



Thermal Radiative Modeling of Spacecraft Windows in Future Human-Rated Spacecraft

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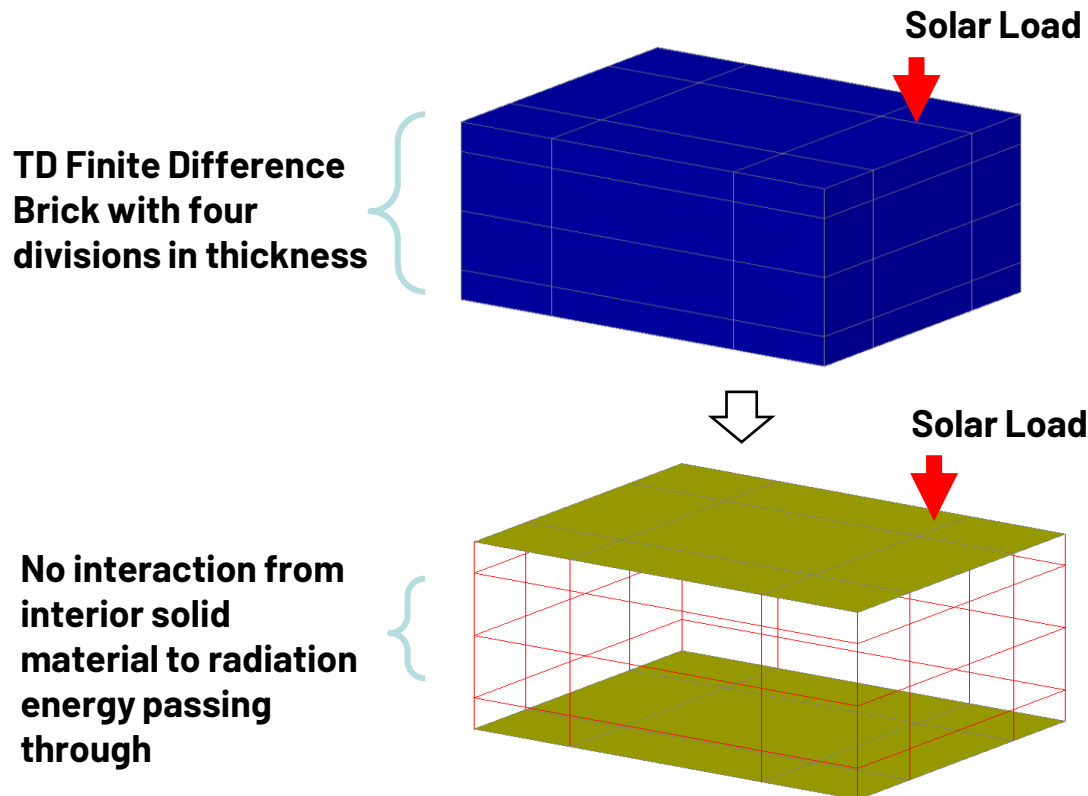
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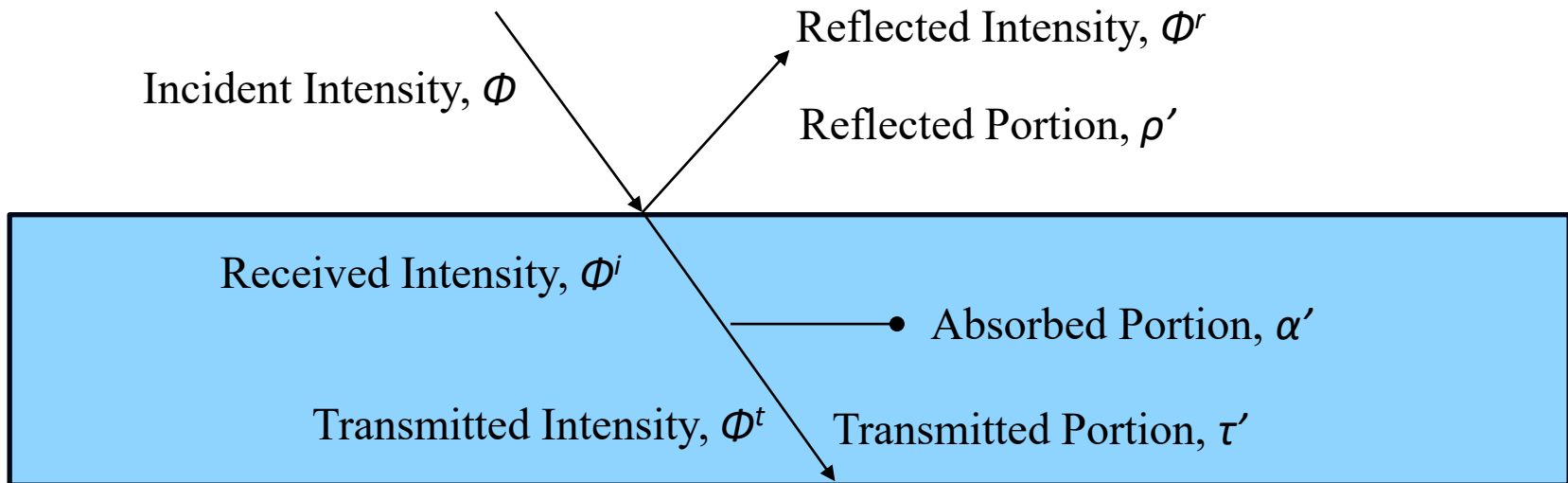
- Blue Origin has the vision for millions of people living and working in space for the betterment of Earth.
- Those folks will want to look outside back at Earth.
- Spacecraft window design for clear viewing and safety will therefore be paramount



- Window thermal analysis is key to this objective and should account for conduction and radiation absorption through the window solid materials.

- 1** Optical property definition
↓
- 2** Single pane analysis by ray trace method
↓
- 3** System level analysis by net radiation method
↓
- 4** Solve for absorption per node within panes
↓
- 5** Implement in Thermal Desktop

- Adjusted transmissivity, τ' , and adjusted reflectivity, ρ' , are the building blocks to the tool.
- Properties shown below apply to a single incident ray.
- Properties must be solved for from testing data.

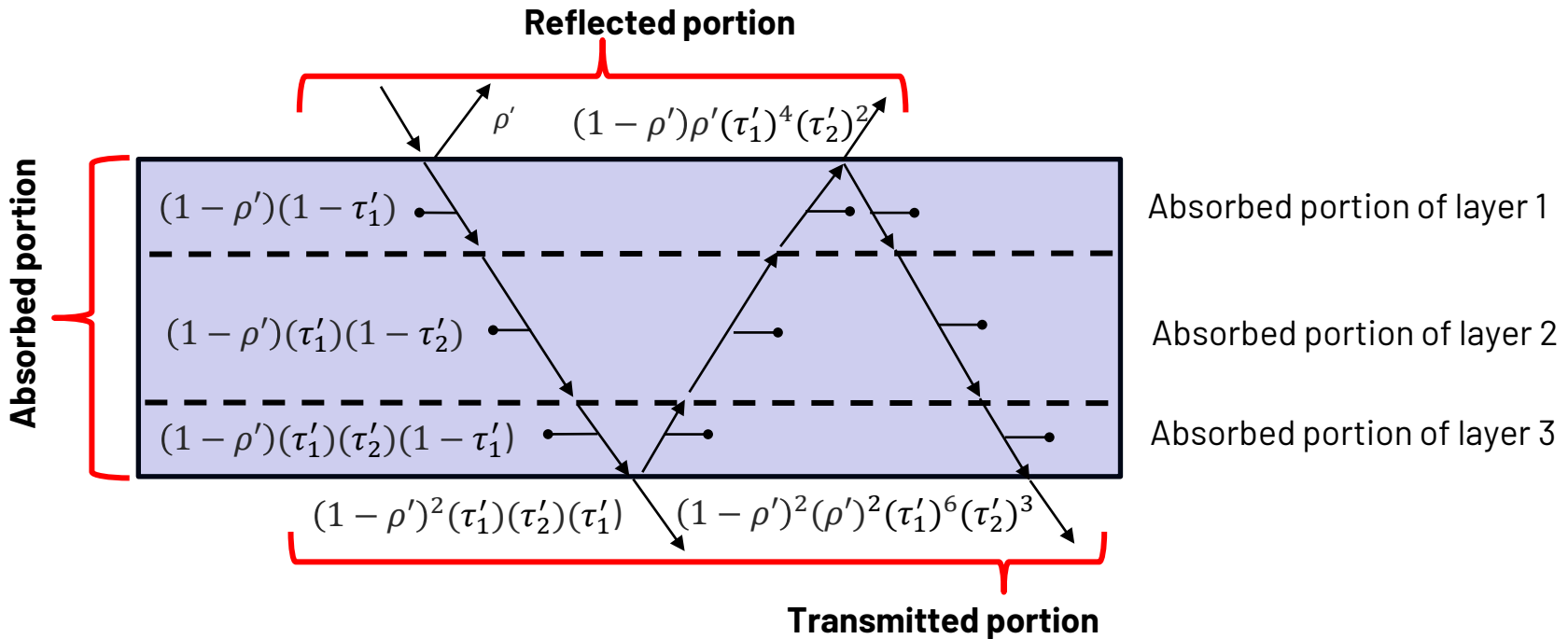


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Optical Props

- Single Pane Analysis
- System Analysis
- Solve for Absorption
- TD Implementation

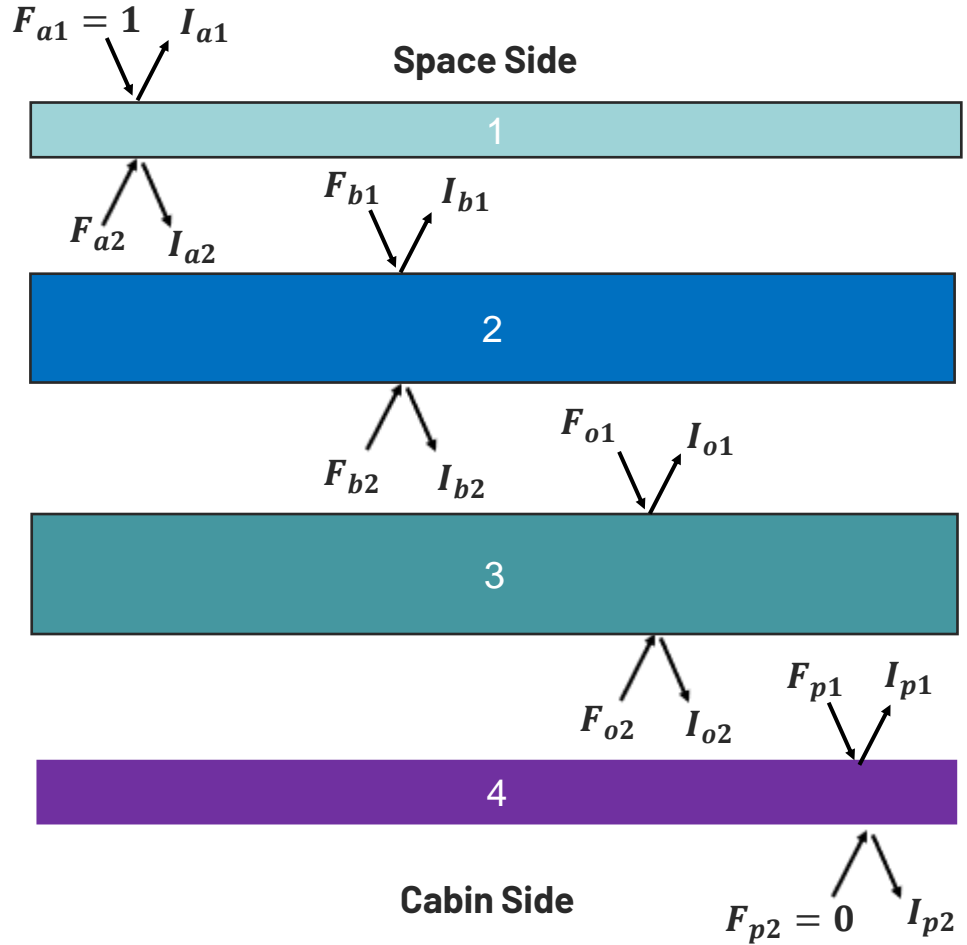
- This is conducted for a wavelength spectrum band.
 - Solar spectrum band is the selected range.
 - Could easily be adapted for other wavelength bands.



- Number of internal divisions must match Thermal Desktop model.
- Two internal reflections is typically adequate for number of terms

- Optical Props
- 2** Single Pane Analysis
- System Analysis
- Solve for Absorption
- TD Implementation

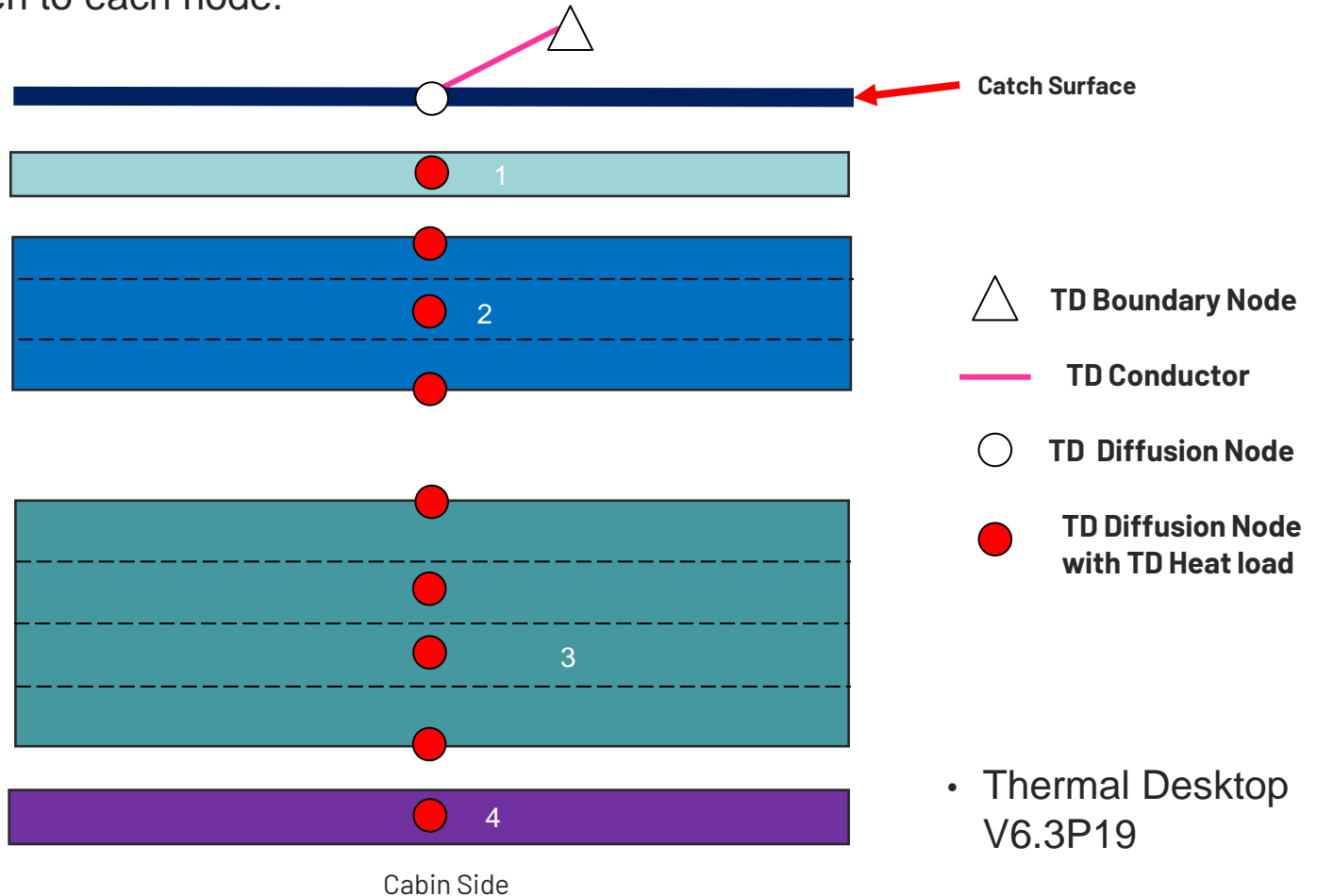
- Net radiation method mathematically ties all the panes together.
 - Equate F terms to I terms by inspection and using ray trace to fill.
 - F_{a1} is the incident sunlight.
 - Solve equations symbolically using MATLAB or preferred software.



- Optical Props
- Single Pane Analysis
- 3** System Analysis
- 4** Solve for Absorption
- TD Implementation

- Net radiation ties ray trace results into the system level relations.

- Each pane in the window is modeled as a FD brick in TD.
- Additionally, a thin shell surface dubbed “catch” surface is placed just above pane 1.
 - Calculates incoming radiation.
- Heat loads are given to each node.



- Optical Props
- Single Pane Analysis
- System Analysis
- Solve for Absorption
- 5** TD Implementation

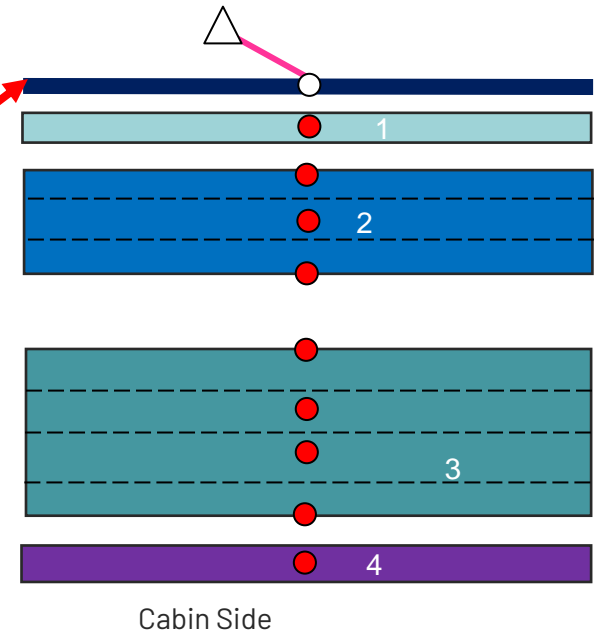
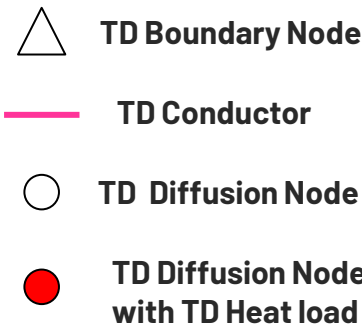
- Heat loads are driven by a combination of the previous steps.
- TD model must match internal divisions from ray trace step.
- It is Imperative to tailor each heat load correctly.

$$Q_{per\ node} = Q_{Catch\ Surface} A_{net\ radiation} \beta \gamma$$

$$\beta_{ray\ trace\ ratio} = \frac{E_{layer}}{E_{total}}$$

$$\gamma_{area\ fraction} = \frac{A_{area\ per\ node}}{A_{top\ surface}}$$

Catch Surface

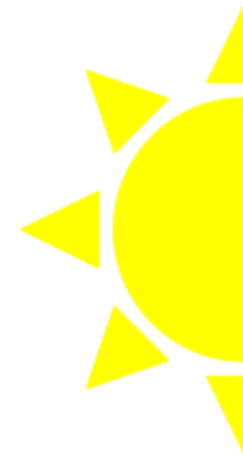
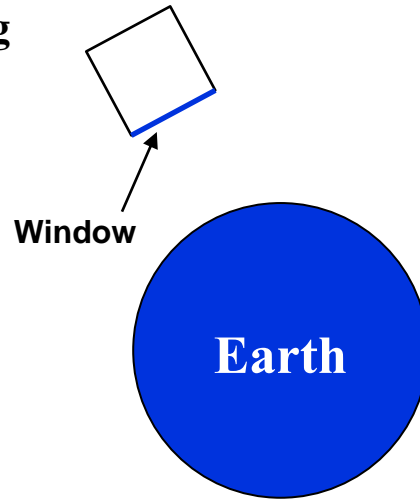


- Optical Props
- Single Pane Analysis
- System Analysis
- Solve for Absorption
- TD Implementation

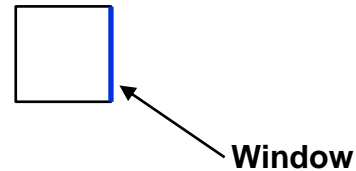
- Must use “catch” surface to tie heat loads to TD radiation calculations.

Orbit orientation for windows facing nadir and sun

Nadir Facing

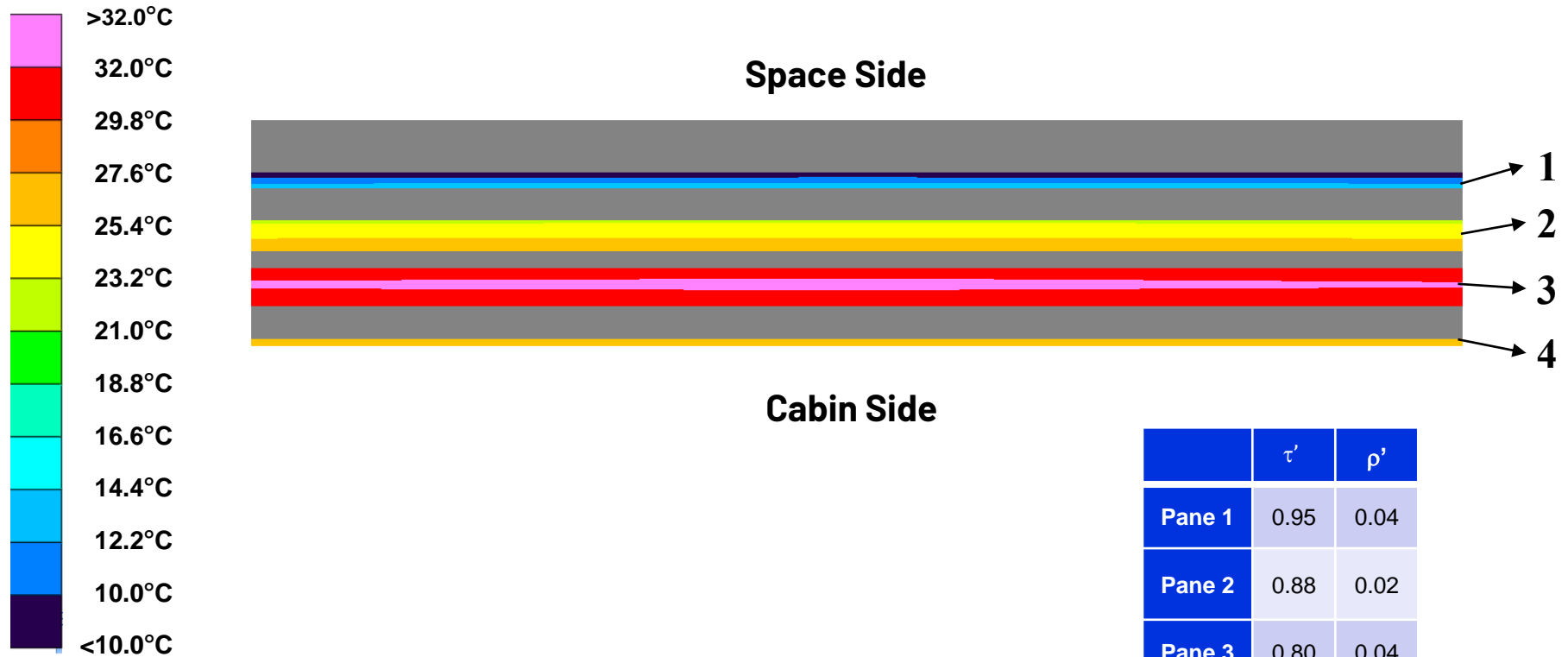


Sun Facing

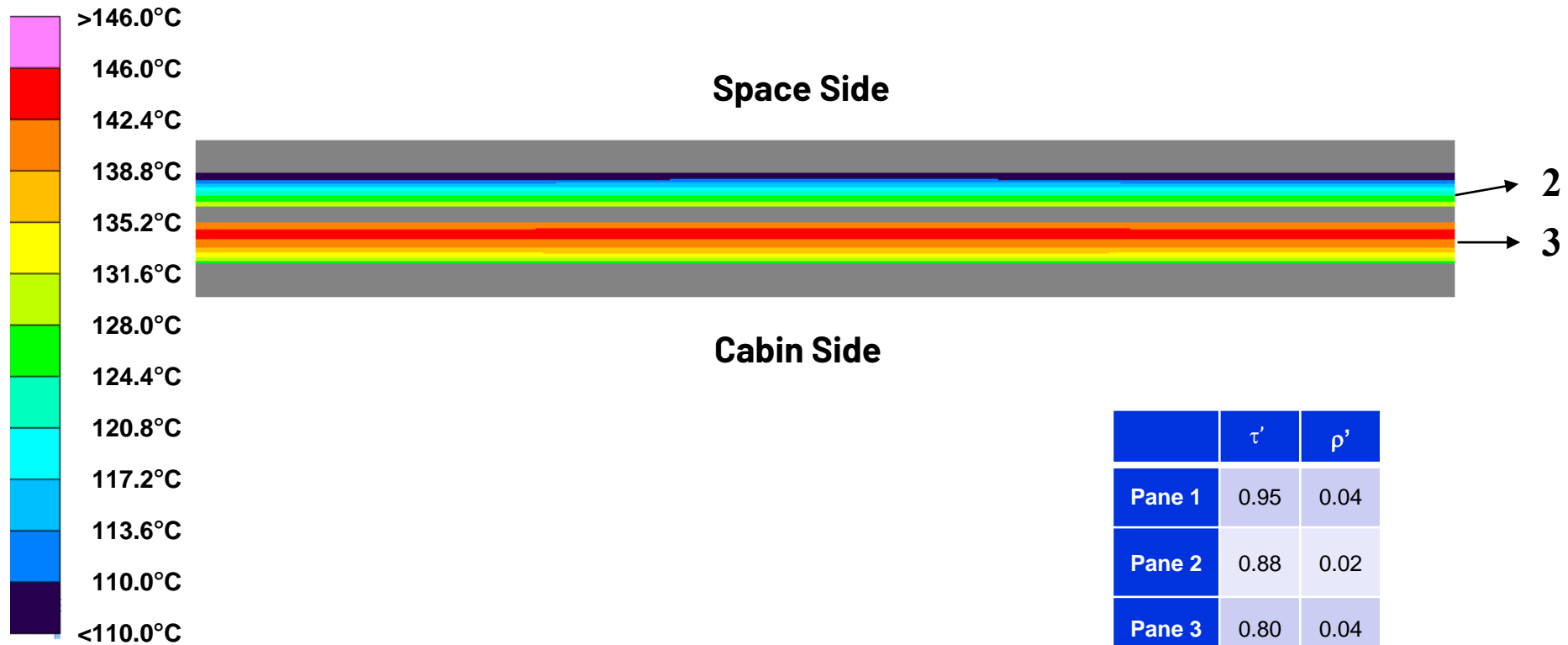


- Two orientations were run for the analysis cases.
 - Each used orbital inputs similar to the ISS orbit.
 - Beta Angle: 75°

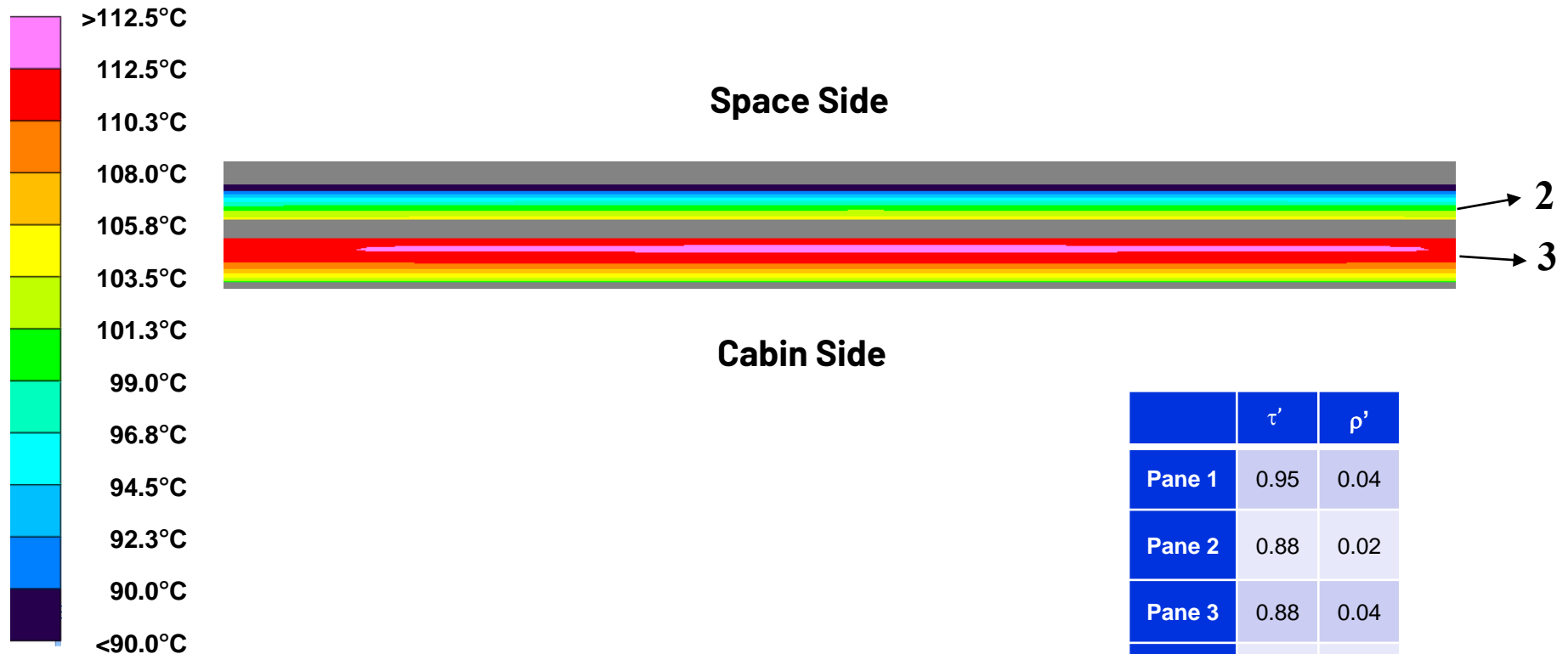
- Pane 2 and Pane 3 account for 86% of the absorbed solar energy on window.
- Inner layers of Pane 3 account for 55% of all absorbed solar energy in the window.
- Highlights the level fidelity achievable with this tool



- Shows a much higher temperature delta across the two panes.
- Higher temperature gradient within Pane 3.
- Shows importance of having high transparency on all windowpanes.

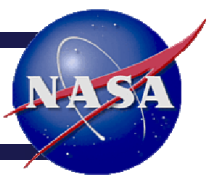


- Increasing τ' by 0.08 to .88 for pane 3 decreases all temperatures significantly
 - Max temperature decreased by 33° C
 - Internal hot spot still identified that would otherwise be unseen without tool
 - Emphasizes and quantifies the critical importance of temperature sensitivities to τ' and ultimate spacecraft window material selection





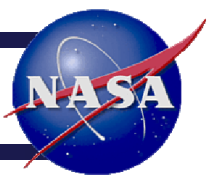
Conclusion



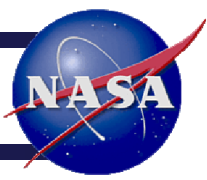
- We developed a high-fidelity window radiative analysis tool to support rapid window thermal analysis.
- It provides a deep insight into the wavelength-dependent radiation and material interaction through the thickness of material.
 - Using native TD tools would not provide this capability.
 - Teases out temperature gradients that would otherwise be unseen.
 - Can be applied across various wavelength ranges to understand window design impacts across full wavelength spectrum
- Delivers results to drive window design changes.
- Results will help develop and refine CONOPS of vehicles.
- Thermal testing must still be done to validate assumptions and analytic results.



Acknowledgements



- Thank you to Drew Smallwood at Blue Origin for his vast experience on spacecraft windows and guidance he provided.



Questions?