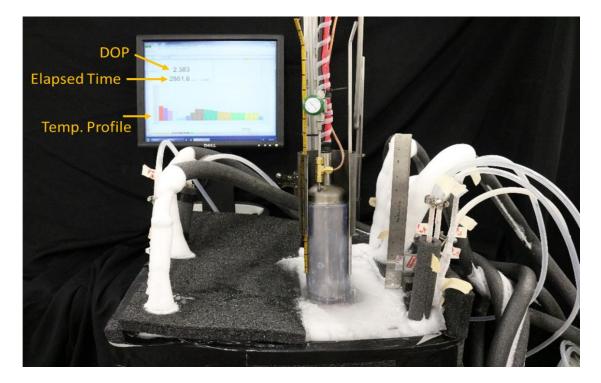
 Comparison between theoretical results and test data Lower plot shows the corresponding temperature evolution



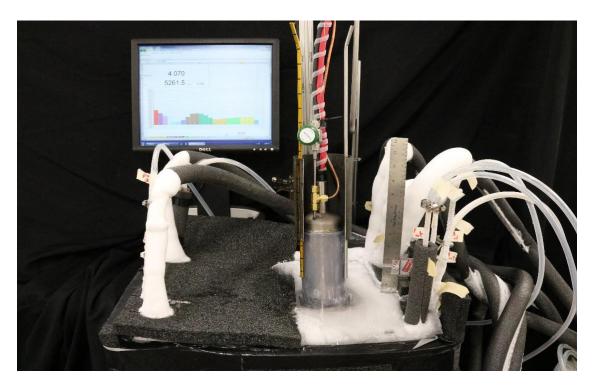


### **First Ice Penetration Test**





### Half of probe body is submerged into the ice (t=5281s)







### **Probe Refreezes and Self-Releases and meets Obstacle**

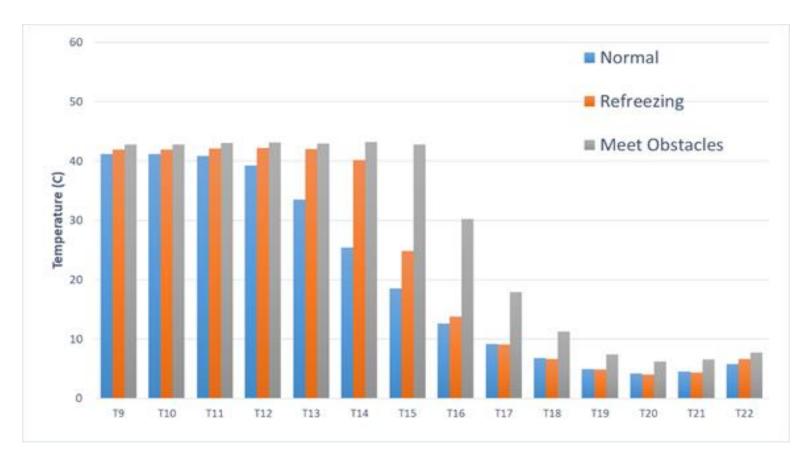


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### **Probe Refreezes and Self-Releases and meets Obstacle**





### **Third Ice Penetration Test**







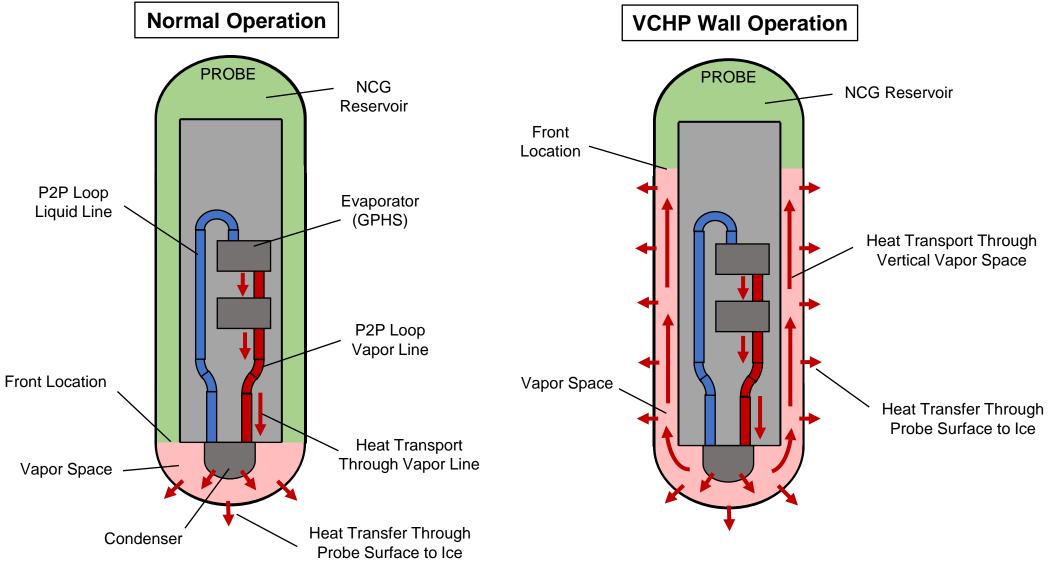


# Phase II work



### **Phase II Concept Overview**

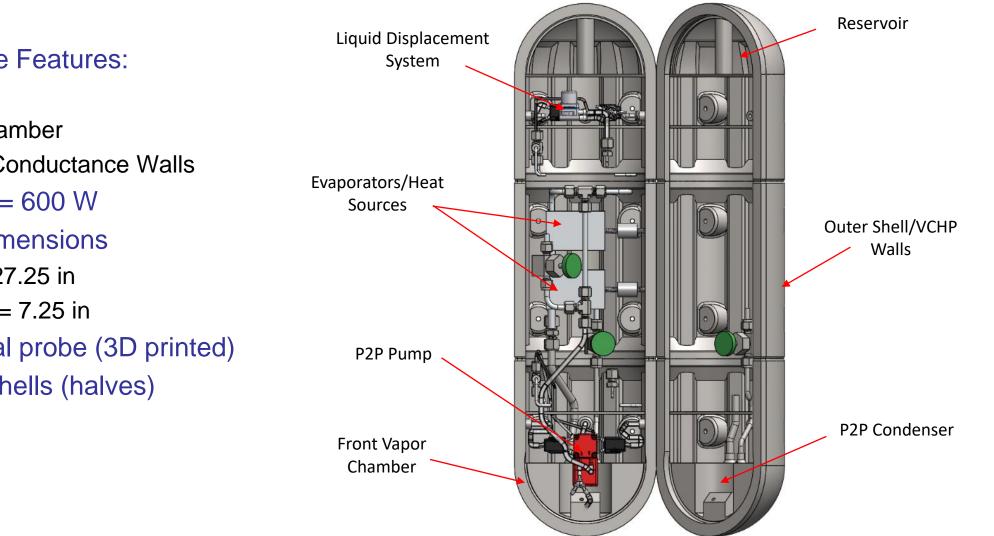






# **Phase II Melting Probe Prototype Development**





- 1st Prototype Features: ullet
  - P2P Loop
  - Vapor Chamber
  - Variable Conductance Walls
- Total power = 600 W۲
- Prototype dimensions
  - Height = 27.25 in
  - Diameter = 7.25 in
- SS 316 metal probe (3D printed)
- 2 separate shells (halves) ۲



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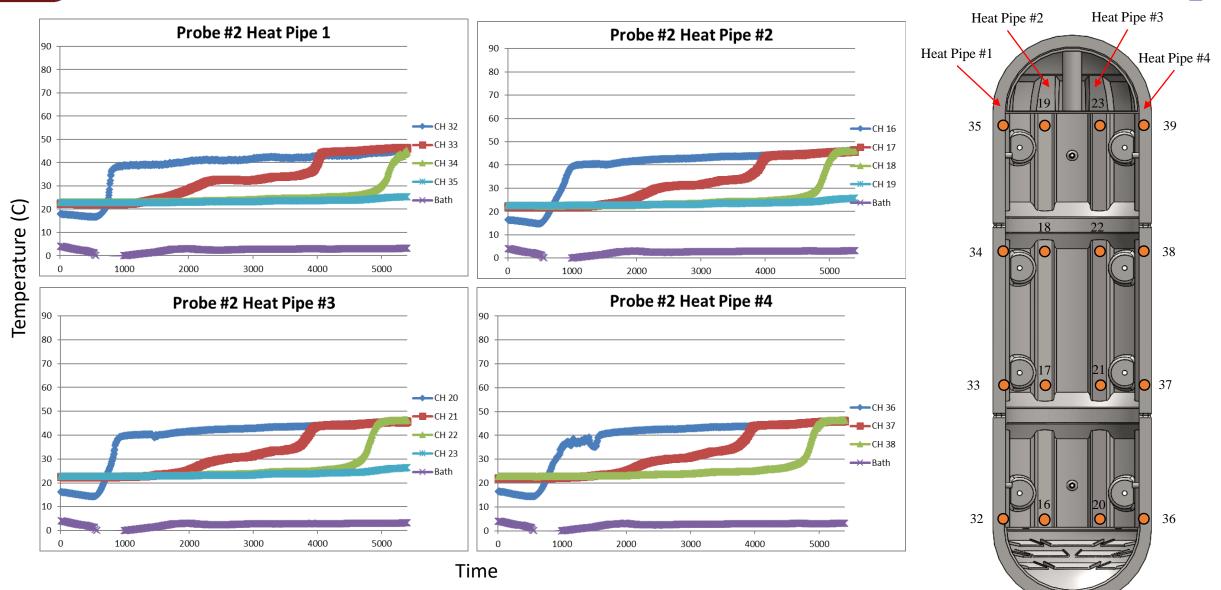
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### **Thermal Performance Testing**



#### Probe Half #2 Probe Half #1 **Testing Parameters:** NCG P2P Loop P2P loop reservoir charging Acetone working fluid valve 120cc charge Front Vapor Chamber Cartridge heaters Methanol working fluid **Evaporators** 40cc charge Front vapor **VCHP NCG** chamber charging valves – Argon 4.5 psia total pressure Power = 500WDiaphragm pump 0°C bath temp Fluid recirculation line Water-glycol "ice" bath LN line

RNRLYSIS LIDEKSHOP



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NASA



# **Large-Scale Experimental Setup**

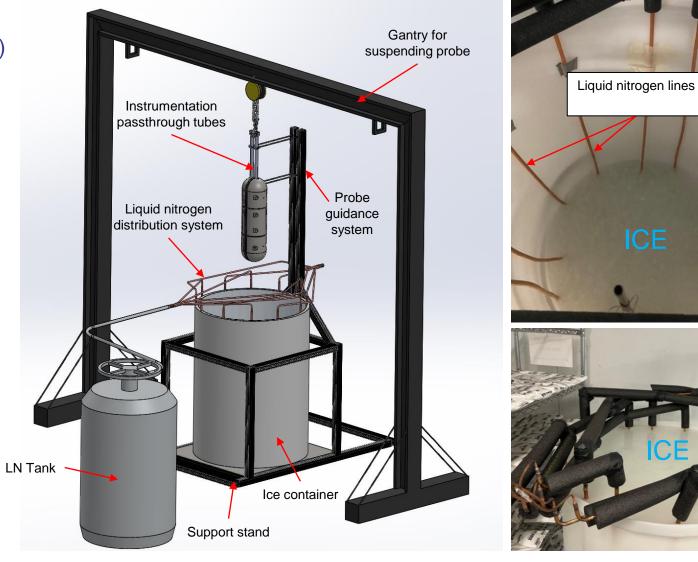


### Testing:

- Regular downward descent (low subcooling)
  - Constant melting
  - Both temperature and descent velocity recorded
- Simulated Obstacle
  - Suspend in place once fully submerged
  - Demonstrate VCHP wall/feature functionality

#### **Instrumentation:**

- TC placement
  - Key probe internals
  - Imbedded in ice
- Pressure transducer on P2P loop
- Displacement sensor to measure speed of descent

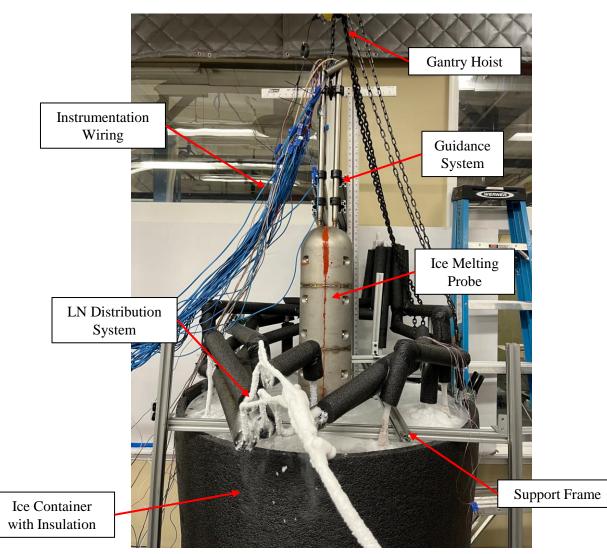


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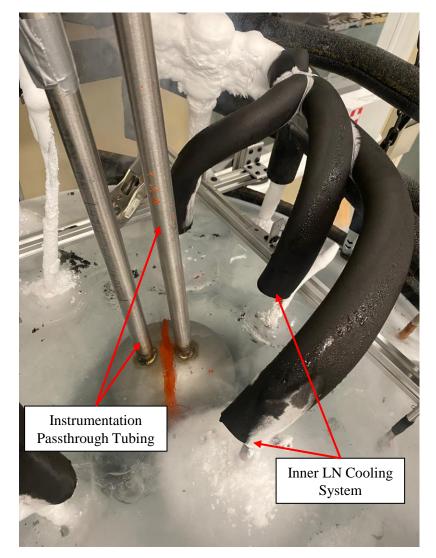


### **Stages of Large-Scale Ice Melting Test**





Probe prior to melting into ice

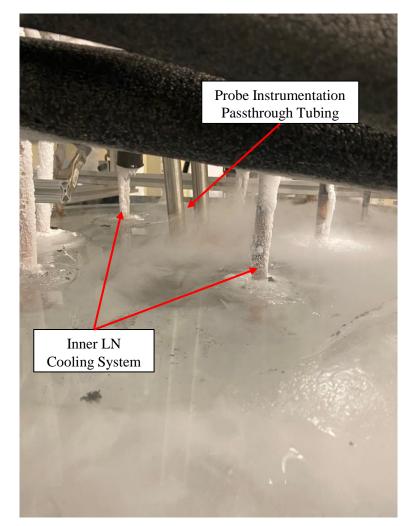


Probe after melting into ice block

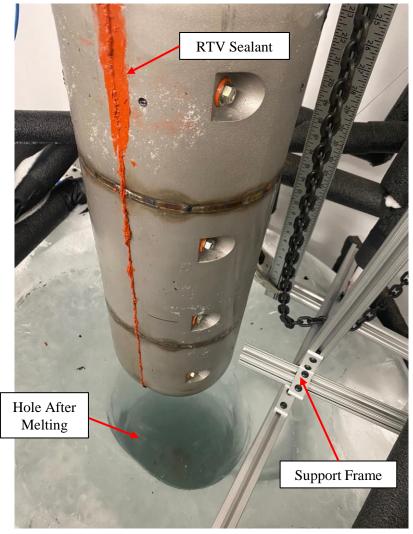


### **Stages of Large-Scale Ice Melting Test**





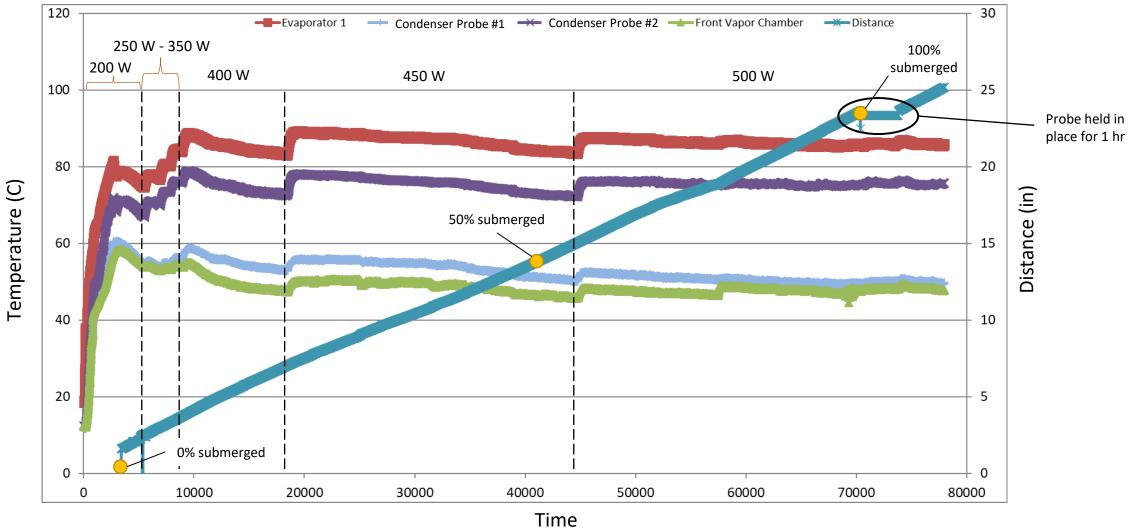
Probe fully submerged under ice surface



Probe removed after ice melting test

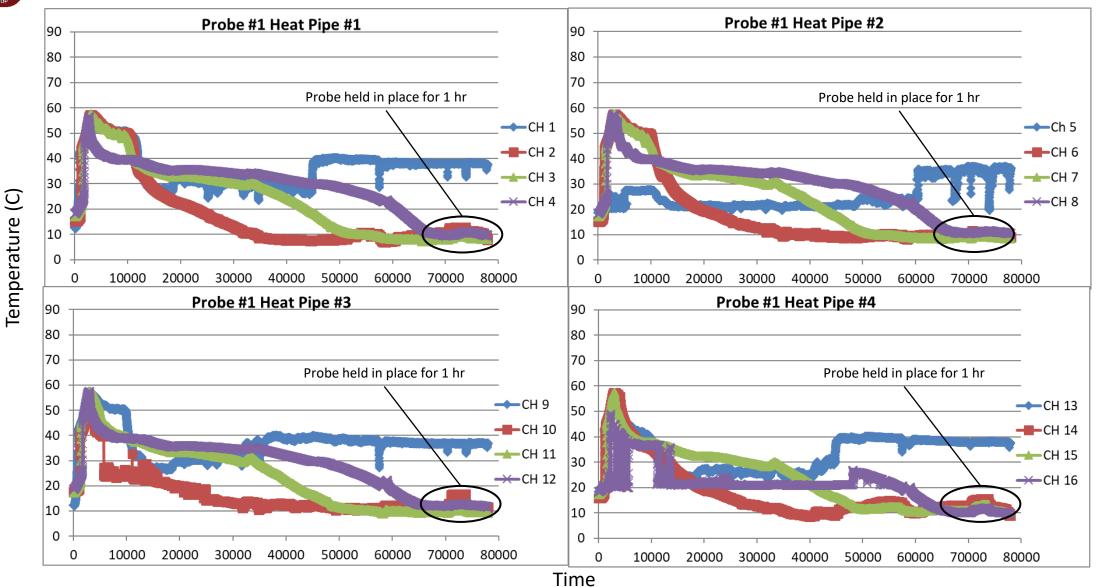


Probe Component Temperature Profiles



# VCHP Wall Performance @ -26°C Ice Temperature





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	Test #1	Test #2	Test #3
Bulk Ice Temperature (°C)	-26	-21	-2
Total Melt Distance (in)	26	26	26
Max Power Reached (W)	500	600	600
Total Melt Time (sec)	70,000	56,000	54,500
Average Melting Speed (in/hr)	1.34	1.67	1.72
Total Energy Deposited (Joules)	3.04E7	3.02E7	2.72E7
Specific Velocity (in/hr/Joule)	4.41E-8	5.53E-8	6.33E-8





The thermal management architecture consists of multiple novel features that can offer following advantages:

- A *pumped two-phase* heat delivery system can uniformly acquire the waste heat from multiple GPHS modules and transport the waste heat to the vapor chamber with minimal temperature drop and using minimal pumping power.
- A <u>front vapor chamber</u> can effectively transfer heat from P2P condenser to the melting head with minimized thermal resistance. The heat transfer performance of front vapor chamber can be further improved with elongated nose design and area enhancement features.
- **Variable conductance wall** that can passively control heat dissipating area to achieve maximized forward melting during normal mode and provide lateral melting capability when the probe was stuck in the ice.
- Liquid displacement-based steering system was not developed/demonstrated





- This project is sponsored by NASA Jet Propulsion Laboratory (JPL) under an SBIR Phase II program (Contract# 80NSSC20C0178).
  - NASA Technical Monitors:
    - Terry Hendricks (until 2021)
    - Benjamin Furst (until early 2022)
    - Benjamin Hockman (2022-2023)
- Contributors at ACT:
  - Technician: Tyler Spinelli, Larry Waltman, Justin Boyer, Phil Texter
  - ACT Safety Committee

### THANK YOU FOR YOUR ATTENTION